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# RESEARCH MEMORANDUM

THE VARIATION WITH WING ASPECT RATIO  
OF FLAP EFFECTIVENESS ON THIN RECTANGULAR  
WINGS AT TRANSONIC SPEEDS

By John G. Lowry and Robert T. Taylor

Langley Aeronautical Laboratory  
Langley Field, Va.

NATIONAL ADVISORY COMMITTEE  
FOR AERONAUTICS

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THE VARIATION WITH WING ASPECT RATIO  
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## SUMMARY

A wind-tunnel investigation has been made in the Langley high-speed 7- by 10-foot tunnel by use of the transonic-bump technique to study the effectiveness of full-span flap-type controls on 31 unswept rectangular wings. Plain flaps with flap-chord ratios of 0.1, 0.2, 0.3, and 0.4 were tested on wings of aspect ratios from 1 to 6 at Mach numbers from 0.4 to 1.1. The data for the most part are presented without analysis.

## INTRODUCTION

The variation of control effectiveness with wing aspect ratio and flap-chord ratio is well known at low subsonic speeds (refs. 1 to 5) and at supersonic speeds (refs. 6 and 7). There is, however, very little known about these effects at transonic speeds. In addition there are only a few systematic data available for controls on the thinner (4- to 6-percent thick) airfoils at transonic speeds.

The present paper presents the results of a wind-tunnel investigation to determine flap-effectiveness parameters using 31 small-scale rectangular semispan wings equipped with full-span plain flaps. The transonic speeds were obtained by using the transonic-bump technique in the Langley high-speed 7- by 10-foot tunnel. The variables investigated were wing thickness (4 and 6 percent), wing aspect ratio (aspect ratios from 1 to 6), and flap-chord ratio (flap-chord to wing-chord ratios of 0.1 to 0.4).

In order to expedite the publication of the results, no detailed analysis or discussion of the data will be made. All of the data are presented in tabulated form and in addition some data showing significant trends are presented in graphic form.

## SYMBOLS

$C_L$	lift coefficient, $\frac{\text{Twice semispan lift}}{qS}$
$C_D$	drag coefficient, $\frac{\text{Twice semispan drag}}{qS}$
$C_m$	pitching-moment coefficient about 0.25c, $\frac{\text{Twice semispan pitching moment}}{qSc}$
$C_l$	rolling-moment coefficient, $\frac{\text{Semispan rolling moment}}{qSb}$
$C_n$	yawing-moment coefficient, $\frac{\text{Semispan yawing moment}}{qSb}$
$b$	wing span, ft
$c$	wing chord, ft
$c_f$	flap chord, ft
$S$	wing area, sq ft
$t$	wing thickness, ft
$A$	wing aspect ratio, $\frac{b^2}{S}$
$q$	free-stream dynamic pressure, $\frac{1}{2}\rho V^2$ , lb/sq ft
$V$	free-stream velocity, ft/sec
$\rho$	free-stream density, slugs/cu ft
$R$	Reynolds number based on wing chord
$M$	free-stream Mach number
$M_l$	local Mach number
$\alpha$	angle of attack, deg
$\delta$	flap deflection, deg

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$\alpha_0$  flap-effectiveness parameter, effective change in wing angle of attack caused by unit angular change in control-surface deflection

$C_{L\alpha}$  lift-curve slope,  $\frac{\partial C_L}{\partial \alpha}$

#### MODELS

The geometric characteristics of the models used in the investigation are given in figure 1. The models were machined from solid steel to either NACA 65A004 or NACA 65A006 airfoil sections. The basic models had no twist or camber and had a taper ratio of 1. The aspect ratio was varied by cutting the wings at the appropriate spanwise station and filing the tip normal to the chord plane.

The flaps were machined integrally with the wing at a deflection of approximately  $10^\circ$ . The flap chords and actual deflections are given on figure 1.

#### TESTS

The tests were made by using the transonic-bump technique in the Langley high-speed 7- by 10-foot tunnel. The models were attached to a five-component electrical-strain-gage balance beneath the bump surface. The tests were made over a Mach number range from 0.4 to 1.1 at Reynolds numbers varying from  $0.5 \times 10^6$  to  $1.5 \times 10^6$  (fig. 2). The variation of the local Mach number over the bump in the vicinity of the model is shown in figure 3.

The test angles of attack varied from  $-10^\circ$  to  $25^\circ$  whenever the loads encountered did not exceed the design limit of the balance. The aspect ratio varied from 6 to 2 on the 6-percent-thick wings and from 4 to 1 on the 4-percent-thick wings. Flap chords varied from 0 to  $0.4c$ .

The data have not been corrected for jet-boundary effects on blocking since the models were sufficiently small with respect to tunnel boundaries to make the corrections negligible. No corrections were applied to account for flap deflection under load since checks indicated these too were small. The roll and yaw data presented represent the rolling- and yawing-moment coefficients resulting from deflection of the control on one wing. Since no reflection-plane corrections have been applied to the data, they represent symmetrically deflected controls and

should be reduced if applied to antisymmetric deflection. The magnitude of the corrections (reflection plane) at  $M = 0$  obtained from references 3 and 4 are given in figure 4. The variation of the correction with Mach number has not been established in the transonic-speed range but does decrease to 0 at supersonic speeds.

Because of the small physical size of the models and the resulting inaccuracies in the measurement of the forces and moments, care should be taken in analyzing the tabulated data especially at the lower Mach numbers.

#### RESULTS AND DISCUSSION

The force and moment data obtained in this investigation are presented in tables 1 to 6.

A comparison of the lift-curve slope  $C_{L\alpha}$  with theory is given in figure 5 at  $M = 0.4$ . The variation of  $C_{L\alpha}$  with Mach number is given in figure 6 for the various wings investigated. Figure 7 presents the flap-effectiveness parameter  $\alpha_8$  as a function of flap-chord ratio at all the test Mach numbers. A comparison of  $\alpha_8$  with theory is shown in figure 8.

The variation of the lift-curve slope with aspect ratio at  $M = 0.4$  (fig. 5) shows exceptional agreement with the theory of reference 8. The variation of  $C_{L\alpha}$  with  $M$  (fig. 6) is presented to give a more complete meaning to the values of  $\alpha_8$  presented.

The variation of the flap-effectiveness parameter  $\alpha_8$  with flap-chord ratio (fig. 7) is presented in order to eliminate the necessity of plotting all the lift data to see the trends with Mach number and aspect ratio. The values of  $\alpha_8$  plotted in figure 7 were obtained by dividing the change in angle of zero lift from the plain wing to the flapped wing by the flap deflection and therefore represent the value of  $\alpha_8$  at  $C_L = 0$ . This method is somewhat more accurate than using  $C_{L8}/C_{L\alpha}$  and except where there are nonlinearities in the lift curve give the same value. A comparison of the variation of  $\alpha_8$  with aspect ratio with the subsonic (ref. 5) and supersonic (ref. 6) theories (fig. 8) show that at  $M = 0.4$  and 1.1 the theories predict the variation quite satisfactorily but not the magnitude. At a Mach number of 0.9 neither theory gives either a satisfactory variation or the correct

magnitude. The disagreement between theory and experiment in magnitude is typical and results to some extent from the thickened boundary layer at the trailing edge. The magnitude at  $M = 0.4$  agrees quite well with other published data, for example reference 9. These results indicate that both in the subsonic- and supersonic-speed ranges the available theories can be used to obtain the variation of  $\alpha_\delta$  with  $A$ , but in the transonic range the variation must be obtained from experimental studies.

It should be pointed out that these data are for only one value of  $\delta$ , and, although they are useful in determining the effects of the several variables, they are not necessarily applicable to the design of a control surface that uses small deflections in the transonic-speed range (see ref. 10).

Langley Aeronautical Laboratory,  
National Advisory Committee for Aeronautics,  
Langley Field, Va., May 4, 1956.

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TABLE 1-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 1 MODEL  
 $\frac{c}{a} = 0.04$        $\frac{c_t}{c} = \text{NONE}$

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$
$M = 0.40$											
-10	-.4253					-10	-.4323	.0825	.0001	-.0490	.0150
-7	-.2982					-7	-.2832	.0440	-.0081	-.0316	.0096
-5	-.2324					-5	-.2025	.0276	-.0139	-.0222	.0071
-3	-.1666					-3	-.1280	.0201	-.0138	-.0147	.0049
-2	-.1228					-2	-.0783	.0196	-.0084	-.0094	.0033
-1	-.0965					-1	-.0534	.0164	-.0065	-.0057	.0018
0	.0702					0	-.0199	.0164	-.0009	-.0026	.0018
1	.0351					1	.0087	.0159	.0023	.0015	.0024
2	.0132					2	.0460	.0196	.0090	.0048	.0025
3	.0038					3	.0832	.0214	.0119	.0094	.0042
5	.0570					5	.1665	.0306	.0122	.0170	.0071
7	.1272					7	.2509	.0477	.0142	.0275	.0119
10	.2280					10	.3938	.0855	.0069	.0448	.0193
15	.4297					15	.6634	.1893	-.0357	.0765	.0354
20	.6884					20	.8870	.3421	-.1009	.1044	.0529
25	.8419					25	.9689	.4765	-.1365	.1123	.0672
$M = 0.60$											
-10	-.4089	.0705	-.0178	-.0383	.0193	-10	-.4396	.0901	.0047	-.0505	.0121
-7	-.2915	.0461	-.0135	-.0310	.0120	-7	-.2887	.0468	-.0098	-.0332	.0074
-5	-.2132	.0374	-.0183	-.0225	.0082	-5	-.1937	.0292	-.0126	-.0223	.0050
-3	-.1414	.0331	-.0119	-.0152	.0063	-3	-.1184	.0216	-.0147	-.0137	.0042
-2	-.1044	.0322	-.0134	-.0112	.0054	-2	-.0748	.0204	-.0114	-.0090	.0031
-1	-.0740	.0322	-.0116	-.0072	.0044	-1	-.0392	.0192	-.0059	-.0047	.0023
0	.0435	.0365	-.0048	-.0033	.0026	0	-.0154	.0188	-.0017	-.0025	.0012
1	.0174	.0374	-.0002	.0000	.0035	1	.0143	.0188	.0013	.0011	.0014
2	.0044	.0374	-.0013	.0033	.0038	2	.0558	.0192	.0115	.0051	.0014
3	.0392	.0426	.0035	.0079	.0051	3	.0855	.0221	.0165	.0090	.0031
5	.1022	.0579	.0021	.0165	.0054	5	.1711	.0321	.0172	.0180	.0050
7	.1784	.0748	.0106	.0257	.0089	7	.2756	.0513	.0092	.0296	.0117
10	.3197	.1122	.0296	.0416	.0016	10	.4301	.0946	-.0071	.0490	.0173
15	.5068	.2162	-.0173	.0673	.0294	15	.7105	.2074	-.0575	.0829	.0337
20	.7830	.3689	-.0825	.0983	.0506	20	.9362	.3600	-.1068	.1103	.0571
25	.9027	.5029	-.1925	.1089	.0699	25	1.0740	.5201	-.1571	.1280	.0791
$M = 0.80$											
-10	-.4059	.0717	-.0080	-.0457	.0150	-10	-.4471	.0894	.0143	-.0511	.0081
-7	-.2848	.0398	-.0130	-.0304	.0092	-7	-.2878	.0480	-.0064	-.0331	.0055
-5	-.2066	.0289	-.0136	-.0215	.0058	-5	-.1854	.0291	-.0168	-.0214	.0023
-3	-.1299	.0254	-.0097	-.0124	.0039	-3	-.1229	.0223	-.0167	-.0138	.0002
-2	-.0826	.0239	-.0045	-.0099	.0030	-2	-.0751	.0207	-.0103	-.0086	-.0020
-1	-.0517	.0233	-.0001	-.0067	.0030	-1	-.0466	.0212	-.0078	-.0052	-.0022
0	.0236	.0239	.0010	-.0027	.0028	0	-.0148	.0191	-.0023	-.0010	-.0017
1	.0030	.0233	.0017	.0013	.0028	1	-.0159	.0207	.0001	.0034	-.0053
2	.0280	.0254	.0084	.0045	.0032	2	.0592	.0218	.0077	.0086	-.0020
3	.0635	.0289	.0083	.0080	.0049	3	.0933	.0234	.0110	.0104	-.0007
5	.1417	.0398	.0120	.0166	.0049	5	.1752	.0346	.0125	.0193	.0020
7	.2022	.0543	.0147	.0255	.0082	7	.2730	.0526	.0039	.0304	.0061
10	.3335	.0921	.0172	.0403	.0172	10	.4380	.0951	-.0181	.0493	.0144
15	.5564	.1966	-.0176	.0694	.0311	15	.7167	.2155	-.0655	.0846	.0293
20	.8235	.3338	-.0962	.0954	.0493	20	.9488	.3770	-.1176	.1149	.0518
25	.8973	.4501	-.1333	.1052	.0607	25	1.1467	.5684	-.1673	.1381	.0744
$M = 0.90$											
-10	-.4232	.0781	-.0047	-.0474	.0159	-10	-.4467	.0823	.0158	-.0515	.0053
-7	-.2839	.0417	-.0119	-.0316	.0104	-7	-.2934	.0442	-.0052	-.0339	.0008
-5	-.1901	.0268	-.0068	-.0217	.0078	-5	-.1960	.0274	-.0133	-.0219	-.0011
-3	-.1276	.0193	-.0118	-.0134	.0055	-3	-.1204	.0195	-.0132	-.0133	-.0025
-2	-.0755	.0185	-.0076	-.0091	.0047	-2	-.0766	.0173	-.0098	-.0083	-.0027
-1	-.0560	.0167	-.0065	-.0059	.0034	-1	-.0471	.0155	-.0054	-.0050	-.0027
0	.0224	.0172	-.0010	-.0020	.0025	0	-.0120	.0151	-.0031	-.0017	-.0030
1	.0026	.0159	.0004	.0020	.0013	1	-.0131	.0145	-.0003	.0027	-.0035
2	.0378	.0193	.0092	.0051	.0017	2	.0602	.0204	.0099	.0067	-.0022
3	.0677	.0198	.0062	.0091	.0034	3	.0876	.0232	.0121	.0100	-.0033
5	.1589	.0320	.0113	.0174	.0070	5	.1741	.0344	.0108	.0199	-.0014
7	.2370	.0461	.0165	.0269	.0125	7	.2628	.0539	.0028	.0299	.0014
10	.3737	.0833	.0096	.0435	.0201	10	.4204	.1023	-.0183	.0482	.0083
15	.6368	.1922	-.0250	.0755	.0337	15	.7051	.2277	-.0675	.0830	.0231
20	.8516	.3331	-.1014	.0976	.0494	20	.9416	.3931	-.1206	.1156	.0452
25	.9298	.4560	-.1371	.1086	.0632	25	1.1168	.5709	-.1656	.1375	.0657
$M = 1.00$											
-10	-.4467	.0823	.0158	-.0515	.0053						
-7	-.2934	.0442	-.0052	-.0339	.0008						
-5	-.1960	.0274	-.0133	-.0219	-.0011						
-3	-.1204	.0195	-.0132	-.0133	-.0025						
-2	-.0766	.0173	-.0098	-.0083	-.0027						
-1	-.0471	.0155	-.0054	-.0050	-.0027						
0	-.0120	.0151	-.0031	-.0017	-.0030						
1	-.0131	.0145	-.0003	.0027	-.0035						
2	.0602	.0204	.0099	.0067	-.0022						
3	.0876	.0232	.0121	.0100	-.0033						
5	.1741	.0344	.0108	.0199	-.0014						
7	.2628	.0539	.0028	.0299	.0014						
10	.4204	.1023	-.0183	.0482	.0083						
15	.7051	.2277	-.0675	.0830	.0231						
20	.9416	.3931	-.1206	.1156	.0452						
25	1.1168	.5709	-.1656	.1375	.0657						

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TABLE 1.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 1 MODEL. Continued

$$\frac{S}{c} = 0.04 \quad \frac{C_L}{c} = 0.10$$

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
$M = 0.55$											
-10	-.2981					-10	-.3042	.0507	-.0513	-.0351	.0200
-7	-.2104	.0278	-.0826	-.0152	.0148	-7	-.1751	.0268	-.0705	-.0181	.0171
-5	-.1227					-5	-.0993	.0209	-.0713	-.0075	.0144
-3	-.0526					-3	-.0000	.0176	-.0678	-.0026	.0121
-2	-.0044					-2	.0335	-.0415	-.0691	-.0072	.0126
-1	.0307					-1	.0807	.0226	-.0667	.0117	.0123
0	.0570					0	.1036	.0268	-.0642	.0158	.0119
1	.0745					1	.1465	.0318	-.0611	.0200	.0128
2	.1008					2	.1888	.0373	-.0580	.0233	.0157
3	.1096					3	.2173	.0390	-.0619	.0271	.0190
5	.1622					5	.3080	.0586	-.0598	.0377	.0202
7	.2411					7	.3850	.0849	-.0590	.0478	.0269
10	.3507					10	.5377	.1356	-.0692	.0667	.0350
15	.5567					15	.8072	.2620	-.1072	.1006	.0523
20	.8109					20	1.0059	.4262	-.1552	.1266	.0733
25	.9863					25	1.0853	.5764	-.1910	.1345	.0881
$M = 0.60$											
$M = 1.00$											
-10	-.2610	.0344	-.0732	-.0271	.0202	-10	-.3302	.0637	-.0394	-.0371	.0190
-7	-.1631	.0283	-.0784	-.0132	.0156	-7	-.1782	.0373	-.0592	-.0191	.0155
-5	-.0696	.0309	-.0849	-.0059	.0127	-5	-.0772	.0228	-.0670	-.0065	.0131
-3	-.0152	.0331	-.0741	.0013	.0117	-3	-.0024	.0200	-.0767	-.0025	.0119
-2	.0196	.0309	-.0922	.0066	.0098	-2	.0416	.0221	-.0712	-.0072	.0112
-1	.0305	.0344	-.0813	.0099	.0117	-1	.0772	.0257	-.0674	-.0112	.0107
0	.0631	.0183	-.0795	.0145	.0117	0	.1069	.0285	-.0675	-.0151	.0105
1	.0979	.0492	-.0685	.0178	.0117	1	.1413	.0292	-.0620	-.0187	.0110
2	.1196	.0522	-.0760	.0225	.0139	2	.1805	.0373	-.0730	-.0220	.0124
3	.1588	.0631	-.0681	.0257	.0145	3	.2138	.0409	-.0558	-.0269	.0138
5	.2284	.0792	-.0684	.0276	.0174	5	.2945	.0577	-.0566	-.0360	.0188
7	.2958	.1057	-.0620	.0462	.0221	7	.3921	.0853	-.0625	-.0484	.0255
10	.4350	.1649	-.0607	.0640	.0228	10	.5570	.1428	-.0845	-.0481	.0333
15	.6634	.2801	-.0917	.0937	.0553	15	.8314	.2710	-.1285	.1024	.0508
20	.9179	.4750	-.1395	.1234	.0791	20	1.0452	.4404	.11705	.1323	.0734
25	1.0636	.6290	-.1787	.1379	.1063	25	1.1924	.6285	-.2157	.1501	.0980
$M = 0.80$											
$M = 1.05$											
-10	-.2775	.0413	-.0711	-.0304	.0204	-10	-.3470	.1297	-.0298	-.0400	.0182
-7	-.1594	.0283	-.0784	-.0132	.0156	-7	-.1843	.0425	-.0571	-.0207	.0075
-5	-.0841	.0248	-.0749	-.0058	.0120	-5	-.0853	.0314	-.0681	-.0083	.0078
-3	-.0074	.0248	-.0665	.0031	.0110	-3	-.0205	.0257	-.0705	.0010	.0058
-2	.0266	.0248	-.0692	.0076	.0110	-2	.0364	.0273	-.0689	.0062	.0058
-1	.0620	.0283	-.0676	.0112	.0110	-1	.0705	.0291	-.0666	.0104	.0058
0	.1107	.0354	-.0546	.0143	.0110	0	.0935	.0303	-.0645	.0138	.0059
1	.1210	.0354	-.0647	.0179	.0118	1	.1160	.0319	-.0655	.0179	.0046
2	.1503	.0413	-.0563	.0220	.0133	2	.1729	.0403	-.0555	.0214	.0061
3	.1845	.0463	-.0593	.0269	.0148	3	.2036	.0441	-.0571	.0255	.0102
5	.2686	.0646	-.0499	.0358	.0180	5	.2867	.0637	-.0577	.0349	.0122
7	.3321	.0865	-.0531	.0457	.0294	7	.3834	.0890	-.0676	.0466	.0189
10	.5649	.1593	-.0548	.0855	.0352	10	.5472	.1422	-.0877	.0663	.0256
15	.7010	.2642	-.0901	.0922	.0515	15	.8077	.2735	-.1344	.1008	.0491
20	.9475	.4527	-.1592	.1182	.0719	20	1.0390	.4575	-.1770	.1325	.0612
25	1.0331	.5576	-.1796	.1298	.0886	25	1.1922	.6534	-.2215	.1540	.0853
$M = 0.90$											
$M = 1.10$											
-10	-.2851	.0443	-.0597	-.0328	.0218	-10	-.2766	.0749	-.0220	-.0294	.0021
-7	-.1627	.0250	-.0769	-.0158	.0172	-7	-.1470	.0760	-.0510	-.0126	.0025
-5	-.0885	.0172	-.0773	-.0055	.0140	-5	-.0545	.0440	-.0625	-.0010	.0055
-3	-.0130	.0206	-.0728	-.0036	.0121	-3	.0023	.0407	-.0728	-.0083	.0063
-2	.0417	.0188	-.0685	-.0079	.0121	-2	.0534	.0418	-.0674	-.0124	.0073
-1	.0755	.0219	-.0669	-.0116	.0123	-1	.0926	.0440	-.0629	-.0162	.0078
0	.1146	.0250	-.0661	-.0158	.0123	0	.1056	.0451	-.0648	-.0201	.0065
1	.1406	.0281	-.0643	-.0198	.0125	1	.1394	.0481	-.0624	-.0238	.0066
2	.1835	.0364	-.0562	-.0233	.0142	2	.1906	.0547	-.0529	-.0271	.0084
3	.2109	.0396	-.0581	-.0272	.0146	3	.2330	.0599	-.0502	-.0311	.0117
5	.2890	.0583	-.0563	-.0364	.0193	5	.3071	.0615	-.0536	-.0403	.0133
7	.3749	.0820	-.0545	-.0470	.0257	7	.3920	.1093	-.0623	-.0504	.0149
10	.5116	.1356	-.0584	-.0648	.0350	10	.5336	.1703	-.0848	-.0690	.0245
15	.7810	.2541	-.0937	-.0956	.0520	15	.7950	.3043	-.1237	-.1021	.0378
20	.9893	.4251	-.1635	-.1201	.0706	20	1.0128	.4735	-.1652	-.1335	.0560
25	1.0284	.5493	-.1851	-.1272	.0849	25	1.1631	.6556	-.2090	-.1589	.0769
-10	-.2973	.0686	-.0263	-.0317	.0082	-2	.0212	.0331	-.0685	.0122	.0057

TABLE I.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 1 MODEL - Continued

 $\frac{t}{c} = 0.04$      $\frac{C_f}{t} = 0.20$ 

$a,$ deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$a,$ deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
$M = 0.60$											
-10	-0.2322					-10	-0.2334	0.0519	-0.0758	-0.0264	0.0078
-7	-0.1227					-7	-0.0881	0.0281	-0.0888	-0.0109	0.0056
-5	-0.0570					-5	-0.0062	0.0214	-0.0958	0.0000	0.0047
-3	0.0000					-3	0.0646	0.0226	-0.0947	0.0083	0.0045
-2	0.0526					-2	0.1179	0.0276	-0.0945	0.0132	0.0045
-1	0.0833					-1	0.1677	0.0293	-0.0934	0.0169	0.0051
0	0.1183					0	0.1825	0.0348	-0.0947	0.0211	0.0054
1	0.1402					1	0.2135	0.0377	-0.0964	0.0252	0.0056
2	0.1709					2	0.2570	0.0469	-0.0886	0.0282	0.0067
3	0.1972					3	0.2880	0.0524	-0.0880	0.0320	0.0092
5	0.2410					5	0.3687	0.0745	-0.0718	0.0411	0.0137
7	0.3330					7	0.4581	0.1008	-0.0886	0.0508	0.0181
10	0.4250					10	0.4108	0.1587	-0.0989	0.0697	0.0274
15	0.6179					15	0.8864	0.2930	-0.1318	0.1017	0.0457
20	0.8238					20	1.0726	0.4531	-0.1654	0.1250	0.0664
25	1.0254					25	1.0949	0.5837	-0.1910	0.1642	0.0778
$M = 0.80$											
$M = 1.00$											
-10	-0.2195	0.0265	-0.0800	-0.0264	0.0110	-10	-0.2576	0.0641	-0.0733	-0.0299	0.0081
-7	-0.1022	0.0270	-0.0811	-0.0132	0.0070	-7	-0.0962	0.0321	-0.0894	-0.0119	0.0047
-5	-0.0326	0.0161	-0.0787	-0.0040	0.0057	-5	-0.0024	0.0264	-0.1023	0.0000	0.0045
-3	0.0413	0.0248	-0.0760	0.0040	0.0054	-3	0.0712	0.0264	-0.1042	0.0083	0.0045
-2	0.0696	0.0265	-0.0784	0.0086	0.0051	-2	0.1175	0.0283	-0.1046	0.0133	0.0045
-1	0.1025	0.0322	-0.0734	0.0112	0.0063	-1	0.1579	0.0321	-0.1024	0.0173	0.0048
0	0.1282	0.0322	-0.0801	0.0165	0.0044	0	0.1864	0.0351	-0.0991	0.0213	0.0053
1	0.1717	0.0387	-0.0731	0.0191	0.0066	1	0.2220	0.0408	-0.0973	0.0245	0.0064
2	0.1978	0.0448	-0.0798	0.0231	0.0089	2	0.2612	0.0480	-0.0942	0.0281	0.0079
3	0.2282	0.0535	-0.0709	0.0271	0.0101	3	0.2873	0.0525	-0.0905	0.0313	0.0095
5	0.2978	0.0748	-0.0757	0.0362	0.0127	5	0.3740	0.0748	-0.0908	0.0403	0.0134
7	0.3739	0.1039	-0.0707	0.0442	0.0177	7	0.4713	0.1021	-0.0971	0.0515	0.0183
10	0.4999	0.1582	-0.0668	0.0620	0.0272	10	0.6387	0.1593	-0.1182	0.0695	0.0262
15	0.7042	0.2756	-0.0868	0.0877	0.0449	15	0.8833	0.2978	-0.1528	0.1009	0.0437
20	0.9542	0.4512	-0.1383	0.1134	0.0711	20	1.0452	0.4694	-0.1859	0.1272	0.0659
25	1.0998	0.6156	-0.1638	0.1247	0.0954	25	1.2063	0.6492	-0.2145	0.1423	0.0918
$M = 0.90$											
$M = 1.05$											
-10	-0.2110	0.0363	-0.0750	-0.0237	0.0099	-10	-0.2651	0.0698	-0.0701	-0.0297	0.0061
-7	-0.0989	0.0239	-0.0817	-0.0112	0.0058	-7	-0.1172	0.0437	-0.0914	-0.0138	0.0010
-5	-0.0236	0.0195	-0.0815	-0.0009	0.0043	-5	-0.0148	0.0397	-0.1017	-0.0017	0.0003
-3	0.0546	0.0218	-0.0835	0.0076	0.0043	-3	0.0592	0.0307	-0.1056	0.0076	0.0020
-2	0.0959	0.0254	-0.0747	0.0112	0.0045	-2	0.1081	0.0307	-0.1037	0.0138	0.0018
-1	0.1181	0.0275	-0.0824	0.0139	0.0045	-1	0.1411	0.0337	-0.1027	0.0162	0.0023
0	0.1609	0.0328	-0.0728	0.0179	0.0052	0	0.1729	0.0364	-0.1014	0.0200	0.0026
1	0.1830	0.0363	-0.0785	0.0224	0.0064	1	0.2036	0.0411	-0.1012	0.0235	0.0035
2	0.2214	0.0422	-0.0712	0.0280	0.0077	2	0.2469	0.0476	-0.0941	0.0264	0.0053
3	0.2538	0.0493	-0.0762	0.0300	0.0077	3	0.2810	0.0514	-0.0928	0.0311	0.0069
5	0.3350	0.0697	-0.0734	0.0385	0.0120	5	0.3640	0.0739	-0.0928	0.0390	0.0117
7	0.4118	0.0945	-0.0773	0.0479	0.0172	7	0.4539	0.1019	-0.0984	0.0501	0.0150
10	0.5387	0.1508	-0.0738	0.0636	0.0251	10	0.6075	0.1568	-0.1170	0.0673	0.0236
15	0.7645	0.2671	-0.1019	0.0896	0.0421	15	0.8578	0.2933	-0.1553	0.0991	0.0408
20	0.9918	0.4383	-0.1603	0.1142	0.0624	20	1.0739	0.4721	-0.1952	0.1267	0.0628
25	1.0390	0.5661	-0.1818	0.1191	0.0753	25	1.2241	0.6714	-0.2262	0.1453	0.0880
$M = 1.10$											
$M = 0.90$											
-10	-0.2265	0.0469	-0.0790	-0.0245	0.0076	-10	-0.2639	0.0613	-0.0656	-0.0299	0.0066
-7	-0.0924	0.0255	-0.0878	-0.0095	0.0049	-7	-0.1380	0.0377	-0.0888	-0.0140	0.0000
-5	-0.0143	0.0180	-0.0897	0.0000	0.0042	-5	-0.0252	0.0269	-0.0946	-0.0017	-0.0013
-3	0.0677	0.0206	-0.0939	0.0087	0.0040	-3	0.0515	0.0258	-0.1033	0.0076	-0.0006
-2	0.1159	0.0242	-0.0881	0.0130	0.0042	-2	0.1062	0.0280	-0.1035	0.0120	0.0000
-1	0.1393	0.0268	-0.0869	0.0158	0.0047	-1	0.1358	0.0296	-0.1019	0.0156	0.0010
0	0.1718	0.0320	-0.0844	0.0198	0.0049	0	0.1719	0.0339	-0.1011	0.0199	0.0011
1	0.2070	0.0364	-0.0886	0.0233	0.0066	1	0.1993	0.0420	-0.1014	0.0236	0.0024
2	0.2473	0.0448	-0.0784	0.0285	0.0077	2	0.2452	0.0447	-0.0935	0.0256	0.0035
3	0.2695	0.0500	-0.0767	0.0308	0.0097	3	0.2759	0.0523	-0.0914	0.0309	0.0053
5	0.3567	0.0755	-0.0814	0.0395	0.0138	5	0.3580	0.0742	-0.0923	0.0399	0.0091
7	0.4517	0.0992	-0.0845	0.0505	0.0200	7	0.4566	0.1040	-0.0959	0.0502	0.0139
10	0.5884	0.1523	-0.0902	0.0579	0.0286	10	0.6022	0.1627	-0.1184	0.0668	0.0217
15	0.8305	0.2784	-0.1098	0.0968	0.0463	15	0.8430	0.3070	-0.1574	0.0973	0.0382
20	1.0284	0.4418	-0.1678	0.1177	0.0638	20	1.0445	0.4771	-0.1929	0.1232	0.0582
25	1.0596	0.5722	-0.1846	0.1224	0.0740	25	1.2044	0.6676	-0.2289	0.1432	0.0802

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TABLE 1.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 1 MODEL - Continued

 $\frac{L}{C} = 0.04$      $\frac{C_L}{C} = 0.30$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$										$M = 0.95$	
-10	.1532					-10	.1588	.0514	-.0805	-.0166	.0099
-7	-.0350					-7	-.0174	.0323	-.0908	-.0004	.0072
-5	.0569					-5	.0608	.0275	-.0907	.0087	.0060
-3	.1094					-3	.1241	.0288	-.0961	.0173	.0056
-2	.1313					-2	.1799	.0335	-.0968	.0222	.0060
-1	.1663					-1	.2209	.0377	-.0955	.0260	.0069
0	.2101					0	.2462	.0439	-.0981	.0309	.0072
1	.2320					1	.2829	.0506	-.0966	.0354	.0081
2	.2408					2	.3227	.0598	-.0914	.0392	.0097
3	.2714					3	.3636	.0683	-.0968	.0441	.0123
5	.3590					5	.4492	.0928	-.0950	.0535	.0171
7	.4071					7	.5460	.1251	-.1015	.0652	.0228
10	.5340					10	.7074	.1857	-.1136	.0843	.0321
15	.7004					15	.9481	.1648	-.1349	.1163	.0501
20	.9018					20	1.1343	.2514	-.1592	.1416	.0711
25	1.1162					25	1.1045	.3040	-.1574	.1371	.0785
$M = 0.60$										$M = 1.00$	
-10	-.1412	.0460	-.0725	-.0172	.0139	-10	-.1840	.0608	-.0841	-.0184	.0112
-7	-.0326	.0374	-.0713	-.0040	.0089	-7	-.0360	.0404	-.0994	-.0015	.0076
-5	.0326	.0365	-.0733	.0046	.0079	-5	.0629	.0321	-.1042	.0105	.0064
-3	.1086	.0426	-.0715	.0125	.0064	-3	.1365	.0328	-.1053	.0187	.0064
-2	.1412	.0482	-.0690	.0165	.0070	-2	.1959	.0368	-.1126	.0245	.0069
-1	.1716	.0513	-.0696	.0204	.0070	-1	.2280	.0420	-.1076	.0277	.0072
0	.2064	.0586	-.0648	.0250	.0070	0	.2576	.0480	-.1114	.0313	.0074
1	.2326	.0643	-.0673	.0290	.0073	1	.2873	.0537	-.1114	.0357	.0078
2	.2607	.0717	-.0692	.0323	.0098	2	.3584	.0632	-.1022	.0393	.0100
3	.3041	.0799	-.0649	.0376	.0120	3	.3681	.0712	-.1016	.0436	.0123
5	.3910	.1047	-.0647	.0488	.0152	5	.4630	.0976	-.1043	.0537	.0164
7	.4605	.1334	-.0643	.0587	.0205	7	.5580	.1285	-.1112	.0645	.0214
10	.5930	.1925	-.0630	.0778	.0329	10	.7124	.1897	-.1294	.0832	.0293
15	.7820	.3163	-.0856	.1015	.0493	15	.9617	.3410	-.1674	.1167	.0480
20	1.0122	.5022	-.1467	.1278	.0782	20	1.1516	.5162	-.2014	.1427	.0708
25	1.1515	.6729	-.1728	.1417	.1017	25	1.2205	.6703	-.2111	.1509	.0880
$M = 0.80$										$M = 1.05$	
-10	-.1327	.0428	-.0715	-.0174	.0122	-10	-.1933	.0643	-.0848	-.0184	.0083
-7	-.0221	.0304	-.0760	-.0046	.0079	-7	-.0598	.0418	-.1026	-.0021	.0033
-5	.0310	.0274	-.0813	.0049	.0069	-5	.0568	.0337	-.1115	.0093	.0040
-3	.1062	.0327	-.0795	.0130	.0062	-3	.1364	.0337	-.1163	.0186	.0049
-2	.1549	.0363	-.0759	.0174	.0069	-2	.1853	.0346	-.1183	.0228	.0046
-1	.1844	.0407	-.0764	.0210	.0069	-1	.2217	.0391	-.1158	.0245	.0049
0	.2271	.0472	-.0723	.0251	.0071	0	.2524	.0448	-.1149	.0307	.0040
1	.2478	.0522	-.0784	.0295	.0084	1	.2853	.0514	-.1120	.0341	.0044
2	.2832	.0587	-.0733	.0336	.0103	2	.3308	.0587	-.1067	.0376	.0073
3	.3171	.0675	-.0780	.0381	.0122	3	.3638	.0682	-.1080	.0428	.0099
5	.4071	.0920	-.0699	.0474	.0161	5	.4491	.0928	-.1085	.0521	.0142
7	.4867	.1212	-.0746	.0577	.0229	7	.5559	.1237	-.1137	.0649	.0182
10	.6150	.1799	-.0821	.0756	.0311	10	.6980	.1844	-.1320	.0821	.0248
15	.8407	.3047	-.1003	.1025	.0489	15	.9550	.3354	-.1705	.1152	.0390
20	1.0472	.4787	-.1621	.1240	.0680	20	1.1414	.5166	-.2061	.1418	.0611
25	1.0767	.6079	-.1797	.1293	.0815	25	1.2778	.7201	-.2376	.1618	.0843
$M = 0.90$										$M = 1.10$	
-10	-.1470	.0447	-.0828	-.0150	.0110	-10	-.1959	.0571	-.0849	-.0196	.0029
-7	-.0325	.0302	-.0910	-.0016	.0079	-7	-.0438	.0365	-.1018	-.0027	.0032
-5	.0624	.0268	-.0851	.0083	.0059	-5	.0580	.0306	-.1105	.0047	.0024
-3	.1210	.0289	-.0866	.0170	.0053	-3	.1335	.0306	-.1161	.0179	.0019
-2	.1756	.0320	-.0903	.0213	.0055	-2	.1783	.0344	-.1178	.0222	-.0016
-1	.1977	.0359	-.0870	.0253	.0057	-1	.2188	.0387	-.1150	.0262	-.0011
0	.2368	.0421	-.0872	.0292	.0057	0	.2440	.0442	-.1170	.0295	-.0003
1	.2563	.0468	-.0912	.0336	.0053	1	.2681	.0484	-.1180	.0329	-.0008
2	.3125	.0575	-.0603	.0379	.0072	2	.3250	.0591	-.1074	.0345	-.0027
3	.3486	.0653	-.0843	.0430	.0091	3	.3600	.0689	-.1083	.0412	-.0045
5	.4384	.0908	-.0874	.0529	.0146	5	.4486	.0941	-.1108	.0505	-.0094
7	.5266	.1197	-.0875	.0628	.0219	7	.5328	.1265	-.1149	.0611	.0131
10	.6764	.1811	-.0935	.0813	.0295	10	.6893	.1910	-.1341	.0794	.0173
15	.9184	.3166	-.1235	.1121	.0484	15	.9256	.3497	-.1686	.1112	.0289
20	1.0797	.4798	-.1716	.1318	.0671	20	1.1160	.5381	-.2144	.1408	.0490
25	1.0849	.5950	-.1843	.1322	.0775	25	1.2386	.7318	-.2378	.1590	.0735

TABLE 1-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 1 MODEL - Concluded

$$\frac{k}{c} = 0.04 \quad \frac{C_f}{c} = 0.40$$

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$
$M = 0.40$										$M = 0.95$	
-10	.0306					-10	-.1291	.0569	-.0972	-.0124	.0124
-7	-.0569					-7	.0000	.0415	-.1003	.0030	.0097
-5	.0263					-5	.0906	.0397	-.1033	.0124	.0090
-3	.1007					-3	.1564	.0439	-.1047	.0203	.0092
-2	.1226					-2	.2023	.0482	-.1009	.0249	.0097
-1	.1488					-1	.2334	.0531	-.1003	.0290	.0103
0	.1882					0	.2706	.0611	-.1030	.0335	.0106
1	.2320					1	.3017	.0690	-.1078	.0384	.0112
2	.2320					2	.3363	.0787	-.1017	.0430	.0142
3	.2670					3	.3898	.0886	-.1029	.0478	.0162
5	.3327					5	.4779	.1159	-.1036	.0588	.0222
7	.4640					7	.5710	.0886	-.1093	.0708	.0276
10	.5384					10	.7275	.2105	-.1240	.0896	.0359
15	.6916					15	.9857	.3640	-.1562	.1231	.0558
20	.9061					20	1.1545	.5373	-.1870	.1473	.0742
25	1.0637					25	1.1123	.6349	-.1863	.1393	.0821
$M = 0.60$										$M = 1.00$	
-10	-.0978	.0074	-.0592	-.0112	.0117	-10	-.1484	.0672	-.1011	-.0144	.0114
-7	-.0261	.0235	-.0834	.0000	.0092	-7	-.0024	.0449	-.1122	.0029	.0093
-5	.0304	.0448	-.0720	.0086	.0114	-5	.0926	.0415	-.1133	.0157	.0083
-3	.0869	.0535	-.0738	.0158	.0114	-3	.1753	.0479	-.1191	.0223	.0076
-2	.1217	.0578	-.0695	.0204	.0114	-2	.2172	.0536	-.1187	.0270	.0074
-1	.1521	.0643	-.0713	.0250	.0114	-1	.2587	.0596	-.1211	.0317	.0071
0	.2086	.0491	-.0581	.0284	.0114	0	.2848	.0660	-.1214	.0360	.0074
1	.2390	.0761	-.0768	.0336	.0120	1	.3169	.0729	-.1231	.0407	.0081
2	.2781	.0695	-.0719	.0382	.0133	2	.3691	.0828	-.1169	.0414	.0105
3	.3151	.0813	-.0767	.0435	.0148	3	.4071	.0933	-.1179	.0497	.0131
5	.4085	.0960	-.0737	.0547	.0193	5	.4937	.1184	-.1174	.0609	.0171
7	.4780	.1349	-.0800	.0646	.0256	7	.6006	.1548	-.1261	.0731	.0226
10	.6041	.2064	-.0826	.0818	.0366	10	.7525	.2217	-.1433	.0918	.0318
15	.7779	.3346	-.1002	.1075	.0534	15	.9875	.3760	-.1784	.0918	.0502
20	1.0521	.5267	-.1648	.1371	.0802	20	1.1774	.5564	-.2138	.1512	.0756
25	1.0995	.6627	-.1742	.1411	.0960	25	1.1869	.7005	-.2110	.1523	.0877
$M = 0.80$										$M = 1.05$	
-10	-.1210	.0472	-.0797	-.0107	.0123	-10	-.1466	.0686	-.1248	-.0145	.0071
-7	-.0758	.0398	-.0633	.0013	.0097	-7	-.0102	.0582	-.1210	.0010	.0058
-5	.0708	.0363	-.0752	.0103	.0094	-5	.0863	.0456	-.1219	.0131	.0056
-3	.1372	.0437	-.0797	.0179	.0099	-3	.1647	.0475	-.1263	.0214	.0049
-2	.1697	.0472	-.0742	.0224	.0101	-2	.2113	.0532	-.1286	.0258	.0045
-1	.1995	.0531	-.0785	.0269	.0105	-1	.2511	.0570	-.1253	.0303	.0040
0	.2375	.0596	-.0737	.0304	.0107	0	.2795	.0625	-.1258	.0348	.0051
1	.2641	.0676	-.0850	.0358	.0116	1	.3113	.0704	-.1292	.0393	.0059
2	.3231	.0770	-.0924	.0398	.0135	2	.3704	.0811	-.1218	.0434	.0079
3	.3394	.0856	-.0829	.0439	.0148	3	.4033	.0911	-.1232	.0489	.0111
5	.4308	.1148	-.0783	.0546	.0204	5	.4919	.1172	-.1235	.0589	.0152
7	.5061	.1487	-.0815	.0649	.0255	7	.5862	.1520	-.1337	.0700	.0197
10	.6477	.2068	-.0823	.0842	.0354	10	.7226	.2168	-.1489	.0889	.0266
15	.8498	.3266	-.1092	.1088	.0534	15	.9634	.3776	-.1825	.1234	.0431
20	1.0298	.4904	-.1614	.1276	.0716	20	1.1497	.5699	-.2160	.1493	.0651
25	1.0298	.5949	-.1710	.1294	.0804	25	1.2724	.7621	-.2488	.1679	.0888
$M = 0.90$										$M = 1.10$	
-10	-.1249	.0544	-.0950	-.0107	.0125	-10	-.1706	.0645	-.1013	-.0166	.0003
-7	.0078	.0398	-.0935	.0032	.0093	-7	-.0219	.0516	-.1128	.0017	.0010
-5	.0820	.0385	-.0926	.0115	.0091	-5	.0700	.0473	-.1177	.0120	.0006
-3	.1444	.0435	-.0945	.0194	.0091	-3	.1531	.0324	-.1225	.0202	.0010
-2	.1867	.0461	-.0892	.0293	.0089	-2	.1881	.0558	-.1163	.0242	.0006
-1	.2212	.0333	-.0937	.0280	.0091	-1	.2188	.0602	-.1203	.0279	.0006
0	.2577	.0609	-.0914	.0324	.0093	0	.2538	.0689	-.1203	.0319	.0014
1	.2876	.0697	-.0973	.0367	.0095	1	.3063	.0733	-.1170	.0375	.0030
2	.3357	.0768	-.0874	.0411	.0112	2	.3500	.0860	-.1179	.0408	.0046
3	.3725	.0882	-.0920	.0466	.0140	3	.3806	.0969	-.1196	.0458	.0062
5	.4581	.1153	-.0928	.0564	.0187	5	.4594	.1247	-.1186	.0558	.0102
7	.5504	.1460	-.0946	.0691	.0240	7	.5600	.1614	-.1296	.0674	.0151
10	.6975	.2079	-.1052	.0873	.0329	10	.7000	.2367	-.1512	.0859	.0200
15	.9525	.3495	-.1296	.1196	.0537	15	.9450	.3981	-.1823	.1181	.0348
20	1.0779	.5093	-.1779	.1354	.0702	20	1.1288	.5917	-.2199	.1454	.0564
25	1.0671	.6145	-.1810	.1354	.0789	25	1.2513	.7790	-.2506	.1653	.0792

TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL

 $\frac{L}{c} = 0.04$        $\frac{C_L^2}{c} = \text{NONE}$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$
$M = 0.40$											
$M = 0.45$											
-10	-4.955					-10	-5.973	.1097	.0248	.0725	.0139
-7	-3.859	.0481	-.0284	-.0464	.0075	-7	-4.396	.0571	.0026	.0524	.0074
-5	-2.675					-5	-3.378	.0281	-.0105	-.0364	.0043
-3	-1.710					-3	-2.012	.0115	-.0148	-.0226	.0024
-2	-1.184					-2	-1.1428	.0076	-.0089	-.0154	.0018
-1	-0.833					-1	-0.820	.0040	-.0063	-.0085	.0014
0	-0.329					0	-0.286	.0031	.0030	-.0026	.0009
1	.0153					1	.0224	.0019	.0078	.0038	.0007
2	.0417					2	.0944	.0052	.0176	.0109	.0012
3	.0921					3	.1503	.0074	.0208	.0177	.0023
5	.1776					5	.2849	.0232	.0191	.0330	.0045
7	.2654					7	.4023	.0473	.0087	.0480	.0078
10	.4078					10	.5787	.0992	-.0115	.0712	.0155
15	.6095					15	.4148	.2210	.0548	.1055	.0655
20	.7366					20	.4731	.3604	-.0204	.1168	.0496
25	.6731					25	.4731	.4581	-.0337	.1187	.0596
$M = 0.60$											
$M = 0.75$											
-10	-2.744	.0309	-.0009	-.0701	.0136	-10	-6.342	.1227	.0504	.0766	.0155
-7	-4.069	.0481	-.0284	-.0464	.0075	-7	-4.4608	.0671	.0211	.0557	.0081
-5	-2.872	.0294	-.0227	-.0327	.0045	-5	-3.302	.0980	.0009	-.0378	.0041
-3	-1.893	.0231	-.0178	-.0211	.0028	-3	-2.043	.0190	-.0145	-.0225	.0019
-2	-1.327	.0198	-.0154	-.0149	.0021	-2	-1.154	.0126	-.0120	-.0150	.0012
-1	-0.849	.0187	-.0136	-.0111	.0019	-1	-0.784	.0102	-.0100	-.0081	.0009
0	-.0261	.0187	-.0046	-.0036	.0016	0	-.0214	.0072	-.0029	-.0014	.0007
1	.0131	.0187	-.0063	-.0023	.0016	1	.0499	.0094	-.0010	-.0067	.0009
2	.0609	.0220	-.0024	-.0078	.0017	2	.1116	.0116	.0082	.0135	.0016
3	.1131	.0242	-.0015	-.0140	.0022	3	.1686	.0160	.0103	.0207	.0026
5	.2089	.0374	-.0101	-.0264	.0047	5	.3183	.0350	-.0029	-.0373	.0055
7	.3177	.0616	-.0146	-.0396	.0076	7	.4276	.0584	-.0133	.0514	.0086
10	.4874	.1123	-.0141	-.0611	.0146	10	.6247	.1125	-.0438	.0757	.0173
15	.6984	.2230	-.0507	-.0891	.0265	15	.8979	.2394	-.0812	.1117	.0354
20	.7746	.3198	-.0980	-.0974	.0337	20	1.0713	.4007	-.1218	.1369	.0596
25	.7485	.3884	-.0971	-.0941	.0498	25	1.1497	.5549	-.1864	.1449	.0725
$M = 0.80$											
$M = 1.00$											
-10	-4.5891	.0918	.0020	.0692	.0131	-10	-6.553	.1306	.0689	.0752	.0141
-7	-4.340	.0433	-.0196	-.0488	.0072	-7	-4.641	.0713	.0324	.0526	.0070
-5	-3.015	.0199	-.0165	-.0336	.0041	-5	-3.390	.0433	.0161	-.0375	-.0047
-3	-2.037	.0097	-.0091	-.0213	.0022	-3	-2.207	.0251	.0050	-.0233	.0014
-2	-1.388	.0062	-.0063	-.0146	.0015	-2	-1.502	.0188	.0025	-.0155	.0006
-1	-0.827	.0044	-.0063	-.0081	.0012	-1	-0.887	.0146	-.0017	-.0088	.0003
0	-.0251	.0037	-.0018	-.0027	.0010	0	-.0205	.0117	-.0028	-.0017	.0003
1	.0165	.0025	-.0041	-.0029	.0009	1	.0387	.0126	-.0019	-.0057	.0000
2	.0723	.0090	-.0104	-.0092	.0011	2	.1160	.0159	.0008	.0138	.0014
3	.1270	.0090	-.0119	-.0159	.0017	3	.1797	.0196	-.0033	.0212	.0027
5	.2480	.0229	-.0226	-.0298	.0043	5	.3185	.0370	-.0116	.0368	.0050
7	.3635	.0446	-.0261	-.0441	.0072	7	.4368	.0613	-.0244	.0509	.0081
10	.5433	.0948	-.0152	-.0665	.0142	10	.6302	.1183	-.0549	.0751	.0159
15	.6998	.1961	-.0535	-.0878	.0284	15	.9328	.2590	-.1090	.1142	.0348
20	.7707	.2941	-.0890	-.0952	.0393	20	1.1558	.4308	-.1492	.1432	.0577
25	.7825	.1924	-.0751	-.0974	.0488	25	1.2468	.5930	-.1942	.1551	.0774
$M = 0.90$											
$M = 1.05$											
-10	-4.5891	.0918	.0020	.0692	.0131	-10	-6.438	.1225	.0709	.0070	.0113
-7	-4.543	.0512	-.0038	-.0519	.0079	-7	-4.796	.0670	.0322	.0520	.0052
-5	-3.176	.0240	-.0115	-.0361	.0049	-5	-3.460	.0391	.0171	-.0363	.0018
-3	-2.083	.0096	-.0120	-.0223	.0027	-3	-2.277	.0229	.0069	-.0228	.0000
-2	-1.419	.0044	-.0103	-.0148	.0020	-2	-1.533	.0177	.0031	-.0158	-.0008
-1	-0.833	.0016	-.0062	-.0089	.0017	-1	-0.920	.0142	.0003	-.0091	-.0012
0	-.0247	.0009	-.0019	-.0024	.0013	0	-.0263	.0122	-.0010	-.0017	-.0014
1	.0208	.0000	-.0074	-.0034	.0012	1	.0416	.0113	-.0028	.0061	-.0009
2	.0794	.0033	-.0144	-.0099	.0013	2	.1117	.0162	-.0008	.0133	.0000
3	.1445	.0070	-.0178	-.0174	.0025	3	.1774	.0216	-.0028	.0204	.0010
5	.2747	.0228	-.0233	-.0326	.0050	5	.3086	.0382	-.0136	.0350	.0036
7	.4061	.0448	-.0141	-.0480	.0086	7	.4292	.0646	-.0282	.0507	.0061
10	.5702	.0937	-.0013	-.0693	.0156	10	.6131	.1198	-.0593	.0728	.0127
15	.8097	.2145	-.0349	-.1027	.0326	15	.9000	.2569	-.1113	.1103	.0314
20	.8696	.3291	-.1143	-.1066	.0445	20	1.1430	.4362	-.1620	.1412	.0535
25	.8722	.4225	-.1238	-.1086	.0558	25	1.8613	.6246	-.5222	.1601	.0786

TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

$$\frac{t}{c} = 0.04 \quad \frac{C_L}{c} = 0.10$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$											
$M = 0.60$											
$M = 0.80$											
$M = 0.90$											
$M = 0.95$											
$M = 1.00$											
$M = 1.05$											
$M = 1.10$											

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TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

$$\frac{L}{c} = 0.04 \quad \frac{C_f}{c} = 0.20$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_N$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_N$
$M = 0.40$											
$M = 0.55$											
-10	-.2164					-10	-.3739	.0935	-.0914	-.0418	.0102
-7	-.1027					-7	-.1981	.0524	-.1149	-.0168	.0055
-5	-.0109					-5	-.0520	.0340	-.1281	-.0019	.0038
-3	+.0830					-3	-.0718	.0290	-.1373	.0124	.0038
-2	+.1333					-2	+.1498	.0311	-.1412	.0201	.0038
-1	+.1704					-1	+.2229	.0357	-.1459	.0279	.0047
0	+.2141					0	+.2860	.0402	-.1413	.0352	.0054
1	+.2797					1	+.3442	.0453	-.1369	.0418	.0064
2	+.3015					2	+.4160	.0560	-.1332	.0498	.0085
3	+.3518					3	+.4804	.0655	-.1290	.0573	.0102
5	+.4392					5	+.6265	.0935	-.1388	.0751	.0148
7	+.5375					7	+.7379	.1285	-.1404	.0892	.0204
10	+.6730					10	+.8865	.1961	-.1494	.1108	.0299
15	+.8194					15	1.0474	.3251	-.1555	.1349	.0475
20	+.8806					20	1.0771	.4591	-.2013	.1341	.0610
25	+.7888					25	1.0474	.5565	-.2050	.1672	.0711
$M = 0.60$											
$M = 1.00$											
-10	-.2656	.0475	-.0836	-.0304	.0067	-10	-.3446	.1098	-.0630	-.0490	.0121
-7	-.1051	.0208	-.1136	-.0099	.0039	-7	-.2534	.0460	-.0948	-.0252	.0068
-5	.0044	.0171	-.1109	.0033	.0024	-5	-.1089	.0487	-.1172	-.0075	.0050
-3	.1065	.0197	-.1077	.0148	.0023	-3	+.0403	.0365	-.1418	-.0088	.0041
-2	.1604	.0223	-.1052	.0210	.0020	-2	+.1220	.0375	-.1461	.0180	.0039
-1	.2070	.0287	-.1037	.0266	.0028	-1	+.2013	.0391	-.1529	-.0264	.0041
0	.2634	.0330	-.0910	.0326	.0035	0	+.2759	.0443	-.1534	-.0343	.0050
1	.3067	.0384	-.0960	.0380	.0047	1	+.3386	.0501	-.1516	-.0413	.0062
2	.3501	.0479	-.0951	.0436	.0056	2	+.4121	.0588	-.1492	-.0492	.0082
3	.4075	.0596	-.0888	.0501	.0067	3	+.4772	.0696	-.1515	-.0566	.0099
5	.5105	.0837	-.0774	.0633	.0103	5	+.6134	.0996	-.1582	.0727	.0144
7	.6244	.1168	-.0733	.0776	.0158	7	+.7341	.1359	-.1679	.0871	.0197
10	.7609	.1786	-.0752	.0970	.0253	10	.9070	.2079	-.1825	.1103	.0300
15	.8931	.2942	-.1291	.1135	.0410	15	1.1201	.3517	-.1898	.1419	.0511
20	.8953	.3891	-.1549	.1118	.0512	20	1.2930	.5253	-.2223	.1599	.0723
25	.8411	.4574	-.1489	.1089	.0585						
$M = 0.80$											
$M = 1.05$											
-10	-.2935	.0597	-.0948	-.0325	.0066	-10	-.4637	.1135	-.0415	-.0516	.0111
-7	-.1155	.0285	-.1238	-.0104	.0039	-7	-.2676	.0703	-.0832	-.0275	.0062
-5	.0096	.0196	-.1199	.0039	.0024	-5	-.1213	.0499	-.1048	-.0098	.0041
-3	.1089	.0203	-.1192	.0186	.0024	-3	+.0091	.0396	-.1240	-.0052	.0035
-2	.1714	.0228	-.1207	.0227	.0027	-2	+.0873	.0396	-.1306	-.0134	.0034
-1	.2244	.0278	-.1186	.0285	.0034	-1	+.1585	.0405	-.1354	-.0268	.0038
0	.2855	.0352	-.1108	.0352	.0040	0	+.2506	.0430	-.1454	-.0310	.0047
1	.3406	.0391	-.1065	.0415	.0053	1	+.3016	.0493	-.1439	-.0370	.0054
2	.3870	.0466	-.1060	.0469	.0061	2	+.3810	.0583	-.1439	-.0447	.0071
3	.4532	.0578	-.0985	.0591	.0078	3	+.4410	.0678	-.1505	-.0525	.0092
5	.5694	.0818	-.0907	.0681	.0123	5	+.5782	.0967	-.1568	.0679	.0130
7	.6930	.1154	-.0867	.0848	.0178	7	.7098	.1338	-.1689	-.0834	.0182
10	.8446	.1788	-.0924	.1047	.0274	10	.8798	.2058	-.1829	.1063	.0276
15	.8917	.2837	-.1430	.1116	.0401	15	1.1565	.3646	-.2195	.1393	.0494
20	.9152	.3936	-.1628	.1138	.0514						
25	.8887	.4761	-.1805	.1116	.0622						
$M = 0.90$											
$M = 1.10$											
-10	-.3335	.0798	-.1024	-.0374	.0083	-10	-.4572	.1052	-.0391	-.0505	.0091
-7	-.1778	.0437	-.1211	-.0162	.0051	-7	-.2641	.0649	-.0773	-.0267	.0046
-5	-.0286	.0293	-.1265	.0000	.0037	-5	-.1255	.0469	-.0987	-.0099	.0032
-3	.0795	.0256	-.1271	.0128	.0034	-3	+.0055	.0381	-.1167	-.0043	.0027
-2	.1531	.0287	-.1277	.0201	.0034	-2	+.0775	.0381	-.1210	-.0121	.0027
-1	.2245	.0326	-.1292	.0278	.0042	-1	+.1495	.0399	-.1259	-.0199	.0029
0	.2790	.0374	-.1286	.0348	.0047	0	+.2106	.0421	-.1323	-.0265	.0038
1	.3439	.0437	-.1285	.0422	.0059	1	+.2815	.0475	-.1369	-.0342	.0048
2	.4065	.0549	-.1192	.0487	.0077	2	+.3568	.0577	-.1361	-.0419	.0064
3	.4685	.0628	-.1178	.0561	.0094	3	+.4288	.0690	-.1393	-.0505	.0082
5	.6152	.0899	-.1135	.0738	.0143	5	+.5565	.0971	-.1527	-.0654	.0120
7	.7346	.1236	-.1157	.0886	.0397	7	+.6787	.1328	-.1645	-.0806	.0166
10	.8591	.1864	-.1256	.1083	.0287	10	.8446	.2018	-.1787	.1010	.0253
15	.9811	.3076	-.1506	.1240	.0442	15	1.1043	.3509	-.2182	.1328	.0457
20	1.0071	.4239	-.1874	.1240	.0566						
25	.9734	.5170	-.1840	.1221	.0670						

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TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

$$\frac{k}{c} = 0.04 \quad \frac{c_L}{c} = 0.30$$

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_i$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_i$	$c_n$
$M = 0.40$						$M = 0.95$					
-10	-0.1093					-10	-0.2984	.0755	-0.0992	-0.0330	.0111
-7	-0.0437					-7	-0.1065	.0396	-0.1230	-0.0090	.0069
-5	.0656					-5	.0452	.0284	-0.1359	.0081	.0059
-3	.1486					-3	.1647	.0284	-0.1448	.0220	.0060
-2	.1988					-2	.2495	.0334	-0.1524	.0304	.0070
-1	.2597					-1	.3194	.0390	-0.1481	.0379	.0076
0	.3125					0	.3776	.0478	-0.1505	.0455	.0082
1	.3474					1	.4295	.0557	-0.1443	.0524	.0097
2	.3715					2	.5076	.0657	-0.1370	.0603	.0118
3	.4261					3	.5782	.0780	-0.1419	.0693	.0139
5	.5048					5	.7131	.1117	-0.1458	.0853	.0190
7	.5943					7	.8147	.1486	-0.1498	.0999	.0238
10	.7276					10	.9360	.2149	-0.1514	.1187	.0285
15	.8347					15	1.1068	.3507	-0.1544	.1424	.0493
20	.9003					20	1.0895	.4701	-0.1914	.1360	.0603
25	.7976					25	1.0524	.5760	-0.1911	.1337	.0669
$M = 0.60$						$M = 1.00$					
-10	-0.1734	.0171	-0.0636	-0.0219	.0085	-10	-0.3611	.0926	-0.0741	-0.0396	.0137
-7	-0.0336	.0043	-0.0917	-0.0026	.0052	-7	-0.1776	.0583	-0.1080	-0.0158	.0090
-5	.0813	.0022	-0.0837	.0105	.0044	-5	-0.0059	.0401	-0.1358	.0040	.0070
-3	.1843	.0069	-0.0818	.0217	.0050	-3	.1385	.0367	-0.1546	.0194	.0066
-2	.2428	.0113	-0.0779	.0281	.0057	-2	.2368	.0367	-0.1584	.0281	.0070
-1	.2851	.0165	-0.0762	.0340	.0065	-1	.3043	.0439	-0.1610	.0366	.0076
0	.3338	.0241	-0.0721	.0400	.0078	0	.3659	.0495	-0.1602	.0439	.0082
1	.3775	.0330	-0.0755	.0464	.0093	1	.4281	.0583	-0.1556	.0519	.0099
2	.4346	.0431	-0.0691	.0526	.0098	2	.5032	.0693	-0.1551	.0591	.0117
3	.4823	.0544	-0.0692	.0592	.0113	3	.5684	.0810	-0.1574	.0675	.0139
5	.5961	.0843	-0.0579	.0725	.0141	5	.7033	.1159	-0.1659	.0833	.0189
7	.6937	.1194	-0.0549	.0858	.0199	7	.8170	.1590	-0.1741	.0984	.0238
10	.8281	.1527	-0.0613	.1049	.0298	10	.9709	.2335	-0.1869	.1200	.0337
15	.8931	.3018	-0.1075	.1154	.0429	15	1.2172	.3937	-0.2048	.1523	.0530
20	.8910	.4030	-0.1293	.1131	.0528	20	1.2646	.5392	-0.2135	.1595	.0726
25	.8498	.4724	-0.1176	.1075	.0600	25	1.2078	.6428	-0.2250	.1505	.0747
$M = 0.80$						$M = 1.05$					
-10	-0.2081	.0372	-0.0863	-0.0280	.0077	-10	-0.3776	.0529	-0.0579	-0.0418	.0115
-7	-0.0338	.0156	-0.1107	-0.0029	.0047	-7	-0.1871	.0579	-0.1017	-0.0177	.0076
-5	.0816	.0109	-0.1053	.0105	.0039	-5	-0.0397	.0432	-0.1190	-0.0005	.0063
-3	.1853	.0137	-0.1046	.0228	.0044	-3	.0964	.0388	-0.1377	.0145	.0059
-2	.2464	.0181	-0.1048	.0308	.0052	-2	.1671	.0350	-0.1480	.0240	.0081
-1	.3052	.0258	-0.1002	.0358	.0065	-1	.2687	.0435	-0.1528	.0323	.0064
0	.3530	.0329	-0.1005	.0425	.0075	0	.3356	.0497	-0.1604	.0409	.0066
1	.4045	.0391	-0.1029	.0488	.0083	1	.4014	.0577	-0.1633	.0488	.0073
2	.4633	.0506	-0.0877	.0551	.0095	2	.4739	.0692	-0.1560	.0557	.0094
3	.5261	.0625	-0.0903	.0629	.0113	3	.5408	.0825	-0.1606	.0643	.0112
5	.6413	.0915	-0.0828	.0763	.0161	5	.6689	.1154	-0.1678	.0793	.0153
7	.7561	.1249	-0.0805	.0928	.0220	7	.7823	.1578	-0.1763	.0874	.0194
10	.8914	.1902	-0.0862	.1102	.0313	10	.9501	.2375	-0.1940	.1156	.0290
15	.9120	.3010	-0.1315	.1156	.0421	15	1.2109	.4048	-0.2258	.1493	.0491
20	.9267	.4066	-0.1447	.1160	.0507						
25	.8973	.4933	-0.1468	.1142	.0603						
$M = 0.90$						$M = 1.10$					
-10	-0.2530	.0599	-0.1073	-0.0281	.0093	-10	-0.3819	.0894	-0.0566	-0.0424	.0076
-7	-0.0714	.0293	-0.1256	-0.0055	.0062	-7	-0.1964	.0558	-0.0986	-0.0184	.0042
-5	.0616	.0217	-0.1286	.0093	.0050	-5	-0.0360	.0410	-0.1181	-0.0012	.0034
-3	.1719	.0239	-0.1330	.0220	.0056	-3	.0873	.0384	-0.1341	.0131	.0035
-2	.2530	.0278	-0.1354	.0301	.0067	-2	.1746	.0410	-0.1422	.0224	.0040
-1	.3049	.0322	-0.1324	.0368	.0072	-1	.2401	.0445	-0.1461	.0290	.0044
0	.3633	.0402	-0.1292	.0440	.0081	0	.3023	.0483	-0.1504	.0371	.0049
1	.4320	.0498	-0.1286	.0516	.0098	1	.3721	.0574	-0.1548	.0445	.0059
2	.4865	.0599	-0.1174	.0579	.0114	2	.4441	.0692	-0.1520	.0520	.0074
3	.5553	.0717	-0.1180	.0667	.0138	3	.5074	.0821	-0.1591	.0603	.0088
5	.6928	.1030	-0.1198	.0834	.0187	5	.6460	.1159	-0.1646	.0770	.0120
7	.7940	.1382	-0.1166	.0980	.0228	7	.7573	.1600	-0.1777	.0907	.0154
10	.9159	.2042	-0.1246	.1157	.0318	10	.9188	.2420	-0.1896	.1112	.0236
15	1.0145	.3267	-0.1448	.1283	.0455	15	1.1654	.4057	-0.2275	.1424	.0422
20	1.0171	.4402	-0.1714	.1256	.0558						
25	.9730	.5270	-0.1675	.1236	.0637						

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TABLE 2 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

 $\frac{C_L}{C_D} = 0.04$        $\frac{C_L}{C_D} = 0.40$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_N$	$C_I$	$C_H$	$\alpha$ , deg	$C_L$	$C_D$	$C_N$	$C_I$	$C_H$
$M = 0.40$						$M = 0.95$					
-10	-.0329					-10	-.2217	.0856	-.1403	-.0287	.0055
-7	.0394					-7	-.0286	.0550	-.1685	-.0055	.0015
-5	.1468					-5	.1280	.0443	-.1779	.0119	.0007
-3	.2279					-3	.2559	.0467	-.1815	.0261	.0006
-2	.2761					-2	.3329	.0534	-.1837	.0246	.0012
-1	.3133					-1	.3950	.0598	-.1788	.0114	.0023
0	.3506					0	.4509	.0681	-.1794	.0482	.0034
1	.4009					1	.5068	.0776	-.1778	.0560	.0048
2	.4338					2	.5839	.0861	-.1707	.0635	.0077
3	.4776					3	.6410	.0963	-.1675	.0714	.0103
5	.5631					5	.7528	.1267	-.1621	.0861	.0154
7	.6748					7	.8522	.0825	-.1628	.0999	.0213
10	.7887					10	.9590	.1137	-.1565	.1157	.0308
15	.9158					15	1.1081	.0911	-.1478	.1376	.0251
20	.9290					20	1.1130	.1237	-.1772	.1327	.0312
25	.8457					25	1.0708	.1482	-.1635	.1285	.0252
$M = 0.60$						$M = 1.00$					
-10	-.1250	.0267	-.1009	-.0244	-.0016	-10	-.2924	.0979	-.1143	-.0368	.0073
-7	.0250	.0161	-.1220	-.0058	-.0043	-7	-.0874	.0657	-.1538	-.0115	.0033
-5	.1337	.0176	-.1219	-.0089	-.0032	-5	.0850	.0526	-.1791	.0071	.0019
-3	.2261	.0252	-.1133	-.0213	-.0008	-3	.2217	.0517	-.1894	.0227	.0015
-2	.2816	.0294	-.1097	-.0277	-.0006	-2	.3161	.0549	-.1915	.0317	.0016
-1	.3272	.0391	-.1047	-.0345	.0017	-1	.3732	.0613	-.1909	.0389	.0019
0	.3783	.0498	-.1036	-.0424	.0086	0	.4409	.0493	-.1891	.0448	.0091
1	.4273	.0615	-.0953	-.0503	.0111	1	.5003	.0777	-.1885	.0270	.0046
2	.4675	.0722	-.0993	-.0564	.0146	2	.5681	.0906	-.1835	.0309	.0067
3	.5197	.0883	-.0894	-.0640	.0183	3	.6370	.1029	-.1879	.0349	.0089
5	.6349	.1202	-.0840	-.0787	.0279	5	.7678	.1288	-.1922	.0427	.0143
7	.7480	.1524	-.0811	-.0924	.0371	7	.8771	.1835	-.2009	.0999	.0201
10	.8654	.2140	-.0997	-.1069	.0461	10	1.0292	.2630	-.2109	.1204	.0320
15	.9197	.3251	-.1490	-.1089	.0466	15	1.2027	.3940	-.2042	.1450	.0522
20	.9263	.4203	-.1669	-.1065	.0477	20	1.3454	.5845	-.2339	.1630	.0743
25	.8871	.4973	-.1633	-.1023	.0361	25	1.1932	.6546	-.2423	.1432	.0769
$M = 0.80$						$M = 1.05$					
-10	-.1564	.0443	-.1045	-.0240	-.0004	-10	-.3003	.0970	-.1031	-.0382	.0052
-7	.0148	.0244	-.1360	-.0034	-.0019	-7	-.1033	.0658	-.1435	-.0102	.0016
-5	.1299	.0236	-.1338	-.0105	-.0022	-5	.0410	.0546	-.1644	.0022	.0007
-3	.2457	.0298	-.1295	-.0238	-.0008	-3	.1866	.0529	-.1792	.0187	.0003
-2	.3011	.0370	-.1280	-.0310	-.0003	-2	.2708	.0554	-.1829	.0268	.0008
-1	.3490	.0453	-.1254	-.0376	.0016	-1	.3447	.0602	-.1883	.0356	.0010
0	.4177	.0527	-.1195	-.0340	.0044	0	.4164	.0685	-.1871	.0429	.0016
1	.4664	.0599	-.1222	-.0517	.0057	1	.4755	.0778	-.1883	.0513	.0031
2	.5106	.0793	-.1167	-.0587	.0166	2	.5495	.0807	-.1845	.0587	.0050
3	.5682	.0985	-.1121	-.0654	.0213	3	.6166	.1035	-.1893	.0666	.0069
5	.6833	.1161	-.1053	-.0799	.0325	5	.7372	.1384	-.1929	.0815	.0120
7	.8088	.1592	-.1025	-.0949	.0433	7	.8509	.1818	-.2022	.0956	.0175
10	.9150	.2127	-.1108	-.1084	.0499	10	1.0011	.2629	-.2182	.1174	.0289
15	.9475	.3208	-.1606	-.1093	.0383	15	1.2650	.4364	-.2468	.1481	.0516
20	.9534	.4283	-.1724	-.1106	.0988						
25	.9239	.5109	-.1707	-.1088	.0498						
$M = 0.90$						$M = 1.10$					
-10	-.1810	.0689	-.1426	-.0250	-.0035	-10	-.3033	.0902	-.0976	-.0389	.0028
-7	.0020	.0417	-.1633	-.0028	-.0005	-7	-.1117	.0616	-.1387	-.0150	-.0003
-5	.1455	.0371	-.1628	-.0125	-.0001	-5	.0372	.0511	-.1559	.0014	-.0010
-3	.2617	.0426	-.1670	-.0269	-.0001	-3	.1620	.0511	-.1717	.0163	-.0010
-2	.3386	.0481	-.1619	-.0343	.0007	-2	.2507	.0546	-.1753	.0247	-.0003
-1	.3946	.0557	-.1632	-.0415	.0015	-1	.3186	.0592	-.1804	.0321	-.0002
0	.4506	.0647	-.1623	-.0481	.0028	0	.3920	.0684	-.1846	.0409	-.0006
1	.4987	.0724	-.1602	-.0547	.0050	1	.4522	.0773	-.1902	.0483	-.0013
2	.5508	.0794	-.1434	-.0607	.0074	2	.5288	.0888	-.1826	.0561	-.0034
3	.6094	.0920	-.1440	-.0685	.0092	3	.5934	.1042	-.1871	.0640	-.0049
5	.7425	.1233	-.1386	-.0844	.0137	5	.7117	.1421	-.1918	.0784	-.0087
7	.8285	.1569	-.1368	-.0972	.0185	7	.8277	.1859	-.2044	.0930	.0140
10	.9454	.2216	-.1372	-.1138	.0568	10	.9722	.2692	-.2190	.1133	.0242
15	1.0652	.3522	-.1554	-.1300	.0890	15	1.2175	.4351	-.2544	.1432	.0470
20	1.0443	.4675	-.1967	-.1225	.0564						
25	.9975	.5572	-.1927	-.1193	.0651						

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TABLE 2 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

 $\frac{L}{c} = 0.06$        $\frac{C}{c} = \text{NONE}$ 

$c_s$ deg	$C_L$	$C_D$	$C_M$	$C_t$	$C_n$	$c_s$ deg	$C_L$	$C_D$	$C_M$	$C_t$	$C_n$
$M = 0.40$											
$M = 0.55$											
-10	-0.4485					-10	-0.5376	.1123	.0288	-.0642	-.0133
-7	-0.3405					-7	-0.5622	.0600	-.0008	-.0443	-.0068
-5	-0.2484					-5	-0.2871	.0390	-.0098	-.0308	-.0031
-3	-0.1620					-3	-0.1840	.0251	-.0063	-.0193	-.0002
-2	-0.1296					-2	-0.1301	.0219	-.0049	-.0123	.0007
-1	-0.0864					-1	-0.0666	.0193	-.0028	-.0054	.0022
0	-0.0378					0	-0.1443	.0155	.0028	.0011	.0022
1	.0165					1	.0523	.0181	.0042	.0089	.0026
2	.0540					2	.1078	.0219	.0098	.0146	.0024
3	.1080					3	.1649	.0257	.0112	.0212	.0024
5	.2214					5	.2680	.0406	.0112	.0346	.0033
7	.3078					7	.3822	.0647	.0008	.0292	.0009
10	.4590					10	.5376	.1132	-.0260	.0693	-.0044
15	.6426					15	.7962	.2309	-.0962	.1039	-.0201
20	.7452					20	1.0023	.3860	-.1958	.1251	-.0421
25	.7452					25	1.0309	.5132	-.2379	.1282	-.0594
$M = 0.60$											
$M = 1.00$											
-10	-0.5027	.1094	-.0024	-.0587	-.0074	-10	-0.6184	.1346	.0760	-.0723	-.0163
-7	-0.3555	.0533	-.0062	-.0391	-.0039	-7	-0.4667	.0808	.0410	-.0513	-.0090
-5	-0.2610	.0383	-.0073	-.0249	-.0007	-5	-0.3252	.0553	.0242	-.0349	-.0044
-3	-0.1638	.0244	-.0087	-.0162	.0023	-3	-0.2097	.0395	.0141	-.0221	-.0012
-2	-0.1111	.0217	-.0038	-.0101	.0029	-2	-0.1459	.0343	.0107	-.0151	-.0004
-1	-0.0667	.0139	-.0024	-.0047	.0036	-1	-0.0820	.0298	.0074	-.0081	.0014
0	-0.0111	.0139	.0013	.0020	.0036	0	-0.1222	.0277	.0053	0.0000	.0018
1	.0417	.0139	.0000	.0094	.0029	1	.0547	.0298	.0013	.0085	.0021
3	.0861	.0139	.0013	.0155	.0036	2	.1170	.0322	.0007	.0147	.0019
5	.1361	.0139	.0038	.0202	.0042	3	.1914	.0395	-.0047	.0240	.0014
7	.2444	.0289	.0024	.0303	.0048	5	.3054	.0559	-.0162	.0387	.0014
10	.3527	.0422	.0013	.0438	.0042	7	.4500	.0820	-.0329	.0559	-.0009
15	.5110	.0822	-.0147	.0647	.0007	10	.6169	.1367	-.0699	.0763	-.0074
20	.7109	.2005	-.1069	.0993	-.0097	15	.9086	.2683	-.1506	.1147	-.0242
25	.7665	.3144	-.1782	.0964	-.0236	20	1.1456	.4409	-.2178	.1457	-.0484
	.7637	.3905	-.1806	.0944	-.0333	25	1.2034	.5977	-.2763	.1505	-.0684
$M = 0.80$											
$M = 1.05$											
-10	-0.5343	.1016	-.0125	-.0616	-.0127	-10	-0.6217	.1363	.0942	-.0705	-.0171
-7	-0.3875	.0500	-.0116	-.0438	-.0009	-7	-0.4539	.0826	.0569	-.0514	-.0100
-5	-0.2765	.0297	-.0099	-.0297	-.0026	-5	-0.3371	.0575	.0342	-.0365	-.0054
-3	-0.1750	.0184	-.0092	-.0402	-.0005	-3	-0.2101	.0423	.0207	-.0223	-.0012
-2	-0.1204	.0158	-.0066	-.0110	.0005	-2	-0.1445	.0359	.0142	-.0149	-.0004
-1	-0.0677	.0120	-.0033	-.0046	.0011	-1	-0.0744	.0339	.0103	-.0071	-.0007
0	-0.0075	.0128	-.0026	-.0023	.0020	0	-0.0558	.0324	.0051	.0011	.0014
1	.0489	.0128	.0009	.0082	.0029	1	.0628	.0330	.0007	.0088	.0014
2	.0978	.0214	.0026	.0137	.0029	2	.1255	.0359	-.0046	.0163	.0014
3	.1524	.0184	.0059	.0201	.0033	3	.1897	.0432	-.0103	.0234	.0010
5	.2558	.0278	-.0092	.0320	.0029	5	.3123	.0674	-.0239	.0379	.0005
7	.3819	.0482	.0066	.0479	.0018	7	.4520	.0846	-.0426	.0531	-.0012
10	.5362	.0952	-.0132	.0676	-.0022	10	.6144	.1392	-.0781	.0744	-.0077
15	.7224	.2111	-.0924	.0913	-.0184	15	.9135	.2726	-.1608	.1116	-.0248
20	.7713	.3145	-.1615	.0959	-.0304	20	1.1587	.4477	-.2299	.1438	-.0496
25	.7976	.4116	-.1806	.1009	-.0438	25	1.2725	.6272	-.2853	.1594	-.0725
$M = 0.90$											
$M = 1.10$											
-10	-0.5448	.1050	.0259	-.0631	-.0138	-10	-0.6119	.1401	.0937	-.0681	-.0207
-7	-0.3999	.0550	.0008	.0445	-.0062	-7	-0.4491	.0848	.0552	-.0454	-.0113
-5	-0.2966	.0320	-.0096	.0320	-.0037	-5	-0.3284	.0573	.0354	-.0328	-.0060
-3	-0.1816	.0190	-.0088	.0182	-.0004	-3	-0.2021	.0415	.0223	-.0221	-.0012
-2	-0.1233	.0163	-.0073	.0121	.0006	-2	-0.1417	.0351	.0162	-.0150	-.0015
-1	-0.0616	.0147	-.0029	-.0049	.0023	-1	-0.0758	.0312	.0112	-.0068	-.0005
0	-0.0133	.0130	.0000	.0012	.0025	0	-0.1226	.0298	.0068	0.0000	.0012
1	.0500	.0180	.0029	.0081	.0025	1	.0589	.0312	.0007	.0072	.0008
2	.1066	.0180	.0067	.0150	.0033	2	.1263	.0345	-.0044	.0153	.0008
3	.1649	.0230	.0096	.0214	.0033	3	.1881	.0393	-.0106	.0221	.0005
5	.2732	.0353	.0103	.0340	.0031	5	.3073	.0567	-.0243	.0361	.0005
7	.3899	.0556	.0023	.0485	.0016	7	.4522	.0822	-.0429	.0511	-.0010
10	.5431	.1899	-.0199	.0691	-.0033	10	.6035	.1353	-.0800	.0715	-.0067
15	.7631	.2173	-.0766	.0983	-.0171	15	.8897	.2610	-.1590	.1063	-.0224
20	.8697	.3522	-.1798	.1072	-.0349	20	1.1367	.4348	-.2291	.1373	-.0485
25	.9230	.4702	-.2094	.1132	-.0510	25	1.2658	.6169	-.2800	.1550	-.0689

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TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

 $\frac{t}{c} = 0.06$        $\frac{C_f}{c} = 0.20$ 

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$
$M = 0.40$											
$M = 0.60$											
$M = 0.80$											
$M = 0.90$											
-10	-.1252					-10	-.3692	.0965	-.0474	-.0407	-.0095
-7	-.0653					-7	-.2254	.0559	-.0721	-.0233	-.0039
-5	.0109					-5	-.0959	.0377	-.0969	-.0070	-.0004
-3	.1034					-3	+.0479	.0284	-.1132	.0097	.0007
-2	.1469					-2	-.1247	.0297	-.1180	.0175	.0007
-1	.1905					-1	+.1838	.0323	-.1181	.0240	.0007
0	.2394					0	+.2397	.0355	-.1181	.0307	.0005
1	.2830					1	+.3037	.0432	-.1216	.0372	-.0002
2	.3156					2	+.3644	.0695	-.1195	.0446	-.0004
3	.3646					3	+.4267	.0588	-.1286	.0520	-.0015
5	.4408					5	+.5354	.0809	-.1358	.0648	-.0032
7	.5387					7	+.6457	.1170	-.1464	.0788	-.0071
10	.6384					10	+.7895	.1768	-.1640	.0982	-.0143
15	.7945					15	+.9653	.2979	-.1903	.1203	-.0301
20	.8217					20	1.0708	.4481	-.2552	.1280	-.0487
25	.8108					25	1.0772	.5706	-.2821	.1268	-.0643
$M = 0.95$											
$M = 1.00$											
$M = 1.05$											
$M = 1.10$											

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TABLE 2-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Continued

$$\frac{c}{c_0} = 0.06 \quad \frac{c_e}{c_0} = 0.30$$

$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_l$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_l$	$c_n$
$M = 0.40$						$M = 0.55$					
-10	-0.1089					-10	-0.2939	.0936	-0.0834	-0.0349	-0.0076
-7	-0.0109					-7	-0.1342	.0549	-0.1131	-0.0140	-0.0024
-5	.0708					-5	-0.0096	.0425	-0.1301	.0000	.0002
-3	.1634					-3	-0.1326	.0383	-0.1427	.0155	.0015
-2	.2015					-2	-0.2012	.0425	-0.1497	.0233	.0009
-1	.2559					-1	.2469	.0441	-0.1441	.0314	.0005
0	.2995					0	.3306	.0502	-0.1469	.0384	.0000
1	.3321					1	.3929	.0588	-0.1541	.0450	-0.0005
2	.3866					2	.4536	.0674	-0.1541	.0523	-0.0013
3	.4247					3	.5175	.0776	-0.1526	.0593	-0.0022
5	.5009					5	.6213	.1038	-0.1541	.0717	-0.0048
7	.6044					7	.7267	.1437	-0.1681	.0853	-0.0096
10	.7405					10	.8752	.2035	-0.1823	.1027	-0.0180
15	.8494					15	1.0541	.3348	-0.2025	.1279	-0.0357
20	.8715					20	1.1627	.4932	-0.2626	.1384	-0.0561
25	.8385					25	1.1468	.6191	-0.2918	.1318	-0.0719
$M = 0.60$						$M = 1.00$					
-10	-0.1540	.0661	-0.0793	-0.0204	-0.0026	-10	-0.3549	.1046	-0.0582	-0.0408	-0.0098
-7	-0.0224	.0302	-0.1015	-0.0027	-0.0010	-7	-0.1897	.0591	-0.0880	-0.0212	-0.0041
-5	.0812	.0218	-0.0966	.0102	-0.0003	-5	-0.0642	.0557	-0.1137	-0.0074	-0.0013
-3	.1708	.0190	-0.1028	.0204	.0007	-3	.0918	.0480	-0.1340	.0111	.0000
-2	.2240	.0207	-0.1004	.0265	.0007	-2	.1683	.0480	-0.1442	.0197	.0004
-1	.2744	.0235	-0.0916	.0319	.0003	-1	.2432	.0541	-0.1516	.0271	.0004
0	.3332	.0274	-0.0930	.0374	.0003	0	.3090	.0578	-0.1591	.0345	-0.0004
1	.3724	.0342	-0.1028	.0428	-0.0003	1	.3793	.0661	-0.1624	.0423	-0.0005
2	.4173	.0370	-0.0916	.0476	-0.0007	2	.4436	.0743	-0.1651	.0497	-0.0014
3	.4649	.0498	-0.0966	.0557	-0.0007	3	.5124	.0872	-0.1746	.0575	-0.0023
5	.5489	.0672	-0.0966	.0632	-0.0010	5	.6134	.1120	-0.1848	.0705	-0.0048
7	.6581	.0952	-0.0842	.0748	-0.0039	7	.7403	.1557	-0.1956	.0854	-0.0101
10	.7925	.1557	-0.1004	.0917	-0.0108	10	.8594	.2233	-0.2220	.1040	-0.0190
15	.8709	.2711	-0.1709	.1019	-0.0248	15	1.1380	.3723	-0.2626	.1332	-0.0397
20	.8989	.3842	-0.2229	.1054	-0.0381	20	1.3124	.5552	-0.2964	.1578	-0.0658
25	.8625	.4598	-0.2267	.0972	-0.0476	25	1.3277	.7058	-0.3357	.1578	-0.1039
$M = 0.80$						$M = 1.05$					
-10	-0.1821	.0577	-0.1007	-0.0230	-0.0024	-10	-0.3776	.1070	-0.0442	-0.0410	-0.0099
-7	-0.0436	.0288	-0.1141	-0.0046	-0.0005	-7	-0.1925	.0702	-0.0799	-0.0217	-0.0051
-5	.0645	.0243	-0.1107	.0078	.0011	-5	-0.0749	.0570	-0.1059	-0.0089	-0.0019
-3	.1707	.0250	-0.1174	.0203	.0016	-3	.0690	.0491	-0.1234	.0075	-0.0005
-2	.2295	.0269	-0.1158	.0272	.0006	-2	.1454	.0500	-0.1339	.0160	.0004
-1	.2864	.0300	-0.1074	.0336	.0006	-1	.1998	.0447	-0.1417	.0232	.0002
0	.3357	.0364	-0.1115	.0387	.0000	0	.2733	.0549	-0.1403	.0125	-0.0004
1	.3982	.0440	-0.1200	.0460	-0.0005	1	.3438	.0643	-0.1547	.0374	-0.0021
2	.4438	.0504	-0.1167	.0511	-0.0006	2	.4084	.0729	-0.1611	.0453	-0.0021
3	.4874	.0588	-0.1141	.0562	-0.0016	3	.4745	.0861	-0.1716	.0531	-0.0031
5	.5860	.0793	-0.1115	.0677	-0.0029	5	.5891	.1134	-0.1859	.0663	-0.0043
7	.7055	.1130	-0.1098	.0819	-0.0071	7	.7067	.1516	-0.2062	.0810	-0.0106
10	.8287	.1726	-0.1200	.0985	-0.0146	10	.8785	.2210	-0.2230	.1009	-0.0197
15	.9254	.2852	-0.1602	.1105	-0.0293	15	1.1077	.3743	-0.2639	.1302	-0.0400
20	.9368	.4020	-0.2206	.1086	-0.0437	20	1.2870	.5577	-0.3106	.1533	-0.0642
25	.9254	.4980	-0.2241	.1082	-0.0556	25	1.3927	.7631	-0.3509	.1676	-0.0912
$M = 0.90$						$M = 1.10$					
-10	-0.2493	.0773	-0.1007	-0.0305	-0.0041	-10	-0.3589	.1085	-0.0413	-0.0405	-0.0120
-7	-0.0920	.0416	-0.1206	-0.0106	-0.0010	-7	-0.1808	.0687	-0.0818	-0.0209	-0.0064
-5	.0234	.0298	-0.1281	.0041	.0014	-5	-0.0650	.0543	-0.1031	-0.0075	-0.0036
-3	.1474	.0298	-0.1288	.0175	.0016	-3	.0692	.0480	-0.1206	.0075	-0.0012
-2	.2109	.0338	-0.1281	.0240	.0014	-2	.1356	.0480	-0.1300	.0148	-0.0020
-1	.2661	.0378	-0.1296	.0305	.0006	-1	.2006	.0494	-0.1312	.0223	-0.0026
0	.3247	.0445	-0.1340	.0370	.0000	0	.2600	.0548	-0.1456	.0288	-0.0033
1	.3835	.0519	-0.1317	.0443	.0004	1	.3292	.0633	-0.1531	.0363	-0.0033
2	.4505	.0609	-0.1347	.0516	.0014	2	.3843	.0709	-0.1549	.0429	-0.0051
3	.5087	.0733	-0.1285	.0589	.0019	3	.4479	.0814	-0.1643	.0504	-0.0061
5	.6091	.0954	-0.1517	.0707	.0043	5	.5651	.1089	-0.1788	.0634	-0.0077
7	.7146	.1332	-0.1428	.0841	.0064	7	.6909	.1486	-0.1894	.0778	-0.0137
10	.8434	.1901	-0.1466	.0999	-0.0160	10	.8449	.2139	-0.2175	.0957	-0.0222
15	1.0041	.3119	-0.1762	.1219	-0.0323	15	1.0709	.3628	-0.2581	.1258	-0.0414
20	1.0242	.4426	-0.2443	.1198	-0.0487	20	1.2546	.5462	-0.3038	.1464	-0.0660
25	1.0242	.5529	-0.2613	.1190	-0.0636	25	1.3676	.7533	-0.3476	.1612	-0.0936

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TABLE 2.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 2 MODEL - Concluded

$$\frac{S}{c} = 0.06 \quad \frac{C_f}{c} = 0.40$$

$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_I$	$c_n$
$M = 0.40$						$M = 0.95$					
-10	-1077					-10	-2504	.0785	-0.0974	-0.0290	-0.0052
-7	.0000					-7	-0829	.0456	-0.1256	-0.0085	-0.0026
-5	.0916					-5	.0447	.0342	-0.1411	-0.0070	-0.0032
-3	.1886					-3	.1643	.0345	-0.1447	.0197	.0034
-2	.2370					-2	.2265	.0376	-0.1481	.0271	.0037
-1	.2694					-1	.2823	.0440	-0.1503	.0333	.0043
0	.3235					0	.3429	.0501	-0.1581	.0399	.0050
1	.3609					1	.4099	.0581	-0.1517	.0472	.0057
2	.4040					2	.4626	.0689	-0.1539	.0538	.0067
3	.4418					3	.5343	.0817	-0.1573	.0616	.0080
5	.5387					5	.6300	.1097	-0.1659	.0743	.0113
7	.6411					7	.7369	.1474	-0.1743	.0879	.0152
10	.7650					10	.8773	.2166	-0.1905	.1045	.0232
15	.8515					15	1.0591	.3452	-0.2047	.1289	.0397
20	.8727					20	1.1261	.4941	-0.2680	.1347	.0579
25	.8189					25	1.1229	.6214	-0.2964	.1312	.0734
$M = 0.60$						$M = 1.00$					
-10	-1448	.0479	-.0539	-.0176	-.0039	-10	-3102	.0947	-.0730	-.0360	-.0076
-7	.0139	.0206	-.0876	-.0027	-.0039	-7	-1529	.0617	-.1109	-.0148	-.0041
-5	.0919	.0011	-.0900	.0095	-.0036	-5	.0000	.0480	-.1339	.0011	-.0048
-3	.1866	.0167	-.0911	.0196	-.0032	-3	.1406	.0452	-.1521	.0178	-.0050
-2	.2423	.0189	-.0925	.0284	-.0023	-2	.2185	.0452	-.1575	.0260	-.0059
-1	.2785	.0189	-.0862	.0338	-.0023	-1	.2781	.0526	-.1649	.0328	-.0068
0	.3342	.0273	-.0925	.0379	-.0029	0	.3515	.0571	-.1656	.0397	-.0076
1	.3844	.0340	-.0925	.0426	-.0032	1	.4233	.0660	-.1676	.0471	-.0083
2	.4317	.0412	-.0887	.0500	-.0036	2	.4691	.0752	-.1691	.0542	-.0098
3	.4735	.0479	-.0900	.0534	-.0045	3	.5348	.0902	-.1744	.0620	-.0114
5	.5905	.0668	-.0925	.0656	-.0065	5	.6448	.1201	-.1828	.0742	-.0151
7	.6824	.1003	-.0925	.0784	-.0094	7	.7488	.1608	-.1961	.0875	-.0197
10	.8025	.1616	-.1110	.0940	-.0104	10	.8924	.2329	-.2197	.1057	-.0291
15	.8468	.2808	-.1861	.1001	-.0239	15	1.1094	.3802	-.2568	.1339	-.0316
20	.8746	.3972	-.2286	.1014	-.0376	20	1.2452	.5486	-.2955	.1539	-.0725
25	.8301	.4657	-.2317	.0919	-.0457	25	1.2469	.6913	-.3380	.1498	-.0885
$M = 0.80$						$M = 1.05$					
-10	-1717	.0483	-.1002	-.0197	-.0033	-10	-3155	.0925	-.0701	-.0360	-.0132
-7	-.0245	.0249	-.1109	-.0019	-.0026	-7	-1468	.0658	-.1024	-.0160	-.0057
-5	.0830	.0215	-.1127	.0096	-.0024	-5	-.0220	.0505	-.1266	-.0021	-.0065
-3	.1866	.0249	-.1127	.0220	-.0024	-3	.1248	.0476	-.1446	.0142	-.0054
-2	.2434	.0298	-.1127	.0302	-.0026	-2	.1981	.0505	-.1582	.0221	-.0060
-1	.3018	.0351	-.1102	.0357	-.0033	-1	.2495	.1993	-.1571	.0278	-.0072
0	.3508	.0419	-.1127	.0417	-.0035	0	.3185	.0619	-.1624	.0353	-.0077
1	.4055	.0483	-.1109	.0476	-.0044	1	.3875	.2722	-.1694	.0435	-.0092
2	.4489	.0573	-.1102	.0540	-.0057	2	.4506	.0793	-.1720	.0502	-.0101
3	.5036	.0668	-.1059	.0584	-.0059	3	.5019	.0910	-.1766	.0570	-.0118
5	.6017	.0909	-.1043	.0705	-.0072	5	.6238	.1227	-.1870	.0712	-.0152
7	.7168	.1279	-.1068	.0851	-.0103	7	.7456	.1632	-.2012	.0848	-.0203
10	.8394	.1894	-.1160	.0998	-.0176	10	.9011	.2359	-.2259	.1019	-.0247
15	.8696	.2950	-.1820	.1057	-.0303	15	1.1242	.3863	-.2695	.1322	-.0505
20	.9092	.4082	-.2187	.1062	-.0456	20	1.2451	.5430	-.3052	.1528	-.0737
25	.8884	.4991	-.2270	.1044	-.0572	25	1.3649	.3666	-.3506	.1642	-.0986
$M = 0.90$						$M = 1.10$					
-10	-.2204	.0738	-.1049	-.0239	-.0045	-10	-3105	.0957	-.0656	-.0353	-.0143
-7	-.0584	.0417	-.1256	-.0057	-.0037	-7	-1440	.0675	-.1012	-.0168	-.0095
-5	.0635	.0344	-.1329	.0085	-.0033	-5	-.0183	.0556	-.1249	-.0027	-.0074
-3	.1753	.0354	-.1329	.0215	-.0033	-3	.1059	.0556	-.1387	.0117	-.0074
-2	.2371	.0401	-.1323	.0288	-.0041	-2	.1821	.0590	-.1468	.0192	-.0075
-1	.2906	.0451	-.1315	.0361	-.0050	-1	.2399	.0624	-.1498	.0260	-.0085
0	.3473	.0534	-.1293	.0422	-.0058	0	.3063	.0680	-.1580	.0339	-.0093
1	.4091	.0615	-.1323	.0478	-.0060	1	.3613	.0762	-.1617	.0408	-.0105
2	.4592	.0691	-.1293	.0551	-.0074	2	.4319	.0867	-.1680	.0480	-.0115
3	.5160	.0805	-.1270	.0624	-.0089	3	.4998	.0985	-.1735	.0552	-.0131
5	.6265	.1052	-.1293	.0746	-.0101	5	.6041	.1270	-.1842	.0682	-.0164
7	.7214	.1429	-.1382	.0875	-.0136	7	.7198	.1694	-.1999	.0819	-.0217
10	.8617	.2054	-.1478	.1046	-.0217	10	.8609	.2340	-.2186	.0980	-.0299
15	1.0153	.3327	-.1832	.1265	-.0388	15	1.0727	.3706	-.2591	.1254	-.0489
20	1.0020	.4502	-.2453	.1200	-.0532	20	1.2279	.5428	-.3122	.1459	-.0724
25	.9886	.5568	-.2586	.1180	-.0664	25	1.3267	.7218	-.3435	.1579	-.0959

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TABLE 3. - THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL

$$\frac{k}{c} = 0.04 \quad \frac{c_x}{c} = \text{NONE}$$

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
-10	-.5838					-10	-.7947	.1667	.0846	-.0958	.0197
-7	-.4338					-7	-.6088	.0971	.0540	-.0726	.0014
-5	-.3191					-5	-.4410	.0546	.0200	-.0509	.0045
-3	-.2104					-3	-.2879	.0304	.0049	-.0326	.0035
-2	-.1544					-2	-.2057	.0211	.0006	-.0226	.0026
-1	-.0956					-1	-.1218	.0158	-.0026	-.0133	.0021
0	-.0279					0	-.0296	.0129	.0028	-.0030	.0019
1	.0176					1	.0609	.0129	.0097	.0073	.0025
2	.0691					2	.1431	.0162	.0119	.0163	.0030
3	.1250					3	.2287	.0239	.0108	.0270	.0043
5	.2323					5	.4048	.0453	-.0085	.0479	.0079
7	.3426					7	.5709	.0810	-.0418	.0689	.0132
10	.5382					10	.7848	.1505	-.0769	.0955	.0238
15	.6661					15	1.0530	.2930	-.1140	.1305	.0458
20	.6941										
25	.6764										
$M = 0.60$											
-10	-.6694	.1170	.0142	-.0804	.0156	-10	-.7981	.1643	.1063	-.0918	.0190
-7	-.5137	.0601	-.0208	-.0480	.0082	-7	-.5884	.0962	.0623	-.0558	.0109
-5	-.3711	.0322	-.0196	-.0408	.0047	-5	-.4382	.0604	.0377	-.0491	.0071
-3	-.2474	.0204	-.0084	-.0263	.0031	-3	-.2911	.0393	.0234	-.0317	.0043
-2	-.1746	.0162	-.0066	-.0183	.0024	-2	-.2034	.0297	.0122	-.0218	.0033
-1	-.1019	.0140	-.0006	-.0115	.0018	-1	-.1189	.0231	.0025	-.0120	.0027
0	-.0335	.0122	-.0035	-.0028	.0017	0	-.0219	.0200	-.0019	-.0022	.0025
1	.0291	.0122	-.0007	-.0053	.0019	1	.0782	.0208	-.0058	.0085	.0029
2	.0975	.0162	-.0078	-.0128	.0026	2	.1721	.0239	-.0095	.0193	.0039
3	.1673	.0218	.0123	.0208	.0037	3	.2504	.0315	-.0170	.0291	.0050
5	.2983	.0373	.0235	.0364	.0055	5	.4163	.0523	-.0368	.0481	.0084
7	.4264	.0633	.0223	.0518	.0096	7	.5571	.0816	-.0555	.0639	.0127
10	.6039	.1224	-.0071	.0762	.0181	10	.7668	.1451	-.0956	.0889	.0222
15	.7334	.2269	-.0691	.0963	.0327	15	1.0485	.2901	-.1314	.1272	.0452
20	.7421	.3070	-.0860	.0971	.0441						
25	.7480	.3922	-.0869	.0971	.0539						
$M = 0.80$											
-10	-.7036	.1297	-.0039	-.0835	.0162	-10	-.7557	.1550	.0966	-.0865	.0174
-7	-.5534	.0695	-.0123	-.0826	.0186	-7	-.5592	.0911	.0601	-.0630	.0100
-5	-.4131	.0384	-.0165	-.0456	.0055	-5	-.4232	.0583	.0390	-.0468	.0060
-3	-.2727	.0223	-.0098	-.0294	.0034	-3	-.2811	.0376	.0226	-.0303	.0035
-2	-.1878	.0160	-.0066	-.0200	.0026	-2	-.1965	.0286	.0125	-.0208	.0025
-1	-.1087	.0131	-.0028	-.0113	.0020	-1	-.1058	.0230	.0023	-.0110	.0018
0	-.0395	.0107	-.0033	-.0038	.0018	0	-.0212	.0212	-.0018	-.0040	.0016
1	.0296	.0102	-.0063	-.0047	.0021	1	.0725	.0215	-.0057	.0089	.0023
2	.1127	.0141	-.0127	.0142	.0030	2	.1602	.0260	-.0172	.0180	.0032
3	.1917	.0190	-.0137	.0227	.0036	3	.2418	.0335	-.0190	.0275	.0043
5	.3498	.0340	-.0269	.0407	.0067	5	.4020	.0535	-.0386	.0456	.0072
7	.4961	.0613	-.0226	.0584	.0111	7	.5350	.0825	-.0554	.0618	.0118
10	.6818	.1186	-.0068	.0823	.0192	10	.7254	.1420	-.0912	.0853	.0210
15	.7609	.2221	-.0697	.0987	.0341						
20	.7767	.3091	-.0925	.0991	.0438						
25	.7905	.4004	-.0985	.1007	.0576						
$M = 0.90$											
-10	-.7338	.1444	.0318	-.0903	.0177	-10	-.7320	.1512	.0978	-.0838	.0162
-7	-.5755	.0823	.0112	-.0680	.0099	-7	-.5417	.0897	.0561	-.0616	.0093
-5	-.4358	.0458	-.0037	-.0494	.0057	-5	-.4246	.0583	.0423	-.0456	.0054
-3	-.2963	.0244	-.0101	-.0324	.0033	-3	-.2782	.0385	.0254	-.0296	.0034
-2	-.2004	.0159	-.0106	-.0215	.0026	-2	-.1903	.0295	.0146	-.0201	.0026
-1	-.1220	.0116	-.0056	-.0130	.0020	-1	-.1083	.0242	.0069	-.0107	.0019
0	-.0366	.0094	-.0019	-.0039	.0018	0	-.0234	.0220	.0010	-.0278	.0017
1	.0488	.0085	-.0095	.0064	.0021	1	.0673	.0216	-.0080	.0083	.0023
2	.1290	.0116	.0159	.0148	.0028	2	.1464	.0266	-.0125	.0169	.0029
3	.2266	.0172	.0179	.0254	.0038	3	.2284	.0342	-.0194	.0267	.0038
5	.4009	.0386	.0078	.0469	.0071	5	.3806	.0558	-.0373	.0438	.0063
7	.5438	.0707	-.0094	.0659	.0117	7	.5124	.0853	-.0548	.0598	.0101
10	.7146	.1294	-.0273	.0896	.0206	10	.6881	.1465	-.0868	.0832	.0176
15	.8245	.2418	-.0846	.1054	.0354						
20	.8715	.3403	-.1196	.1082	.0481						
25	.8820	.4406	-.1259	.1089	.0590						
$M = 1.10$											
-10	-.7338	.1444	.0318	-.0903	.0177	-10	-.7320	.1512	.0978	-.0838	.0162
-7	-.5755	.0823	.0112	-.0680	.0099	-7	-.5417	.0897	.0561	-.0616	.0093
-5	-.4358	.0458	-.0037	-.0494	.0057	-5	-.4246	.0583	.0423	-.0456	.0054
-3	-.2963	.0244	-.0101	-.0324	.0033	-3	-.2782	.0385	.0254	-.0296	.0034
-2	-.2004	.0159	-.0106	-.0215	.0026	-2	-.1903	.0295	.0146	-.0201	.0026
-1	-.1220	.0116	-.0056	-.0130	.0020	-1	-.1083	.0242	.0069	-.0107	.0019
0	-.0366	.0094	-.0019	-.0039	.0018	0	-.0234	.0220	.0010	-.0278	.0017
1	.0488	.0085	-.0095	.0064	.0021	1	.0673	.0216	-.0080	.0083	.0023
2	.1290	.0116	.0159	.0148	.0028	2	.1464	.0266	-.0125	.0169	.0029
3	.2266	.0172	.0179	.0254	.0038	3	.2284	.0342	-.0194	.0267	.0038
5	.4009	.0386	.0078	.0469	.0071	5	.3806	.0558	-.0373	.0438	.0063
7	.5438	.0707	-.0094	.0659	.0117	7	.5124	.0853	-.0548	.0598	.0101
10	.7146	.1294	-.0273	.0896	.0206	10	.6881	.1465	-.0868	.0832	.0176
15	.8245	.2418	-.0846	.1054	.0354						
20	.8715	.3403	-.1196	.1082	.0481						
25	.8820	.4406	-.1259	.1089	.0590						

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TABLE 3.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

 $\frac{L}{c} = 0.04$        $\frac{C_f}{c} = 0.10$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$
$M = 0.40$										$M = 0.55$	
-10	.3945					-10	.6736	.1520	.0242	.0797	.0193
-7	.2221					-7	.4643	.0854	-.0214	-.0532	.0113
-5	.1125					-5	.2894	.0515	-.0593	-.0317	.0071
-3	.0132					-3	.1341	.0342	-.0850	-.0127	.0051
-2	.0409					-2	.0523	.0285	-.0924	-.0020	.0048
-1	.0906					-1	.0589	.0253	-.1109	.0101	.0044
0	.1476					0	.1602	.0314	-.1137	.0212	.0052
1	.1973					1	.2420	.0369	-.1185	.0306	.0063
2	.2440					2	.3544	.0406	-.1179	.0435	.0075
3	.2878					3	.4381	.0522	-.1250	.0534	.0094
5	.3945					5	.5984	.0836	-.1377	.0716	.0138
7	.5085					7	.7291	.1206	-.1475	.0886	.0191
10	.6765										
15	.7978										
20	.7978										
25	.7598										
$M = 0.60$										$M = 1.00$	
-10	.4583	.0828	-.0647	-.0544	.0126	-10	.6488	.1484	.0366	-.0134	.0179
-7	.2776	.0437	-.0958	-.0310	.0073	-7	.4621	.0899	-.0078	-.0515	.0114
-5	.1403	.0288	-.0976	-.0135	.0048	-5	.3112	.0604	-.0365	-.0334	.0082
-3	.0174	.0253	-.0845	.0004	.0040	-3	.1774	.0443	-.0532	-.0176	.0063
-2	.0549	.0253	-.0867	.0089	.0040	-2	.0918	.0379	-.0654	-.0074	.0057
-1	.1128	.0278	-.0832	.0158	.0042	-1	.0109	.0356	-.0694	.0020	.0052
0	.1807	.0348	-.0764	.0284	.0048	0	.0794	.0363	-.0792	.0115	.0054
1	.2400	.0431	-.0742	.0314	.0059	1	.1774	.0387	-.0888	.0233	.0063
2	.3181	.0489	-.0719	.0399	.0074	2	.2758	.0443	-.0981	.0338	.0072
3	.3803	.0562	-.0668	.0472	.0087	3	.3672	.0528	-.1109	.0449	.0087
5	.5176	.0814	-.0527	.0639	.0114	5	.5493	.0815	-.1356	.0658	.0125
7	.6405	.1160	-.0497	.0798	.0163	7	.6693	.1167	-.1518	.0828	.0173
10	.7895	.1835	-.0756	.1012	.0255						
15	.8502	.2880	-.1271	.1129	.0391						
20	.8473	.3741	-.1383	.1106	.0496						
25	.8328	.4594	-.1413	.1100	.0601						
$M = 0.80$										$M = 1.05$	
-10	.5045	.0939	-.0769	-.0619	.0128	-10	.6371	.1434	.0351	-.0760	.0156
-7	.3278	.0488	-.0946	-.0381	.0071	-7	.4478	.0890	-.0071	-.0508	.0089
-5	.1845	.0273	-.1016	-.0203	.0049	-5	.3186	.0609	-.0320	-.0347	.0058
-3	.0394	.0200	-.0947	-.0098	.0038	-3	.1863	.0462	-.0456	-.0182	.0040
-2	.0354	.0188	-.0944	-.0056	.0036	-2	.1037	.0403	-.0580	-.0082	.0040
-1	.1139	.0205	-.0962	.0145	.0040	-1	.0120	.0373	-.0645	.0009	.0027
0	.2022	.0278	-.0864	.0244	.0047	0	.0796	.0388	-.0731	.0116	.0036
1	.2670	.0296	-.0912	.0332	.0056	1	.1548	.0406	-.0808	.0207	.0040
2	.3494	.0382	-.0849	.0427	.0068	2	.2449	.0466	-.0878	.0307	.0053
3	.4378	.0478	-.0789	.0522	.0080	3	.3351	.0565	-.0981	.0410	.0068
5	.5791	.0715	-.0715	.0695	.0121	5	.5049	.0842	-.1246	.0617	.0105
7	.7185	.1062	-.0718	.0870	.0174	7	.6446	.1186	-.1473	.0805	.0149
10	.8598	.1704	-.0769	.1096	.0261						
15	.8520	.2733	-.1409	.1112	.0375						
$M = 0.90$										$M = 1.10$	
-10	.5787	.1206	-.0358	-.0708	.0156	-10	.6145	.1389	.0979	-.0736	.0136
-7	.3846	.0651	-.0707	-.0449	.0086	-7	.4470	.0870	.0542	-.0507	.0075
-5	.2426	.0375	-.0863	-.0259	.0057	-5	.3174	.0594	.0244	-.0336	.0044
-3	.0866	.0243	-.0993	-.0082	.0043	-3	.1849	.0459	.0019	-.0183	.0027
-2	.0055	.0204	-.1031	.0028	.0039	-2	.0946	.0412	-.0132	-.0085	.0020
-1	.0918	.0204	-.1057	.0128	.0042	-1	.0291	.0369	-.0264	.0000	.0020
0	.1836	.0260	-.1001	.0239	.0048	0	.0612	.0394	-.0415	.0097	.0021
1	.2824	.0302	-.1066	.0347	.0059	1	.1427	.0419	-.0570	.0200	.0039
2	.3794	.0388	-.1013	.0456	.0074	2	.2359	.0465	-.0708	.0292	.0042
3	.4608	.0464	-.1027	.0548	.0090	3	.3203	.0566	-.0881	.0395	.0054
5	.6205	.0771	-.1087	.0741	.0135	5	.4732	.0852	-.1184	.0583	.0082
7	.7457	.1163	-.1232	.0922	.0191	7	.6116	.1193	-.1463	.0751	.0120
10	.9390	.1856	-.1353	.1156	.0291						

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TABLE 3 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

$$\frac{L}{c} = 0.04 \quad \frac{C_L}{c} = 0.20$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$
$M = 0.40$											
$M = 0.60$											
$M = 0.80$											
$M = 0.90$											
-10	-.3057					-10	-.5765	.1362	-.0406	-.0702	.0144
-7	-.1254					-7	-.3623	.0770	.2230	-.0399	.0069
-5	-.0029					-5	-.1809	.0501	-.1155	-.0180	.0038
-3	.1081					-3	.0129	.0342	-.1487	-.0049	.0028
-2	.1629					-2	.1179	.0342	-.1631	-.0163	.0030
-1	.2206					-1	.2228	.0354	-.1689	-.0284	.0038
0	.2783					0	.3246	.0409	-.1713	-.0399	.0048
1	.3258					1	.4231	.0480	-.1716	-.0596	.0063
2	.4023					2	.5167	.0580	-.1707	-.0611	.0083
3	.4369					3	.6071	.0706	-.1712	-.0725	.0106
5	.5911					5	.7638	.1056	-.1845	-.0905	.0159
7	.6762										
10	.8074										
15	.8708										
20	.8449										
25	.8160										
$M = 0.95$											
-10	-.5945					-10	-.5945	.1420	.2813	-.0699	.0146
-7	-.3871					-7	-.3871	.0873	-.0571	-.0435	.0079
-5	-.2289					-5	-.2289	.0598	-.0848	-.0239	.0050
-3	-.0615					-3	-.0615	.0446	-.1145	-.0037	.0017
-2	.0307					-2	.0307	.0408	-.1271	-.0078	.0034
-1	.1352					-1	.1352	.0400	-.1419	-.0180	.0034
0	.2842					0	.2842	.0400	-.1645	-.0357	.0045
1	.3656					1	.3656	.0465	-.1668	-.0451	.0058
2	.4701					2	.4701	.0571	-.1698	-.0566	.0077
3	.5531					3	.5531	.0691	-.1747	-.0665	.0098
5	.7082					5	.7082	.1024	-.1858	-.0848	.0147
$M = 1.00$											
-10	-.5945					-10	-.5945	.1420	.2813	-.0699	.0146
-7	-.3871					-7	-.3871	.0873	-.0571	-.0435	.0079
-5	-.2289					-5	-.2289	.0598	-.0848	-.0239	.0050
-3	-.0615					-3	-.0615	.0446	-.1145	-.0037	.0017
-2	.0307					-2	.0307	.0408	-.1271	-.0078	.0034
-1	.1352					-1	.1352	.0400	-.1419	-.0180	.0034
0	.2842					0	.2842	.0400	-.1645	-.0357	.0045
1	.3656					1	.3656	.0465	-.1668	-.0451	.0058
2	.4701					2	.4701	.0571	-.1698	-.0566	.0077
3	.5531					3	.5531	.0691	-.1747	-.0665	.0098
5	.7082					5	.7082	.1024	-.1858	-.0848	.0147
$M = 1.05$											
-10	-.5945					-10	-.5945	.1420	.2813	-.0699	.0146
-7	-.3871					-7	-.3871	.0873	-.0571	-.0435	.0079
-5	-.2289					-5	-.2289	.0598	-.0848	-.0239	.0050
-3	-.0615					-3	-.0615	.0446	-.1145	-.0037	.0017
-2	.0307					-2	.0307	.0408	-.1271	-.0078	.0034
-1	.1352					-1	.1352	.0400	-.1419	-.0180	.0034
0	.2842					0	.2842	.0400	-.1645	-.0357	.0045
1	.3656					1	.3656	.0465	-.1668	-.0451	.0058
2	.4701					2	.4701	.0571	-.1698	-.0566	.0077
3	.5531					3	.5531	.0691	-.1747	-.0665	.0098
5	.7082					5	.7082	.1024	-.1858	-.0848	.0147
$M = 1.08$											
-10	-.5945					-10	-.5945	.1420	.2813	-.0699	.0146
-7	-.3871					-7	-.3871	.0873	-.0571	-.0435	.0079
-5	-.2289					-5	-.2289	.0598	-.0848	-.0239	.0050
-3	-.0615					-3	-.0615	.0446	-.1145	-.0037	.0017
-2	.0307					-2	.0307	.0408	-.1271	-.0078	.0034
-1	.1352					-1	.1352	.0400	-.1419	-.0180	.0034
0	.2842					0	.2842	.0400	-.1645	-.0357	.0045
1	.3656					1	.3656	.0465	-.1668	-.0451	.0058
2	.4701					2	.4701	.0571	-.1698	-.0566	.0077
3	.5531					3	.5531	.0691	-.1747	-.0665	.0098
5	.7082					5	.7082	.1024	-.1858	-.0848	.0147
$M = 1.10$											
-10	-.5634					-10	-.5634	.1301	-.0095	-.0657	.0121
-7	-.3665					-7	-.3665	.0809	-.0504	-.0407	.0065
-5	-.2185					-5	-.2185	.0576	-.0762	-.0227	.0039
-3	-.0819					-3	-.0819	.0446	-.0958	-.0052	.0027
-2	.0144					-2	.0144	.0417	-.1093	-.0052	.0026
-1	.0934					-1	.0934	.0403	-.1205	-.0140	.0031
0	.1983					0	.1983	.0435	-.1330	-.0256	.0037
1	.2989					1	.2989	.0474	-.1465	-.0369	.0050
2	.3967					2	.3967	.0562	-.1528	-.0480	.0064
3	.4858					3	.4858	.0682	-.1627	-.0590	.0082
5	.6467					5	.6467	.1018	-.1752	-.0779	.0128

TABLE 3-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

$$\frac{L}{c} = 0.04 \quad \frac{C_L}{c} = 0.30$$

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
-10	-.2534					-10	-.4754	.1149	-.0709	-.0536	.0155
-7	-.0576					-7	-.2659	.0682	-.1084	-.0259	.0097
-5	.0504					-5	-.0725	.0444	-.1440	-.0031	.0070
-3	.1656					-3	.1128	.0376	-.1702	.0176	.0064
-2	.2232					-2	.2176	.0388	-.1802	.0287	.0068
-1	.2750					-1	.3223	.0428	-.1806	.0403	.0074
0	.3441					0	.4077	.0506	-.1826	.0499	.0086
1	.3888					1	.4899	.0587	-.1820	.0593	.0102
2	.4391					2	.5882	.0713	-.1823	.0694	.0124
3	.4895					3	.6607	.0864	-.1830	.0792	.0150
5	.6119					5	.8154	.1240	-.1937	.0981	.0205
7	.7343										
10	.8207										
15	.8639										
20	.8236										
25	.7861										
$M = 0.60$											
-10	-.2827	.0444	-.0774	-.0323	.0084	-10	-.5014	.1203	-.0443	-.0357	.0152
-7	-.0703	.0098	-.1075	-.0097	.0046	-7	-.2959	.0743	-.0854	-.0296	.0098
-5	.0677	.0036	-.1046	.0107	.0034	-5	-.1288	.0531	-.1164	-.0099	.0076
-3	.1901	.0063	-.1049	.0249	.0039	-3	.0414	.0434	-.1454	.0095	.0066
-2	.2655	.0115	-.0984	.0330	.0048	-2	.1457	.0422	-.1602	.0211	.0067
-1	.3255	.0168	-.0962	.0408	.0056	-1	.2683	.0441	-.1741	.0346	.0070
0	.4059	.0245	-.0948	.0485	.0061	0	.3649	.0498	-.1815	.0454	.0076
1	.4515	.0315	-.0956	.0556	.0072	1	.4492	.0584	-.1817	.0552	.0089
2	.5156	.0403	-.0871	.0628	.0089	2	.5412	.0705	-.1816	.0642	.0110
3	.5726	.0508	-.0822	.0707	.0108	3	.6164	.0863	-.1830	.0738	.0135
5	.7264	.0833	-.0716	.0896	.0320	5	.7819	.1237	-.1987	.0927	.0180
7	.8261	.1208	-.0693	.1040	.0216						
10	.9059	.1926	-.1007	.1173	.0311						
15	.9201	.2998	-.1384	.1196	.0436						
20	.8745	.3894	-.1375	.1141	.0527						
25	.8631	.4736	-.1404	.0838	.0631						
$M = 0.80$											
-10	-.3048	.0618	-.0951	-.0347	.0091	-10	-.6886	.1158	-.0297	-.0550	.0134
-7	-.0968	.0238	-.1348	-.0050	.0049	-7	-.2948	.0729	-.0815	-.0301	.0088
-5	.0629	.0143	-.1347	.0079	.0041	-5	-.1407	.0535	-.1083	-.0111	.0135
-3	.1935	.0155	-.1342	.0255	.0043	-3	.0222	.0437	-.1331	.0072	.0058
-2	.2854	.0202	-.1316	.0349	.0049	-2	.1155	.0437	-.1464	.0174	.0057
-1	.3454	.0254	-.1306	.0427	.0056	-1	.2118	.0455	-.1625	.0286	.0061
0	.4189	.0321	-.1251	.0513	.0070	0	.3333	.0503	-.1751	.0422	.0069
1	.5013	.0411	-.1223	.0609	.0088	1	.4177	.0583	-.1809	.0518	.0078
2	.5689	.0497	-.1131	.0685	.0104	2	.5110	.0710	-.1795	.0616	.0094
3	.6386	.0613	-.1083	.0775	.0063	3	.5895	.0856	-.1843	.0713	.0117
5	.7856	.0935	-.1003	.0951	.0092	5	.7406	.1221	-.1945	.0887	.0164
7	.8804	.1342	-.0990	.1119	.0058	7	.8591	.1654	-.2104	.1031	.0221
10	1.0178	.2022	-.1032	.1296	.0082						
$M = 0.90$											
-10	-.3457	.0840	-.1157	-.0383	.0115	-10	-.4810	.1137	-.0355	-.0542	.0119
-7	-.1494	.0440	-.1424	-.0133	.0067	-7	-.2900	.0706	-.0779	-.0298	.0072
-5	.0171	.0279	-.1538	.0057	.0051	-5	-.1393	.0519	-.1044	-.0115	.0052
-3	.1750	.0273	-.1627	.0234	.0051	-3	.0144	.0441	-.1272	.0060	.0048
-2	.2604	.0283	-.1522	.0320	.0055	-2	.1034	.0445	-.1398	.0160	.0048
-1	.3560	.0367	-.1634	.0428	.0070	-1	.2010	.0466	-.1525	.0263	.0052
0	.4097	.0410	-.1510	.0506	.0077	0	.2972	.0508	-.1660	.0376	.0062
1	.5019	.0493	-.1507	.0610	.0097	1	.3977	.0572	-.1762	.0495	.0075
2	.5822	.0588	-.1379	.0701	.0120	2	.4810	.0688	-.1755	.0582	.0085
3	.7529	.0709	-.1610	.0805	.0142	3	.5657	.0858	-.1773	.0680	.0102
5	.7853	.1018	-.1388	.0970	.0193	5	.7093	.1235	-.1916	.0860	.0140
7	.9151	.1444	-.0969	.1115	.0255	7	.8327	.1684	-.1990	.0999	.0193
$M = 1.10$											
-10	-.4810	.1137	-.0355	-.0542	.0119						
-7	-.2900	.0706	-.0779	-.0298	.0072						
-5	-.1393	.0519	-.1044	-.0115	.0052						
-3	.0144	.0441	-.1272	.0060	.0048						
-2	.1034	.0445	-.1398	.0160	.0048						
-1	.2010	.0466	-.1525	.0263	.0052						
0	.2972	.0508	-.1660	.0376	.0062						
1	.3977	.0572	-.1762	.0495	.0075						
2	.4810	.0688	-.1755	.0582	.0085						
3	.5657	.0858	-.1773	.0680	.0102						
5	.7093	.1235	-.1916	.0860	.0140						
7	.8327	.1684	-.1990	.0999	.0193						

TABLE 3 .- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

 $\frac{c}{c} = 0.04$  $\frac{c_f}{c} = 0.40$ 

$\alpha_s$ deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$	$\alpha_s$ deg	$C_L$	$C_D$	$C_H$	$C_l$	$C_n$
$M = 0.40$											
$M = 0.55$											
-10	-0.1961					-10	-0.3709	.1138	-0.1084	-0.0396	.0130
-7	-0.0087					-7	-0.1516	.0761	-0.1425	-0.0119	.0084
-5	.1096					-5	.0468	.0603	-0.1726	-0.0101	.0063
-3	.2076					-3	.2242	.0575	-0.1894	.0297	.0065
-2	.2711					-2	.3322	.0603	-0.1913	.0411	.0072
-1	.3157					-1	.4177	.0670	-0.1875	.0509	.0084
0	.3806					0	.4935	.0761	-0.1876	.0524	.0098
1	.4239					1	.5822	.0860	-0.1856	.0693	.0118
2	.4844					2	.6563	.0992	-0.1845	.0789	.0140
3	.5306					3	.7321	.1158	-0.1838	.0897	.0166
5	.6546					5	.8708	.1526	-0.1898	.1060	.0227
7	.7584										
10	.8276										
15	.8333										
20	.8045										
25	.7757										
$M = 0.60$											
$M = 1.00$											
-10	-0.2161	.0534	-0.0830	-0.0239	.0061	-10	-0.4235	.1188	-0.0725	-0.0447	.0129
-7	-0.0078	.0263	-0.1053	.0021	.0027	-7	-0.1995	.0800	-0.1153	-0.0175	.0086
-5	.1405	.0263	-0.0983	.0190	.0022	-5	-0.0230	.0630	-0.1452	.0026	.0068
-3	.2739	.0312	-0.0972	.0341	.0034	-3	.1596	.0592	-0.1735	.0228	.0062
-2	.3417	.0368	-0.0906	.0429	.0040	-2	.2854	.0596	-0.1849	.0360	.0067
-1	.4052	.0452	-0.0900	.0501	.0045	-1	.3744	.0653	-0.1862	.0461	.0075
0	.4651	.0544	-0.0902	.0573	.0055	0	.4587	.0743	-0.1854	.0562	.0090
1	.5265	.0639	-0.0869	.0651	.0071	1	.5401	.0856	-0.1859	.0652	.0103
2	.5864	.0750	-0.0766	.1440	.0109	2	.6198	.0985	-0.1859	.0743	.0126
3	.6506	.0885	-0.0772	.0814	.0109	3	.6981	.0970	-0.1904	.0857	.0151
5	.7676	.1176	-0.0651	.0961	.0158	5	.8469	.1554	-0.2046	.1015	.0103
7	.8646	.1621	-0.0711	.1111	.0220						
10	.9017	.2280	-0.1021	.1189	.0310						
15	.8950	.3241	-0.1344	.1160	.0209						
20	.8674	.4126	-0.1369	.1134	.0260						
25	.8674	.5081	-0.1413	.1131	.0314						
$M = 0.80$											
$M = 1.05$											
-10	-0.2363	.0671	-0.0926	-0.0277	.0077	-10	-0.4119	.1148	-0.0693	-0.0429	.0106
-7	-0.0107	.0350	-0.1223	.0038		-7	-0.2015	.0783	-0.1090	-0.0183	.0087
-5	.1491	.0326	-0.1323	.0200		-5	-0.0385	.0638	-0.1353	.0003	.0078
-3	.2915	.0390	-0.1279	.0362	.0042	-3	.1171	.0601	-0.1591	.0183	.0073
-2	.3622	.0450	-0.1244	.0447	.0051	-2	.2504	.0601	-0.1763	.0325	.0074
-1	.4300	.0514	-0.1228	.0523	.0064	-1	.3467	.0648	-0.1822	.0430	.0078
0	.4920	.0596	-0.1225	.0605	.0074	0	.4282	.0739	-0.1857	.0527	.0086
1	.5559	.0695	-0.1146	.0690	.0093	1	.5097	.0856	-0.1839	.0619	.0093
2	.6315	.0793	-0.1070	.0768	.0111	2	.5912	.0988	-0.1851	.0715	.0108
3	.6993	.0922	-0.1029	.0856	.0135	3	.6579	.1136	-0.1891	.0812	.0128
5	.8252	.1272	-0.0953	.1046	.0191	5	.8179	.1184	-0.2018	.0980	.0187
7	.9123	.1653	-0.0900	.1171	.0238						
10	1.0034	.2344	-0.0907	.1301	.0334						
$M = 0.90$											
$M = 1.10$											
-10	-0.2701	.0921	-0.1347	-0.0285	.0104	-10	-0.3963	.1101	-0.0668	-0.0433	.0100
-7	-0.0436	.0576	-0.1665	-.0009	.0063	-7	-0.2024	.0773	-0.1042	-0.0193	.0098
-5	.1333	.0491	-0.1805	.0192	.0053	-5	-0.0488	.0643	-0.1290	-0.0016	.0047
-3	.2855	.0513	-0.1812	.0360	.0061	-3	.1091	.0607	-0.1531	.0173	.0047
-2	.3752	.0585	-0.1802	.0462	.0073	-2	.2039	.0625	-0.1632	.0279	.0050
-1	.4513	.0651	-0.1725	.0550	.0085	-1	.3144	.0678	-0.1740	.0399	.0056
0	.5334	.0765	-0.1801	.0652	.0102	0	.4078	.0759	-0.1809	.0504	.0064
1	.5932	.0849	-0.1695	.0730	.0118	1	.4867	.0865	-0.1812	.0598	.0079
2	.6650	.0933	-0.1523	.0811	.0139	2	.5657	.1006	-0.1815	.0681	.0095
3	.7282	.1047	-0.1459	.0906	.0162	3	.6231	.1172	-0.1853	.0816	.0115
5	.8513	.1396	-0.1484	.1062	.0214	5	.7868	.1246	-0.1963	.0950	.0166

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TABLE 3.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

 $\frac{k}{c} = 0.06$        $\frac{C_f}{c} = \text{NONE}$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
-10	-0.5310					-10	-0.7168	0.1453	0.0848	-0.0887	-0.0220
-7	-0.3947					-7	-0.5165	0.0850	0.0382	-0.0631	-0.0121
-5	-0.2942					-5	-0.3900	0.0569	0.0173	-0.0461	-0.0077
-3	-0.1973					-3	-0.2382	0.0363	0.0000	-0.0272	-0.0037
-2	-0.1471					-2	-0.1644	0.0276	-0.0037	-0.0188	-0.0023
-1	-0.0969					-1	-0.0949	0.0249	-0.0028	-0.0102	-0.0015
0	-0.0466					0	-0.0148	0.0238	0.0009	-0.0007	-0.0012
1	0.0179					1	0.0632	0.0259	0.0033	-0.0085	-0.0012
2	0.0789					2	0.1326	0.0280	0.0019	0.0171	-0.0015
3	0.1327					3	0.2087	0.0342	-0.0014	0.0256	-0.0020
5	0.2404					5	0.3373	0.0519	-0.0107	0.0416	-0.0039
7	0.3588					7	0.4891	0.0829	-0.0411	0.0604	-0.0076
10	0.5238					10	0.6852	0.1453	-0.0877	0.0863	-0.0156
15	0.6925					15	0.9593	0.2800	-0.1483	0.2118	-0.0341
20	0.7140					20	0.9804	0.3940	-0.2043	0.2118	-0.0482
25	0.6996					25	1.0225	0.5226	-0.2453	0.1262	-0.0644
$M = 0.60$											
-10	-0.6001	0.1219	0.0261	-0.0741	-0.0163	-10	-0.7372	0.1620	0.1193	-0.0866	-0.0229
-7	-0.4376	0.0598	0.0016	-0.0520	-0.0077	-7	-0.5453	0.0954	0.0733	-0.0631	-0.0137
-5	-0.3250	0.0329	-0.0123	-0.0371	-0.0036	-5	-0.4141	0.0703	0.0496	-0.0467	-0.0097
-3	-0.2068	0.0218	-0.0090	-0.0227	-0.0019	-3	-0.2626	0.0487	0.0288	-0.0301	-0.0061
-2	-0.1293	0.0218	0.0025	-0.0149	-0.0009	-2	-0.1777	0.0426	0.0187	-0.0203	-0.0042
-1	-0.0776	0.0144	0.0041	-0.0081	-0.0004	-1	-0.1030	0.0347	0.0099	-0.0114	-0.0027
0	-0.0203	0.0126	-0.0097	-0.0036	-0.0003	0	-0.0141	0.0347	0.0053	-0.0016	-0.0023
1	0.0480	0.0144	-0.0130	0.0066	0.0004	1	0.0707	0.0378	-0.0036	0.0082	-0.0024
2	0.1052	0.0162	-0.0130	0.0131	0.0012	2	0.1495	0.0428	-0.0103	0.0180	-0.0030
3	0.1699	0.0218	0.0148	0.0203	0.0009	3	0.2323	0.0477	-0.0214	0.0265	-0.0035
5	0.2880	0.0329	0.0171	0.0341	-0.0002	5	0.3737	0.0667	-0.0402	0.0435	-0.0055
7	0.4210	0.0510	0.0196	0.0505	-0.0021	7	0.5191	0.0974	-0.0697	0.0611	-0.0090
10	0.5964	0.1034	0.0359	0.0729	-0.0079	10	0.7231	0.1620	-0.1220	0.0856	-0.168
15	0.7349	0.2197	0.1152	0.0944	-0.0220	15	1.0180	0.3050	-0.1934	0.1245	-0.0359
20	0.7515	0.3143	0.1527	0.0962	-0.0351	20	1.2119	0.4688	-0.2350	0.1497	-0.0584
25	0.7478	0.3996	0.1634	0.0938	-0.0454	25	1.2200	0.6158	-0.2931	0.1497	-0.0384
$M = 0.80$											
-10	-0.6504	0.1156	0.0160	-0.0795	-0.0171	-10	-0.7197	0.1573	0.1189	-0.0851	-0.0228
-7	-0.5053	0.0640	-0.0061	-0.0593	-0.0096	-7	-0.5374	0.0964	0.0760	-0.0618	-0.0142
-5	-0.3627	0.0345	-0.0144	-0.0417	-0.0066	-5	-0.4016	0.0677	0.0545	-0.0461	-0.0101
-3	-0.2201	0.0185	-0.0127	-0.0245	-0.0021	-3	-0.2503	0.0477	0.0322	-0.0289	-0.0056
-2	-0.1581	0.0160	-0.0088	-0.0168	-0.0014	-2	-0.1707	0.0429	0.0223	-0.0198	-0.0045
-1	-0.0926	0.0148	-0.0055	-0.0099	-0.0009	-1	-0.0970	0.0382	0.0129	-0.0116	-0.0034
0	-0.0180	0.0135	0.0000	-0.0008	-0.0009	0	-0.0136	0.0372	0.0047	-0.0025	-0.0026
1	0.0575	0.0160	0.0006	-0.0079	-0.0006	1	0.0640	0.0382	-0.0034	0.0072	-0.0026
2	0.1201	0.0173	0.0022	-0.0152	-0.0006	2	0.1416	0.0429	-0.0129	0.0157	-0.0029
3	0.1926	0.0223	0.0066	-0.0233	-0.0010	3	0.2231	0.0497	-0.0227	0.0248	-0.0037
5	0.2252	0.0345	0.0061	-0.0388	-0.0020	5	0.3589	0.0677	-0.0454	0.0408	-0.0056
7	0.4778	0.0615	-0.0044	-0.0577	-0.0048	7	0.5083	0.1001	-0.0721	0.0587	-0.0091
10	0.6429	0.1143	-0.0265	-0.0789	-0.0110	10	0.6887	0.1573	-0.1184	0.0807	-0.165
15	0.7505	0.2251	-0.1113	-0.0963	-0.0257	15	1.0243	0.3053	-0.2017	0.1218	-0.0360
20	0.7830	0.3200	-0.1555	-0.0988	-0.0387	20	1.2649	0.4980	-0.2686	0.1532	-0.0611
25	0.8055	0.4158	-0.1704	-0.1012	-0.0514	25	1.3308	0.6679	-0.3004	0.1664	-0.0839
$M = 0.90$											
$M = 1.10$											
-10	-0.6870	0.1330	0.0500	-0.0864	-0.0202	-10	-0.7087	0.1537	0.1172	-0.0832	-0.0245
-7	-0.5186	0.0740	0.0172	-0.0628	-0.0109	-7	-0.5259	0.0949	0.0750	-0.0616	-0.0147
-5	-0.3989	0.0437	0.0010	-0.0461	-0.0066	-5	-0.3916	0.0651	0.0545	-0.0459	-0.0097
-3	-0.2549	0.0239	-0.0064	-0.0287	-0.0031	-3	-0.2424	0.0459	0.0322	-0.0287	-0.0062
-2	-0.1773	0.0197	-0.0064	-0.0038	-0.0021	-2	-0.1623	0.0390	0.0215	-0.0196	-0.0050
-1	-0.1045	0.0175	-0.0039	-0.0113	-0.0014	-1	-0.0877	0.0352	0.0186	-0.0115	-0.0036
0	-0.0222	0.0164	-0.0015	-0.0018	-0.0009	0	-0.0093	0.0343	0.0046	-0.0027	-0.0028
1	0.0621	0.0164	-0.0015	-0.0081	-0.0008	1	0.0746	0.0358	-0.0054	0.0069	-0.0029
2	0.1374	0.0197	-0.0035	-0.0170	-0.0009	2	0.1492	0.0399	-0.0148	0.0151	-0.0033
3	0.2172	0.0262	-0.0035	-0.0254	-0.0012	3	0.2294	0.0459	-0.0231	0.0255	-0.0038
5	0.3435	0.0414	-0.0029	-0.0614	-0.0028	5	0.3637	0.0651	-0.0454	0.0395	-0.0060
7	0.4876	0.0709	-0.0225	-0.0595	-0.0062	7	0.5073	0.0953	-0.0672	0.0570	-0.0098
10	0.6649	0.1308	-0.0554	-0.0843	-0.0137	10	0.6900	0.1537	-0.1147	0.0791	-0.169
15	0.8865	0.2529	-0.1010	-0.1140	-0.0302	15	0.9996	0.2980	-0.2005	0.1177	-0.0356
20	0.8754	0.3342	-0.1804	-0.1079	-0.0215	20	1.2421	0.4897	-0.2624	0.1479	-0.0606
25	0.9175	0.4665	-0.2025	-0.1130	-0.0291	25	1.3801	0.6953	-0.3160	0.1690	-0.0870

TABLE 3 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

$$\frac{c}{c} = 0.06 \quad \frac{c_f}{c} = 0.20$$

$\alpha$ , deg	$c_L$	$c_D$	$c_K$	$c_l$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_K$	$c_l$	$c_n$
$M = 0.40$											
$M = 0.55$											
-10	-.2415					-10	.5179	.1279	-.0211	-.0634	-.0148
-7	-.0972					-7	.3615	.0801	-.0444	-.0412	-.0082
-5	.0252					-5	.2114	.0552	-.0748	-.0224	-.0037
-3	.1512					-3	.0317	.0395	-.1099	-.0019	-.0008
-2	.1908					-2	.0740	.0395	-.1239	.0101	-.0003
-1	.2520					-1	.1797	.0395	-.1426	.0226	-.0002
0	.3060					0	.2790	.0446	-.1510	.0234	-.0010
1	.3600					1	.3699	.0552	-.1637	.0436	-.0020
2	.4104					2	.4545	.0654	-.1637	.0534	-.0032
3	.4680					3	.5390	.0738	-.1730	.0640	-.0046
5	.5508					5	.6743	.0520	-.1824	.0636	-.0082
7	.6696					7	.8223	.1518	-.2105	.0947	-.0144
10	.8208					10	.9978	.2266	-.2450	.1201	-.0243
15	.9071					15	1.2134	.3670	-.2665	.1485	-.0425
20	.8783					20	1.0908	.4501	-.2684	.1279	-.0518
25	.8424					25	1.0781	.5676	-.2759	.1259	-.0659
$M = 0.60$											
$M = 1.00$											
-10	-.2777	.0800	-.0983	-.0324	-.0036	-10	.5509	.1385	.0067	-.0672	-.0160
-7	-.1055	.0327	-.1229	-.0093	.0010	-7	.3605	.0907	-.0381	-.0429	-.0087
-5	.0167	.0237	-.1246	.0042	.0024	-5	.2086	.0676	-.0694	-.0251	-.0053
-3	.1444	.0211	-.1253	.0186	.0032	-3	.0608	.0557	-.0919	-.0084	-.0031
-2	.2037	.0211	-.1246	.0252	.0035	-2	.0223	.0529	-.1066	.0015	-.0024
-1	.2610	.0255	-.1229	.0312	.0030	-1	.1154	.0518	-.1223	.0116	-.0021
0	.3240	.0300	-.1212	.0387	.0023	0	.2309	.0557	-.1388	.0233	-.0022
1	.4036	.0344	-.1212	.0473	.0020	1	.3220	.0628	-.1532	.0342	-.0028
2	.4646	.0437	-.1212	.0545	.0017	2	.4253	.0717	-.1702	.0457	-.0039
3	.5091	.0481	-.1146	.0602	.0014	3	.5124	.0857	-.1859	.0567	-.0055
5	.6258	.0674	-.1106	.0740	-.0007	5	.6622	.1146	-.2074	.0744	-.0091
7	.7461	.1011	-.1130	.0896	-.0046	7	.8162	.1584	-.2311	.0924	-.0148
10	.8757	.1692	-.1351	.1073	-.0122	10	.9924	.2351	-.2598	.1137	-.0243
15	.9201	.2822	-.1965	.1153	-.0267	15	1.2637	.2939	-.3046	.1455	-.0444
20	.8961	.3744	-.2137	.1105	-.0389	20	1.3974	.4289	-.3180	.1665	-.1350
25	.8720	.4562	-.2154	.1061	-.0492	25	1.3447	.3516	-.3566	.1566	-.0813
$M = 0.80$											
$M = 1.05$											
-10	-.3361	.0758	-.1002	-.0338	-.0068	-10	.5464	.1377	.0121	-.0712	-.0162
-7	-.1768	.0381	-.1248	-.0187	-.0017	-7	.3578	.0908	-.0323	-.0497	-.0091
-5	-.0439	.0233	-.1292	-.0030	.0009	-5	.2178	.0698	-.0624	-.0332	-.0057
-3	.1041	.0216	-.1315	.0134	.0015	-3	.0603	.0554	-.0838	-.0154	-.0029
-2	.1819	.0173	-.1304	.0221	.0015	-2	.0117	.0535	-.0968	-.0071	-.0026
-1	.2546	.0266	-.1304	.0302	.0014	-1	.1050	.0535	-.1110	.0032	-.0023
0	.3198	.0314	-.1298	.0386	.0010	0	.2003	.0554	-.1239	.0135	-.0023
1	.4013	.0389	-.1331	.0481	.0005	1	.2820	.0622	-.1368	.0230	-.0032
2	.4803	.0469	-.1359	.0578	-.0002	2	.3733	.0698	-.1514	.0332	-.0042
3	.5456	.0554	-.1292	.0656	-.0012	3	.4628	.0813	-.1669	.0431	-.0057
5	.6823	.0790	-.1248	.0808	-.0043	5	.6320	.1118	-.1987	.0628	-.0246
7	.8027	.1179	-.1304	.0962	-.0087	7	.7778	.1540	-.2237	.0806	-.0149
10	.9256	.1776	-.1399	.1136	-.0168	10	.9606	.2295	-.2564	.1019	-.0245
15	.9230	.2824	-.2009	.1132	-.0310	15	1.2328	.3903	-.3071	.1337	-.0445
20	.9406	.3873	-.2219	.1132	-.0476	20	1.4273	.5890	-.3484	.1598	-.0678
25	.9356	.4811	-.2275	.1120	-.0552	25	1.4195	.7364	-.3613	.1586	-.0889
$M = 0.90$											
$M = 1.10$											
-10	-.4663	.1055	-.0565	-.0559	-.0120	-10	.5311	.1361	.0116	-.0737	-.0167
-7	-.2920	.0606	-.0835	-.0325	-.0049	-7	.3459	.0883	-.0310	-.0523	-.0104
-5	-.1144	.0344	-.1228	-.0106	-.0004	-5	.2113	.0671	-.0571	-.0365	-.0076
-3	.0477	.0278	-.1375	-.0077	.0008	-3	.0636	.0542	-.0815	-.0198	-.0044
-2	.1443	.0284	-.1425	-.0190	.0009	-2	.0224	.0524	-.0951	-.0103	-.0036
-1	.2243	.0318	-.1425	-.0279	.0005	-1	.1028	.0524	-.1138	-.0015	-.0032
0	.2976	.0389	-.1449	-.0455	.0001	0	.1870	.0542	-.1195	.0079	-.0033
1	.3775	.0449	-.1449	-.0451	-.0005	1	.2767	.0598	-.1332	.0180	-.0040
2	.4563	.0535	-.1483	-.0555	-.0015	2	.3590	.0681	-.1460	.0274	-.0046
3	.5529	.0644	-.1522	-.0672	-.0034	3	.4469	.0791	-.1605	.0369	-.0062
5	.6884	.0917	-.1596	-.0825	-.0065	5	.5946	.1075	-.1882	.0543	-.0096
7	.8061	.1310	-.1695	-.0985	-.0116	7	.7517	.1481	-.2151	.0728	-.0149
10	.9438	.1972	-.1866	.1182	-.0208	10	.9256	.2216	-.2482	.0941	-.0241
15	1.1414	.3309	.2137	.1416	-.0396	15	1.1967	.3779	-.2978	.1253	-.0431
20	1.0437	.4248	.2555	.1236	-.0491	20	1.3837	.5720	-.3408	.1492	-.0664
25	1.0437	.5330	.2604	.1232	-.0630	25	1.4885	.7781	-.3830	.1646	-.0936

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TABLE 3.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Continued

 $\frac{k}{c} = 0.06$        $\frac{c_f}{c} = 0.30$ 

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$
$M = 0.40$						$M = 0.95$					
-10	.1905					-10	.4415	.1185	-.0739	-.0537	-.0160
-7	-.0395					-7	.2884	.0744	-.0911	-.0313	-.0099
-5	.0898					-5	.1447	.0534	-.1191	-.0115	-.0056
-3	.1977					-3	.0866	.0458	-.1542	-.0125	-.0034
-2	.2444					-2	.1996	.0484	-.1705	.0253	-.0038
-1	.3055					-1	.2873	.0613	-.1916	.0355	-.0050
0	.3738					0	.3876	.0872	-.1869	.0467	-.0053
1	.4169					1	.4774	.0744	-.1986	.0578	-.0071
2	.4852					2	.5598	.0847	-.1963	.0667	-.0084
3	.5391					3	.6232	.0925	-.1893	.0747	-.0093
5	.6325					5	.7647	.1314	-.2103	.0923	-.0139
7	.7475					7	.8978	.1756	-.2173	.1094	-.0197
10	.8769					10	1.0605	.2535	-.2477	.1299	-.0304
15	.9128					15	1.2126	.3927	-.2477	.1521	-.0246
20	.8697					20	1.4492	.5017	-.2757	.1378	-.0308
25	.8194					25	1.1070	.6088	-.2757	.1309	-.0375
$M = 0.60$						$M = 1.00$					
-10	.2312	.0629	-.0966	-.0287	-.0082	-10	.5058	.1321	-.0403	-.0575	-.0167
-7	.0518	.0237	-.1228	-.0045	-.0023	-7	.2934	.0868	-.0805	-.0319	-.0100
-5	.0869	.0165	-.1228	.0114	-.0014	-5	.1315	.0698	-.1119	-.0124	-.0063
-3	.2201	.0181	-.1219	.0269	-.0014	-3	.0445	.0587	-.1410	-.0074	-.0048
-2	.2885	.0211	-.1235	.0344	-.0016	-2	.1558	.0587	-.1589	-.0188	-.0047
-1	.3496	.0255	-.1210	.0419	-.0017	-1	.2610	.0627	-.1767	-.0316	-.0049
0	.4235	.0329	-.1228	.0494	-.0023	0	.3682	.0702	-.1924	-.0434	-.0056
1	.4846	.0392	-.1228	.0569	-.0026	1	.4593	.0836	-.2050	-.0540	-.0071
2	.5308	.0462	-.1187	.0625	-.0032	2	.5442	.0945	-.2126	-.0635	-.0084
3	.5863	.0555	-.1129	.0688	-.0044	3	.6252	.1095	-.2282	-.0732	-.0102
5	.7010	.0773	-.1105	.0829	-.0069	5	.7627	.1408	-.2372	-.0900	-.0143
7	.8249	.1165	-.1088	.0988	-.0116	7	.9145	.1890	-.2506	.1080	-.0209
10	.9248	.1838	-.1309	.1134	-.0204	10	1.0804	.2717	-.2730	.1287	-.0314
15	.9507	.2985	-.1801	.1179	-.0360	15	1.3151	.4338	-.3043	.1568	-.0529
20	.9137	.3928	-.1972	.1161	-.0483	20	1.4243	.6070	-.3150	.1748	-.0765
25	.8767	.4783	-.2005	.1048	-.0585	25	1.3313	.7283	-.3357	.1617	-.0925
$M = 0.80$						$M = 1.05$					
-10	.2795	.0872	-.1247	-.0325	-.0084	-10	.5013	.1287	-.0215	-.0548	-.0166
-7	.0940	.0308	-.1442	-.0021	-.0025	-7	.3031	.0852	-.0223	-.0303	-.0108
-5	.0514	.0223	-.1442	.0577	-.0017	-5	.1516	.0641	-.0894	-.0135	-.0069
-3	.1968	.0233	-.1414	.0243	-.0017	-3	.0136	.0550	-.1182	-.0052	-.0049
-2	.2697	.0266	-.1414	.0325	-.0019	-2	.1108	.0554	-.1324	-.0159	-.0046
-1	.3346	.1547	-.1420	.0404	-.0025	-1	.1982	.0583	-.1461	-.0258	-.0054
0	.4098	.0634	-.1425	.0489	-.0036	0	.2934	.0645	-.1612	-.0362	-.0062
1	.4850	.0481	-.1414	.0578	-.0046	1	.3984	.0731	-.1762	-.0465	-.0072
2	.5577	.0574	-.1414	.0665	-.0055	2	.4761	.0841	-.1891	-.0571	-.0084
3	.6204	.0667	-.1319	.0740	-.0064	3	.5674	.0985	-.2012	-.0674	-.0103
5	.7570	.0942	-.1286	.0989	-.0099	5	.7209	.1329	-.2219	-.0858	-.0147
7	.8723	.1374	-.1319	.1067	-.0152	7	.8589	.1792	-.2390	.1022	-.0204
10	.9876	.2010	-.1375	.1233	-.0235	10	1.0377	.2590	-.2622	.1226	-.0236
15	.9400	.3083	-.1957	.1160	-.0377	15	1.2709	.4234	-.3034	.1522	-.0529
20	.9575	.4141	-.2135	.1148	-.0513	20	1.4185	.6117	-.3352	.1755	-.0798
25	.9350	.5141	-.2190	.1123	-.0643	25	1.3874	.7608	-.3525	.1736	-.0997
$M = 0.90$						$M = 1.10$					
-10	.3865	.0961	-.0933	-.0453	-.0139	-10	.4823	.1279	-.0188	-.0537	-.0173
-7	.2087	.0562	-.1203	-.0225	-.0063	-7	.3010	.0819	-.0550	-.0310	-.0113
-5	.0533	.0371	-.1448	-.0031	-.0032	-5	.1552	.0624	-.0889	-.0135	-.0077
-3	.1299	.0333	-.1547	.0171	-.0024	-3	.0150	.0538	-.1157	-.0047	-.0065
-2	.2164	.0360	-.1547	.0271	-.0028	-2	.1028	.0542	-.1282	-.0145	-.0062
-1	.2908	.0415	-.1547	.0356	-.0033	-1	.1851	.0565	-.1406	-.0225	-.0067
0	.3718	.0480	-.1562	.0447	-.0037	0	.2710	.0611	-.1546	-.0337	-.0074
1	.4540	.0590	-.1596	.0544	-.0048	1	.3720	.0703	-.1675	-.0439	-.0080
2	.5375	.0682	-.1596	.0643	-.0061	2	.4486	.0809	-.1790	-.0535	-.0098
3	.6285	.0830	-.1611	.0751	-.0078	3	.5328	.0948	-.1910	-.0634	-.0114
5	.7595	.1134	-.1669	.0916	-.0116	5	.6767	.1279	-.2133	-.0805	-.0135
7	.8705	.1534	-.1719	.1074	-.0170	7	.8262	.1738	-.2299	-.0974	-.0212
10	.9987	.2200	-.1792	.1257	-.0264	10	.9982	.2337	-.2547	-.1177	-.0315
15	1.0966	.3472	-.2087	.1365	-.0427	15	1.2225	.4137	-.2936	.1455	-.0519
20	1.0522	.4564	-.2430	.1246	-.0556	20	1.4020	.6086	-.3324	.1694	-.0783
25	1.0744	.5634	-.2475	.1218	-.0692	25	1.4880	.8128	-.3672	.1819	-.1049

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TABLE 3 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 3 MODEL - Concluded

$$\frac{t}{c} = 0.06 \quad \frac{c_t}{c} = 0.40$$

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$
$M = 0.40$											
$M = 0.55$											
-10	-0.1438					-10	-0.3745	.1162	-.0723	-.0408	-.0144
-7	.0108					-7	-0.1899	.0757	-.1073	-.0179	-.0088
-5	.1258					-5	-0.0179	.0643	-.1446	.0026	-.0065
-3	.2372					-3	.01941	.0622	-.1773	.0261	-.0062
-2	.3127					-2	.02848	.0675	-.1867	.0364	-.0057
-1	.3522					-1	.03692	.0726	-.1890	.0461	-.0074
0	.4133					0	.04557	.0829	-.1937	.0563	-.0089
1	.4528					1	.05338	.0954	-.1960	.0649	-.0106
2	.5103					2	.06118	.1089	-.1983	.0743	-.0124
3	.5571					3	.06899	.1245	-.2030	.0836	-.0144
5	.6613					5	.07954	.1525	-.2030	.0980	-.0183
7	.7799					7	.09135	.1970	-.2170	.1127	-.0237
10	.8913					10	1.00549	.2749	-.2333	.1314	-.0680
15	.8985					15	1.0127	.4150	-.2567	.1536	-.0538
20	.8733					20	1.11181	.5084	-.2754	.1348	-.0640
25	.8158					25	1.0759	.6184	-.2846	.1297	-.0775
$M = 0.60$											
$M = 1.00$											
-10	-0.1755	.0591	-.0678	-.0209	-.0062	-10	-0.4142	.1263	-.0482	-.0451	-.0161
-7	.0095	.0255	-.0981	.0030	-.0029	-7	-0.2101	.0885	-.0992	-.0203	-.0101
-5	.1478	.0255	-.1021	.0182	-.0019	-5	-0.0505	.0249	-.1318	-.0013	-.0072
-3	.2772	.0299	-.1047	.0335	-.0020	-3	.01364	.0695	-.1654	.0191	-.0068
-2	.3381	.0344	-.1021	.0404	-.0024	-2	.02526	.0715	-.1832	.0322	-.0073
-1	.4046	.0410	-.1021	.0478	-.0032	-1	.03485	.0786	-.1945	.0433	-.0086
0	.4601	.0473	-.1021	.0550	-.0036	0	.04223	.0875	-.2007	.0515	-.0089
1	.5229	.0547	-.1021	.0613	-.0042	1	.05092	.0994	-.2056	.0613	-.0103
2	.5691	.0654	-.0998	.0682	-.0047	2	.05859	.1123	-.2109	.0701	-.0117
3	.6356	.0765	-.0981	.0756	-.0057	3	.06668	.1291	-.2154	.0793	-.0141
5	.7484	.1016	-.0956	.0894	-.0087	5	.08082	.1669	-.2324	.0968	-.0188
7	.8595	.1438	-.0981	.1034	-.0139	7	.09193	.2136	-.2369	.1128	-.0249
10	.9608	.2162	-.1348	.1154	-.0232	10	1.0911	.3011	-.2726	.1318	-.0360
15	.9275	.3215	-.1913	.1136	-.0380	15	1.2850	.4570	-.3039	.1576	-.0572
20	.9054	.4161	-.2185	.1085	-.0500	20	1.3861	.6259	-.3263	.1700	-.0790
25	.8795	.5033	-.2281	.1043	-.0602	25	1.2810	.7454	-.3504	.1569	-.0937
$M = 0.80$											
$M = 1.05$											
-10	-0.2191	.0648	-.0286	-.0277	-.0084	-10	-0.4075	.1250	-.0428	-.0444	-.0156
-7	.0275	.0363	-.1298	-.0010	-.0040	-7	-0.2135	.0887	-.0901	-.0212	-.0099
-5	.1152	.0326	-.1318	.0192	-.0033	-5	-0.0640	.0745	-.1223	-.0039	-.0083
-3	.2504	.0326	-.1329	.0312	-.0036	-3	.01106	.0697	-.1524	.0157	-.0071
-2	.3180	.0431	-.1329	.0391	-.0040	-2	.02135	.0716	-.1695	.0272	-.0075
-1	.3881	.0506	-.1340	.0472	-.0048	-1	.03066	.0782	-.1803	.0377	-.0085
0	.4520	.0586	-.1318	.0547	-.0053	0	.03881	.0860	-.1910	.0471	-.0090
1	.5221	.0676	-.1302	.0636	-.0068	1	.04716	.0974	-.1996	.0565	-.0102
2	.5659	.0739	-.1230	.0689	-.0075	2	.05492	.1098	-.2060	.0659	-.0117
3	.6511	.0861	-.1185	.0778	-.0092	3	.06307	.1259	-.2125	.0751	-.0138
5	.7738	.1152	-.1163	.0934	-.0131	5	.07821	.1622	-.2275	.0911	-.0180
7	.8739	.1570	-.1230	.1094	-.0181	7	.08927	.2119	-.2447	.1071	-.0243
10	.9991	.2215	-.1302	.1256	-.0274	10	1.0596	.2940	-.2704	.1266	-.0346
15	.9115	.3203	-.1981	.1114	-.0398	15	1.2730	.4628	-.3176	.1532	-.0564
20	.9365	.4285	-.2216	.1135	-.0534	20	1.4089	.6453	-.3587	.1727	-.0803
25	.9090	.5173	-.2282	.1102	-.0658	25	1.3972	.7922	-.3777	.1715	-.1005
$M = 0.90$											
$M = 1.10$											
-10	-0.3216	.0976	-.0907	-.0354	-.0123	-10	-0.3974	.1211	-.0425	-.0453	-.0163
-7	-0.1297	.0599	-.1227	-.0120	-.0068	-7	-0.2183	.0871	-.0866	-.0219	-.0112
-5	.0255	.0490	-.1433	.0063	-.0048	-5	-0.0653	.0716	-.1205	-.0041	-.0083
-3	.1940	.0479	-.1497	.0251	-.0047	-3	.01063	.0670	-.1465	.0148	-.0074
-2	.2783	.0534	-.1530	.0350	-.0053	-2	.01959	.0688	-.1630	.0249	-.0076
-1	.3471	.0599	-.1545	.0431	-.0060	-1	.02798	.0670	-.1733	.0340	-.0083
0	.4158	.0654	-.1520	.0513	-.0069	0	.03638	.0826	-.1836	.0438	-.0091
1	.5043	.1353	-.1619	.0619	-.0083	1	.04608	.0918	-.1939	.0543	-.0103
2	.5877	.0863	-.1545	.0709	-.0095	2	.05410	.1056	-.2022	.0635	-.0122
3	.6697	.1020	-.1594	.0816	-.0113	3	.06156	.1201	-.2063	.0720	-.0139
5	.7829	.1326	-.1619	.0969	-.0131	5	.07369	.1541	-.2228	.0872	-.0181
7	.8982	.1772	-.1741	.1127	-.0214	7	.08751	.2047	-.2419	.1035	-.0242
10	1.0201	.2482	-.1864	.1292	-.0310	10	1.0335	.2845	-.2657	.1217	-.0347
15	1.0535	.3524	-.2208	.1236	-.0438	15	1.2499	.4405	-.3054	.1464	-.0548
20	1.0423	.4744	-.2551	.1249	-.0585	20	1.3805	.6423	-.3549	.1672	-.0836
25	1.0113	.5782	-.2620	.1220	-.0715	25	1.4551	.8348	-.3879	.1739	-.1082

TABLE 4-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL

 $\frac{L}{c} = 0.04$        $\frac{C_f}{c} = \text{NONE}$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$											
-10	.6610					-7	.7197	.1094	.0930	-.0803	.0115
-7	.5073					-5	.5466	.0823	.0606	-.0615	.0068
-5	.3667					-3	.3685	.0325	.0326	-.0413	.0039
-3	.2438					-2	.2597	.0192	.0189	-.0291	.0020
-2	.1801					-1	.1509	.0137	.0122	-.0169	.0013
-1	.1186					0	.0346	.0116	.0104	-.0038	.0011
0	.0395					1	.0866	.0106	.0074	-.0105	.0015
1	.0285					2	.2003	.0152	.0011	-.0229	.0024
2	.0878					3	.2943	.0228	-.0088	-.0347	.0041
3	.1537					5	.5095	.0480	-.0441	.0591	.0095
5	.2833					7	.6826	.0845	-.0779	.0782	.0142
7	.4282										
10	.5907										
15	.6830										
20	.6896										
25	.6852										
$M = 0.60$											
-10	.7326	.1257	.0284	-.0844	.0156	-7	.6049	.0915	.0729	-.0642	.0097
-7	.5826	.0615	-.0158	-.0660	.0083	-5	.4112	.0552	.0360	-.0479	.0057
-5	.4152	.0293	-.0144	-.0462	.0040	-3	.2836	.0343	.0173	-.0315	.0034
3	.2717	.0145	-.0053	-.0297	.0021	-2	.1983	.0265	.0116	-.0219	.0028
-2	.1935	.0107	-.0027	-.0204	.0016	-1	.1111	.0215	.0037	-.0118	.0020
-1	.1174	.0086	.0041	-.0113	.0010	0	.0095	.0195	-.0009	-.0004	.0020
0	.0217	.0064	.0024	-.0010	.0008	1	.0922	.0180	-.0003	.0115	.0021
1	.0543	.0080	.0057	.0089	.0012	2	.2056	.0244	-.0114	.0233	.0036
2	.1283	.0117	.0129	.0181	.0018	3	.2883	.0297	-.0195	.0341	.0047
3	.2043	.0171	.0179	.0270	.0028	5	.4348	.0503	-.0342	.0505	.0079
5	.3587	.0348	.0307	.0452	.0059	7	.5931	.0805	-.0624	.0674	.0120
7	.5130	.0647	.0258	.0643	.0109						
10	.6761	.1283	.0158	.0864	.0205						
15	.7370	.2191	-.0677	.1006	.0333						
20	.7370	.3015	-.0757	-.0996	.0442						
25	.7630	.3902	-.0819	.1016	.0545						
$M = 0.80$											
-10	.8224	.1436	.0233	-.0927	.0166	-7	.5612	.0888	.0681	-.0615	.0088
-7	.6585	.0799	-.0072	-.0730	.0092	-5	.4226	.0564	.0411	-.0464	.0055
-5	.4946	.0399	-.0139	-.0531	.0050	-3	.2817	.0352	.0253	-.0300	.0031
-3	.3248	.0194	-.0072	-.0332	.0025	-2	.1954	.0277	.0176	-.0217	.0022
-2	.2264	.0139	-.0010	-.0222	.0017	-1	.1045	.0226	.0080	-.0097	.0017
-1	.1281	.0092	-.0001	-.0113	.0011	0	.0068	.0204	.0001	.0014	.0013
0	.0387	.0066	.0076	-.0014	.0008	1	.1045	.0210	-.0058	.0138	.0024
1	.0685	.0073	.0109	.0109	.0011	2	.1908	.0252	-.0124	.0234	.0029
2	.1669	.0107	.0159	.0215	.0019	3	.2817	.0321	-.0230	.0336	.0043
3	.2562	.0150	.0244	.0323	.0029	5	.4294	.0511	-.0399	.0508	.0073
5	.4559	.0337	.0300	.0531	.0065	7	.5680	.0804	-.0607	.0664	.0115
7	.6227	.0648	.0206	.0728	.0113						
10	.7419	.1253	-.0192	.0904	.0200						
15	.7866	.2242	-.0768	.0999	.0335						
20	.8164	.3165	-.0957	.1013	.0436						
$M = 0.90$											
-7	.7107	.0993	.0416	-.0806	.0114	-7	.5597	.0887	.0690	-.0597	.0086
-5	.5404	.0529	.0143	-.0607	.0059	-5	.4242	.0556	.0449	-.0451	.0052
-3	.3619	.0239	-.0018	-.0394	.0028	-3	.2842	.0352	.0276	-.0290	.0028
-2	.2518	.0142	-.0059	-.0265	.0017	-2	.1946	.0285	.0192	-.0196	.0020
-1	.1521	.0097	.0018	-.0159	.0010	-1	.1115	.0237	.0092	-.0096	.0015
0	.0420	.0064	.0087	-.0040	.0008	0	.0131	.0207	-.0003	.0005	.0014
1	.0787	.0064	.0148	.0097	.0011	1	.0855	.0223	-.0078	.0124	.0019
2	.1862	.0097	.0205	.0213	.0019	2	.1749	.0274	-.0128	.0224	.0026
3	.3121	.0148	.0144	.0352	.0038	3	.2602	.0342	-.0236	.0321	.0037
5	.5245	.0613	-.0126	.0607	.0089	5	.4067	.0546	-.0414	.0498	.0070
7	.6871	.0767	-.0404	.0806	.0143	7	.5357	.0828	-.0601	.0647	.0108

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TABLE 4.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

$$\frac{c}{c} = 0.04 \quad \frac{C_f}{c} = 0.10$$

$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_H$	$c_I$	$c_n$
$M = 0.40$											
-10	-0.3849					-10	-0.7523	.1640	.0159	-0.0873	.0178
-7	-0.2357					-7	-0.5745	.0994	.0028	-0.0632	.0112
-5	-0.0930					-5	-0.3798	.0610	-0.0334	-0.0416	.0067
-3	.0035					-3	-0.1972	.0365	-0.0728	-0.0185	.0042
-2	.0695					-2	-0.0755	.0299	-0.0885	-0.0043	.0034
-1	.1179					-1	.0443	.0293	-0.1091	.0006	.0034
0	.1881					0	.1753	.0335	-0.1206	.0332	.0042
1	.2314					1	.2946	.0383	-0.1331	.0369	.0055
2	.2920					2	.4114	.0467	-0.1395	-0.0236	.0073
3	.3503					3	.5283	.0575	-0.1475	.0628	.0096
5	.4747										
7	.6012										
10	.7764										
15	.8086										
20	.7875										
25	.7526										
$M = 0.60$											
-10	-0.4740	.0826	-0.0525	-0.0567	.0116	-10	-0.7444	.1619	.0460	-0.0850	.0178
-7	-0.2955	.0411	-0.094	-0.0325	.0059	-7	-0.5264	.0969	.0054	-0.0581	.0106
-5	-0.1327	.0231	-0.0890	-0.0130	.0032	-5	-0.3734	.0650	-0.0230	-0.0396	.0075
-3	-0.0064	.0216	-0.0863	.0026	.0026	-3	-0.2110	.0485	-0.0474	-0.0206	.0054
-2	.0696	.0221	-0.0894	.0120	.0025	-2	-0.1113	.0394	-0.0598	-0.0088	.0045
-1	.1434	.0274	-0.0798	.0203	.0030	-1	-0.0139	.0245	-0.0714	.0026	.0043
0	.2161	.0321	-0.0789	.0289	.0036	0	.0974	.0376	-0.0865	.0144	.0045
1	.2878	.0353	-0.0792	.0361	.0046	1	.2226	.0410	-0.1033	.0294	.0056
2	.3767	.0426	-0.0673	.0471	.0058	2	.3293	.0468	-0.1192	.0413	.0066
3	.4430	.0532	-0.0651	.0552	.0065	3	.4569	.0582	-0.1399	.0545	.0085
5	.5821	.0853	-0.0510	.0726	.0104						
7	.7041	.1142	-0.0563	.0906	.0158						
10	.8068	.1784	-0.0951	.1058	.0245						
15	.8325	.1400	-0.1196	.1133	.0371						
20	.8218	.1898	-0.1184	.1120	.0493						
25	.8218	.2231	-0.1133	.1104	.0583						
$M = 0.80$											
-10	-0.5954	.1096	-0.0757	-0.0712	.0131	-7	-0.5034	.0953	.0065	-0.0558	.0090
-7	-0.4048	.0570	-0.0982	-0.0456	.0087	-5	-0.3586	.0652	-0.0214	-0.0385	.0055
-5	-0.2305	.0307	-0.0993	-0.0165	.0040	-3	-0.2161	.0487	-0.0422	-0.0211	.0035
-3	-0.0660	.0205	-0.0943	-0.0045	.0027	-2	-0.1181	.0422	-0.0514	-0.0101	.0025
-2	.0352	.0205	-0.0921	.0073	.0025	-1	-0.0245	.0405	-0.0614	.0005	.0023
-1	.1305	.0224	-0.0925	.0182	.0029	0	.0757	.0417	-0.0756	.0127	.0031
0	.2258	.0267	-0.0865	.0296	.0036	1	.1827	.0433	-0.0840	.0245	.0031
1	.3461	.0310	-0.0939	.0458	.0046	2	.2762	.0498	-0.1002	.0355	.0045
2	.4356	.0375	-0.0880	.0534	.0060	3	.3920	.0597	-0.1153	.0476	.0065
3	.5309	.0465	-0.0820	.0641	.0079	5	.5769	.0893	-0.1480	.0693	.0099
5	.7025	.0735	-0.0727	.0845	.0128						
$M = 0.90$											
-10	-0.7771	.1466	-0.0107	-0.0842	.0172	-7	-0.4845	.0891	.0079	-0.0538	.0078
-7	-0.5318	.0822	-0.0454	-0.0574	.0093	-5	-0.3451	.0617	-0.0177	-0.0374	.0048
-5	-0.4286	.0441	-0.0579	-0.0317	.0049	-3	-0.2101	.0448	-0.0370	-0.0207	.0030
-3	-0.1239	.0267	-0.1023	-0.0108	.0031	-2	-0.1158	.0401	-0.0442	-0.0106	.0027
-2	-.0024	.0225	-0.1062	.0023	.0028	-1	-0.0236	.0379	-0.0562	.0000	.0021
-1	.1110	.0241	-0.1112	.0163	.0030	0	.0579	.0432	-0.0631	.0109	.0024
0	.2349	.0283	-0.1123	.0294	.0038	1	.1543	.0437	-0.0755	.0216	.0033
1	.3434	.0321	-0.1109	.0423	.0049	2	.2487	.0511	-0.0904	.0330	.0044
2	.4447	.0403	-0.1172	.0558	.0068	3	.3516	.0611	-0.1046	.0442	.0058
3	.5783	.0523	-0.1215	.0683	.0091	5	.5230	.0880	-0.1393	.0642	.0094
$M = 1.00$											
$M = 1.05$											
$M = 1.10$											

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TABLE 4.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

 $\frac{c}{c} = 0.04$        $\frac{c_f}{c} = 0.20$ 

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$
$M = 0.40$											
-10	-0.3288					-10	-0.6673	0.1482	-0.0273	-0.0769	0.0149
-7	-0.1270					-7	-0.4277	0.0835	-0.0657	-0.0484	0.0094
-5	0.0109					-5	-0.2273	0.0505	-0.1031	-0.0247	0.0056
-3	0.1802					-3	-0.0367	0.0354	-0.1977	-0.0019	0.0040
-2	0.1954					-2	0.1027	0.0324	-0.1625	0.0134	0.0042
-1	0.2529					-1	0.2273	0.0337	-0.1736	0.0271	0.0051
0	0.3169					0	0.3422	0.0571	-0.1754	0.0399	0.0046
1	0.3722					1	0.4522	0.0460	-0.1740	0.0519	0.0050
2	0.4287					2	0.5622	0.0599	-0.1768	0.0630	0.0102
3	0.4928					3	0.6599	0.0688	-0.1846	0.0732	0.0126
5	0.6186										
7	0.7271										
10	0.8139										
15	0.8356										
20	0.7987										
25	0.6728										
$M = 0.60$											
-10	-0.3898	0.0769	-0.0747	-0.0448	0.0086	-10	-0.6566	0.1477	-0.0032	-0.0744	0.0163
-7	-0.1611	0.0280	-0.1154	-0.0163	0.0028	-7	-0.4354	0.0876	-0.0438	-0.0495	0.0059
-5	0.0021	0.0125	-0.1148	0.0034	0.0014	-5	-0.2678	0.0590	-0.0756	-0.0288	0.0068
-3	0.1536	0.0137	-0.1141	0.0207	0.0016	-3	-0.0931	0.0430	-0.1069	-0.0080	0.0050
-2	0.2395	0.0169	-0.1108	0.0301	0.0024	-2	0.0116	0.0384	-0.1268	0.0044	0.0047
-1	0.3136	0.0193	-0.1068	0.0383	0.0032	-1	0.1281	0.0367	-0.1457	0.0177	0.0049
0	0.3823	0.0241	-0.1033	0.0468	0.0041	0	0.2678	0.0389	-0.1674	0.0327	0.0058
1	0.4511	0.0299	-0.0944	0.0541	0.0053	1	0.3958	0.0446	-0.1738	0.0463	0.0073
2	0.5262	0.0369	-0.0872	0.0624	0.0068	2	0.5053	0.0550	-0.1769	0.0579	0.0092
3	0.6014	0.0467	-0.0818	0.0709	0.0087	3	0.5984	0.0676	-0.1812	0.0680	0.0113
5	0.7346	0.0747	-0.0675	0.0886	0.0155						
7	0.8420	0.1125	-0.0727	0.1059	0.0175						
10	0.8914	0.1759	-0.1100	0.1140	0.0285						
15	0.8634	0.1363	-0.1223	0.1147	0.0417						
20	0.8656	0.1785	-0.1194	0.1082	0.0314						
25	0.8570	0.2208	-0.1153	0.1066	0.0617						
$M = 0.80$											
-10	-0.3974	0.0874	-0.0868	-0.0486	0.0098	-10	-0.6307	0.1408	-0.0044	-0.0712	0.0154
-7	-0.2178	0.0394	-0.1360	-0.0249	0.0041	-7	-0.4249	0.0841	-0.0404	-0.0475	0.0091
-5	0.0324	0.0197	-0.1421	-0.0034	0.0023	-5	-0.2684	0.0594	-0.0679	-0.0288	0.0062
-3	0.1281	0.0169	-0.1368	0.0153	0.0023	-3	-0.1118	0.0440	-0.0921	-0.0105	0.0047
-2	0.2311	0.0177	-0.1361	0.0262	0.0031	-2	-0.0067	0.0401	-0.1092	0.0014	0.0045
-1	0.3165	0.0206	-0.1352	0.0355	0.0040	-1	0.0917	0.0390	-0.1229	0.0131	0.0046
0	0.4062	0.0249	-0.1292	0.0556	0.0051	0	0.2125	0.0401	-0.1446	0.0268	0.0055
1	0.4931	0.0313	-0.1205	0.0547	0.0065	1	0.3511	0.0440	-0.1640	0.0424	0.0067
2	0.5785	0.0376	-0.1094	0.0652	0.0083	2	0.4652	0.0539	-0.1717	0.0538	0.0085
3	0.6727	0.0481	-0.1006	0.0759	0.0106	3	0.5591	0.0660	-0.1776	0.0641	0.0105
5	0.8596	0.0798	-0.1002	0.0985	0.0184						
$M = 0.90$											
-10	-0.5506	0.0854	-0.0814	-0.0678	0.0137	-10	-0.6091	0.1371	0.0024	-0.0686	0.0143
-7	-0.3433	0.0656	-0.1064	-0.0403	0.0070	-7	-0.4089	0.0836	-0.0389	-0.0462	0.0086
-5	-0.1684	0.0376	-0.1278	-0.0177	0.0040	-5	-0.2626	0.0588	-0.0630	-0.0291	0.0060
-3	0.0324	0.0249	-0.1461	0.0059	0.0031	-3	-0.1141	0.0439	-0.0858	-0.0111	0.0045
-2	0.1581	0.0236	-0.1525	0.0187	0.0034	-2	-0.0215	0.0384	-0.1003	0.0000	0.0043
-1	0.2695	0.0264	-0.1554	0.0309	0.0044	-1	0.2949	0.0402	-0.1687	0.0111	0.0045
0	0.3692	0.0306	-0.1482	0.0423	0.0057	0	0.1786	0.0429	-0.1280	0.0225	0.0052
1	0.4677	0.0376	-0.1479	0.0535	0.0073	1	0.2992	0.0466	-0.1444	0.0359	0.0064
2	0.5830	0.0472	-0.1482	0.0668	0.0097	2	0.3982	0.0545	-0.1533	0.0473	0.0082
3	0.6994	0.0603	-0.1525	0.0786	0.0122	3	0.5187	0.0667	-0.1697	0.0607	0.0100
$M = 1.00$											
-10	-0.6566	0.1477	-0.0032	-0.0744	0.0163						
-7	-0.4354	0.0876	-0.0438	-0.0495	0.0059						
-5	-0.2678	0.0590	-0.0756	-0.0288	0.0068						
-3	-0.0931	0.0430	-0.1069	-0.0080	0.0050						
-2	0.0116	0.0384	-0.1268	0.0044	0.0047						
-1	0.1281	0.0367	-0.1457	0.0177	0.0049						
0	0.2678	0.0389	-0.1674	0.0327	0.0058						
1	0.3958	0.0446	-0.1738	0.0463	0.0073						
2	0.5053	0.0550	-0.1769	0.0579	0.0092						
3	0.5984	0.0676	-0.1812	0.0680	0.0113						
$M = 1.05$											
-10	-0.6307	0.1408	-0.0044	-0.0712	0.0154						
-7	-0.4249	0.0841	-0.0404	-0.0475	0.0091						
-5	-0.2684	0.0594	-0.0679	-0.0288	0.0062						
-3	-0.1118	0.0440	-0.0921	-0.0105	0.0047						
-2	-0.0067	0.0401	-0.1092	0.0014	0.0045						
-1	0.0917	0.0390	-0.1229	0.0131	0.0046						
0	0.2125	0.0401	-0.1446	0.0268	0.0055						
1	0.3511	0.0440	-0.1640	0.0424	0.0067						
2	0.4652	0.0539	-0.1717	0.0538	0.0085						
3	0.5591	0.0660	-0.1776	0.0641	0.0105						
$M = 1.10$											
-10	-0.6091	0.1371	0.0024	-0.0686	0.0143						
-7	-0.4089	0.0836	-0.0389	-0.0462	0.0086						
-5	-0.2626	0.0588	-0.0630	-0.0291	0.0060						
-3	-0.1141	0.0439	-0.0858	-0.0111	0.0045						
-2	-0.0215	0.0384	-0.1003	0.0000	0.0043						
-1	0.2949	0.0402	-0.1687	0.0111	0.0045						
0	0.1786	0.0429	-0.1280	0.0225	0.0052						
1	0.2992	0.0466	-0.1444	0.0359	0.0064						
2	0.3982	0.0545	-0.1533	0.0473	0.0082						
3	0.5187	0.0667	-0.1697	0.0607	0.0100						

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TABLE 4-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

$$\frac{k}{c} = 0.04 \quad \frac{C_L}{c} = 0.30$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$
$M = 0.40$											
-10	-0.2995					-7	-0.3324	.0423	.0597	-0.046	.0100
-5	-0.0803					-5	-0.1247	.0290	.0097	-0.097	.0068
-3	-0.0459					-3	-0.0733	.0233	-0.0381	-0.134	.0058
-2	-0.2344					-2	-0.2114	.0236	-0.0701	-0.278	.0061
-1	-0.2952					-1	-0.3178	.0255	-0.0941	-0.392	.0071
0	-0.3646					0	-0.4253	.0293	-0.1177	-0.519	.0084
1	-0.4232					1	-0.5231	.0332	-0.1386	-0.630	.0104
2	-0.4797					2	-0.6355	.0394	-0.1637	-0.736	.0127
3	-0.5426										
5	-0.6685										
7	-0.7640										
10	-0.8400										
15	-0.8378										
20	-0.7944										
25	-0.7683										
$M = 0.60$											
-10	-0.3351	.0708	-0.0700	-0.0379	.0085	-7	-0.3459	.0873	-0.0828	-0.0376	.0101
-5	-0.1010	.0293	-0.1178	-0.0889	.0032	-5	-0.1735	.0636	-0.1164	-0.159	.0075
-3	-0.0577	.0190	-0.1162	.0108	.0022	-3	-0.0467	.0524	-0.1443	.0041	.0063
-2	-0.2073	.0224	-0.1188	.0273	.0026	-2	-0.1165	.0501	-0.1672	.0182	.0062
-1	-0.3007	.0259	-0.1117	.0370	.0032	-1	-0.2492	.0510	-0.1877	.0323	.0066
0	-0.3737	.0317	-0.1111	.0456	.0040	0	-0.3913	.0558	-0.1912	.0477	.0080
1	-0.4489	.0363	-0.1051	.0546	.0050	1	-0.4868	.0653	-0.1947	.0580	.0091
2	-0.5133	.0449	-0.1047	.0622	.0063	2	-0.5893	.0773	-0.1962	.0684	.0113
3	-0.5928	.0528	-0.0951	.0701	.0080						
5	-0.6551	.0639	-0.0931	.0782	.0099						
7	-0.7883	.0948	-0.0767	.0961	.0148						
10	-0.8871	.1321	-0.0845	.1095	.0209						
15	-0.9128	.2018	-0.1207	.1173	.0262						
20	-0.8871	.2936	-0.1441	.1180	.0451						
25	-0.8720	.3834	-0.1456	.1140	.0539						
	-0.8628	.4711	-0.1518	.1124	.0621						
$M = 0.80$											
-10	-0.3378	.0807	-0.0843	-0.0391	.0097	-7	-0.3377	.0841	-0.0787	-0.0366	.0090
-5	-0.1917	.0369	-0.1433	-0.0123	.0041	-5	-0.1789	.0632	-0.1086	-0.174	.0065
-3	-0.0611	.0232	-0.1449	.0095	.0028	-3	-0.0268	.0522	-0.1330	.0013	.0059
-2	-0.2156	.0246	-0.1423	.0156	.0035	-2	-0.0816	.0517	-0.1497	.0133	.0057
-1	-0.3076	.0283	-0.1397	.0374	.0043	-1	-0.1990	.0522	-0.1695	.0267	.0059
0	-0.3915	.0315	-0.1364	.0463	.0052	0	-0.3388	.0564	-0.1867	.0416	.0067
1	-0.4607	.0380	-0.1050	.0550	.0066	1	-0.4450	.0652	-0.1907	.0543	.0079
2	-0.5446	.0453	-0.1152	.0650	.0083	2	-0.5323	.0756	-0.1893	.0551	.0096
3	-0.6285	.0550	-0.1148	.0748	.0100	3	-0.6284	.1045	-0.1694	.0729	.0119
5	-0.7448	.0687	-0.1185	.0860	.0130						
	-0.9067	.1032	-0.1181	.1061	.0186						
$M = 0.90$											
-7	-0.2359	.0663	-0.1473	-0.0246	.0074	-7	-0.3337	.0807	-0.0725	-0.0363	.0081
-5	-0.0415	.0446	-0.1693	-0.0010	.0051	-5	-0.1873	.0614	-0.0997	-0.188	.0058
-3	-0.1594	.0389	-0.1826	.0226	.0049	-3	-0.0312	.0519	-0.1237	-0.0004	.0049
-2	-0.2761	.0427	-0.1847	.0344	.0057	-2	-0.0624	.0511	-0.1384	.0110	.0049
-1	-0.3797	.0462	-0.1843	.0462	.0069	-1	-0.1593	.0524	-0.1527	.0216	.0055
0	-0.4705	.0562	-0.1835	.0570	.0084	0	-0.2799	.0569	-0.1712	.0351	.0063
1	-0.5651	.0631	-0.1809	.0678	.0104	1	-0.4015	.0640	-0.1840	.0498	.0074
2	-0.6714	.0711	-0.1719	.0777	.0122	2	-0.4930	.0741	-0.1863	.0596	.0090
3	-0.7455	.0844	-0.1731	.0865	.0146	3	-0.5877	.0900	-0.1944	.0694	.0106
$M = 1.00$											
$M = 1.05$											
$M = 1.10$											

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TABLE 4.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

$$\frac{t}{c} = 0.04 \quad \frac{c_f}{c} = 0.40$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_l$	$C_n$
$M = 0.40$											
-10	-.1875					-10	-.4715	.1297	-.0965	-.0521	.0155
-7	-.0066					-7	-.2544	.0854	-.1422	-.0243	.0099
-5	.1300					-5	-.0310	.0635	-.1748	.0011	.0073
-3	.2423					-3	-.1799	.0565	-.2027	.0250	.0068
-2	.3084					-2	.3102	.0579	-.2060	.0392	.0075
-1	.3657					-1	.4343	.0641	-.2064	.0520	.0088
0	.4317					0	.5211	.0738	-.2010	.0623	.0104
1	.4846					1	.6179	.0839	-.2049	.0727	.0123
2	.5595					2	.7320	.0976	-.2099	.0823	.0147
3	.6080										
5	.7379										
7	.8018										
10	.8415										
15	.8150										
20	.7930										
25	.7820										
$M = 0.50$											
-10	-.2618	.0577	-.0727	-.0301	.0075	-10	-.4880	.1307	-.0669	-.0545	.0145
-7	-.0145	.0223	-.1128	-.0005	.0026	-7	-.2706	.0868	-.1152	-.0276	.0098
-5	.1527	.0201	-.1079	-.0204	.0024	-5	-.0969	.0677	-.1462	-.0061	.0076
-3	.3000	.0241	-.1072	-.0373	.0034	-3	.0945	.0601	-.1743	.0163	.0067
-2	.3928	.0300	-.0987	-.0473	.0044	-2	.2481	.0590	-.1954	.0330	.0070
-1	.4430	.0357	-.0963	-.0543	.0052	-1	.3722	.0639	-.2005	.0462	.0077
0	.5150	.0443	-.0910	-.0627	.0065	0	.4667	.0727	-.2003	.0572	.0090
1	.5804	.0524	-.0913	-.0708	.0082	1	.5636	.0828	-.1994	.0679	.0110
2	.6481	.0625	-.0819	-.0798	.0101	2	.6617	.0965	-.2020	.0769	.0132
3	.7266	.0759	-.0756	-.0887	.0125						
5	.8510	.1094	-.0483	-.1053	.0180						
7	.9099	.1556	-.0814	-.1155	.0245						
10	.9055	.2227	-.1228	-.1195	.0333						
15	.8750	.3187	-.1402	-.1155	.0452						
20	.8641	.4078	-.1414	-.1142	.0547						
25	.8641	.5011	-.1449	-.1139	.0658						
$M = 0.60$											
-10	-.2618	.0577	-.0727	-.0301	.0075	-10	-.4880	.1307	-.0669	-.0545	.0145
-7	-.0145	.0223	-.1128	-.0005	.0026	-7	-.2706	.0868	-.1152	-.0276	.0098
-5	.1527	.0201	-.1079	-.0204	.0024	-5	-.0969	.0677	-.1462	-.0061	.0076
-3	.3000	.0241	-.1072	-.0373	.0034	-3	.0945	.0601	-.1743	.0163	.0067
-2	.3928	.0300	-.0987	-.0473	.0044	-2	.2481	.0590	-.1954	.0330	.0070
-1	.4430	.0357	-.0963	-.0543	.0052	-1	.3722	.0639	-.2005	.0462	.0077
0	.5150	.0443	-.0910	-.0627	.0065	0	.4667	.0727	-.2003	.0572	.0090
1	.5804	.0524	-.0913	-.0708	.0082	1	.5636	.0828	-.1994	.0679	.0110
2	.6481	.0625	-.0819	-.0798	.0101	2	.6617	.0965	-.2020	.0769	.0132
3	.7266	.0759	-.0756	-.0887	.0125						
5	.8510	.1094	-.0483	-.1053	.0180						
7	.9099	.1556	-.0814	-.1155	.0245						
10	.9055	.2227	-.1228	-.1195	.0333						
15	.8750	.3187	-.1402	-.1155	.0452						
20	.8641	.4078	-.1414	-.1142	.0547						
25	.8641	.5011	-.1449	-.1139	.0658						
$M = 0.80$											
-10	-.2728	.0717	-.0839	-.0314	.0087	-10	-.4732	.1254	-.0638	-.0532	.0132
-7	-.0514	.0244	-.1475	-.0216	.0039	-7	-.2750	.0682	-.1123	-.0288	.0086
-5	.1532	.0300	-.1482	-.0200	.0033	-5	-.1123	.0684	-.1356	-.0088	.0066
-3	.3139	.0342	-.1455	-.0381	.0048	-3	.0624	.0611	-.1616	.0119	.0060
-2	.3961	.0397	-.1379	-.0477	.0057	-2	.1872	.0608	-.1800	.0260	.0062
-1	.4634	.0448	-.1375	-.0569	.0070	-1	.3159	.0636	-.1946	.0406	.0066
0	.5456	.0493	-.1335	-.0684	.0084	0	.4256	.0726	-.1993	.0532	.0078
1	.6203	.0616	-.1235	-.0762	.0105	1	.5183	.0832	-.1953	.0634	.0093
2	.7173	.0726	-.1158	-.0864	.0128	2	.6128	.0957	-.1997	.0725	.0112
3	.4081	.0854	-.0715	-.0959	.0154						
$M = 0.90$											
-10	-.3355	.1045	-.1479	-.0368	.0122	-10	-.4479	.1168	-.0677	-.0512	.0125
-7	-.1086	.0663	-.1797	-.0089	.0075	-7	-.2655	.0806	-.1014	-.0283	.0079
-5	.0921	.0518	-.2023	-.0141	.0057	-5	-.1092	.0664	-.1303	-.0090	.0060
-3	.2829	.0501	-.2057	-.0352	.0065	-3	.0546	.0591	-.1536	.0101	.0056
-2	.3842	.0557	-.1981	-.0465	.0077	-2	.1584	.0599	-.1666	.0224	.0061
-1	.4684	.0631	-.1976	-.0573	.0092	-1	.2677	.0645	-.1802	.0350	.0066
0	.5592	.0712	-.1929	-.0681	.0108	0	.3824	.0714	-.1880	.0486	.0075
1	.6579	.0799	-.1873	-.0780	.0129	1	.4807	.0819	-.1906	.0598	.0091
2	.7553	.0899	-.1808	-.0870	.0148	2	.5856	.0959	-.1912	.0698	.0107
$M = 1.00$											
-10	-.4715	.1297	-.0965	-.0521	.0155						
-7	-.2544	.0854	-.1422	-.0243	.0099						
-5	-.0310	.0635	-.1748	-.0011	.0073						
-3	-.1799	.0565	-.2027	.0250	.0068						
-2	.3102	.0579	-.2060	.0392	.0075						
-1	.4343	.0641	-.2064	.0520	.0088						
0	.5211	.0738	-.2010	.0623	.0104						
1	.6179	.0839	-.2049	.0727	.0123						
2	.7320	.0976	-.2099	.0823	.0147						
$M = 1.05$											
-10	-.4732	.1254	-.0638	-.0532	.0132						
-7	-.2750	.0682	-.1123	-.0288	.0086						
-5	-.1123	.0684	-.1356	-.0088	.0066						
-3	.0624	.0611	-.1616	.0119	.0060						
-2	.1872	.0608	-.1800	.0260	.0062						
-1	.3159	.0636	-.1946	.0406	.0066						
0	.4256	.0726	-.1993	.0532	.0078						
1	.5183	.0832	-.1953	.0634	.0093						
2	.6128	.0957	-.1997	.0725	.0112						
$M = 1.10$											
-10	-.4479	.1168	-.0677	-.0512	.0125						
-7	-.2655	.0806	-.1014	-.0283	.0079						
-5	-.1092	.0664	-.1303	-.0090	.0060						
-3	.0546	.0591	-.1536	.0101	.0056						
-2	.1584	.0599	-.1666	.0224	.0061						
-1	.2677	.0645	-.1802	.0350	.0066						
0	.3824	.0714	-.1880	.0486	.0075						
1	.4807	.0819	-.1906	.0598	.0091						
2	.5856	.0959	-.1912	.0698	.0107						

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TABLE 4.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

 $\frac{t}{c} = 0.06 \quad \frac{C_L}{C_D} = \text{NONE}$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$											
-10	-0.5810					-10	-0.8571	.1739	.1523	-0.1044	-0.0258
-7	-0.4900					-7	-0.6015	.0958	.0765	-0.0726	-0.0151
-5	-0.3240					-5	-0.4426	.0628	.0438	-0.0537	-0.0088
-3	-0.2008					-3	-0.2805	.0421	.0266	-0.0340	-0.0052
-2	-0.1339					-2	-0.1714	.0299	.0041	-0.0208	-0.0034
-1	-0.0803					-1	-0.0841	.0284	.0041	-0.0102	-0.0018
0	-0.0134					0	-0.0000	.0284	.0024	0.0000	-0.0015
1	.0536					1	.0873	.0284	-0.0027	.0106	-0.0015
2	.1125					2	.1714	.0337	-0.0079	.0204	-0.0019
3	.1767					3	.2649	.0452	-0.0183	.0318	-0.0027
5	.2972					5	.4207	.0667	-0.0338	.0507	-0.0046
7	.4338					7	.5922	.1088	-0.0717	.0711	-0.0088
10	.6159					10	.8633	.1917	-0.1475	.1029	-0.0387
15	.7230					15	1.1376	.3425	-0.1985	.1392	-0.0379
20	.7256					20	1.1064	.4399	-0.2054	.1324	-0.0509
25	.7149					25	1.1064	.5702	-0.2399	.1343	-0.0663
$M = 0.60$											
-10	-0.6492	.1276	.0447	.0823	-.0172	-10	-0.7676	.1660	.1216	-0.0921	-0.0238
-7	-0.5002	.0585	-.0079	-.0602	-.0083	-7	-0.5397	.1003	.0769	-0.0648	-0.0142
-5	-0.3635	.0282	-.0048	-.0428	-.0041	-5	-0.3988	.0685	.0477	-0.0473	-0.0099
-3	-0.2214	.0161	-.0048	-.0254	-.0021	-3	-0.2519	.0487	.0249	-0.0306	-0.0064
-2	-0.1449	.0115	-.0060	-.0166	-.0012	-2	-0.1709	.0435	.0146	-0.0211	-0.0047
-1	-0.0724	.0101	-.0012	-.0083	0.0000	-1	-0.0930	.0391	.0080	-0.0120	-0.0030
0	0.0000	.0074	-.0043	-.0003	0.0000	0	-0.0030	.0384	.0040	-0.0018	-0.0025
1	.0752	.0082	.0012	.0095	.0002	1	.0840	.0399	-0.0023	.0084	-0.0030
2	.1435	.0107	.0000	.0168	.0003	2	.1649	.0442	-0.0100	.0186	-0.0030
3	.2187	.0148	-.0012	.0257	.0003	3	.2519	.0523	-0.0215	.0288	-0.0035
5	.3567	.0262	.0024	.0423	-.0006	5	.3958	.0723	-0.0421	.0448	-0.0051
7	.5057	.0552	-.0012	.0599	-.0018	7	.5487	.1069	-0.0766	.0630	-0.0086
10	.6970	.1225	-.0290	.0839	-.0088	10	.7886	.1748	-0.1307	.0913	-0.0264
15	.7790	.2318	-.1004	.1002	-.0230	15	1.1124	.3310	-0.2142	.1332	-0.0356
20	.7653	.2520	-.1288	.0979	-.0347	20	1.3014	.5087	-0.2374	.1387	-0.0589
25	.7817	.4141	-.1305	.0975	-.0464	25	1.2234	.6105	-0.2739	.1514	-0.0729
$M = 0.80$											
-10	-0.7464	.1249	.0253	-.0931	-.0186	-10	-0.7513	.1634	.1275	-.0905	-0.0241
-7	-0.5820	.0623	.0029	.0702	-.0108	-7	-0.5519	.0995	.0812	-.0662	-0.0151
-5	-0.4379	.0327	-.0135	-.0511	-.0054	-5	-0.4103	.0696	.0543	-.0491	-0.0108
-3	-0.2568	.0155	-.0163	-.0296	0.0026	-3	-0.2572	.0497	.0336	-.0316	-0.0066
-2	-0.1755	.0122	-.0094	-.0200	0.0014	-2	-0.1734	.0455	.0236	-.0214	-0.0052
-1	-0.0784	.0105	-.0041	-.0090	0.0004	-1	-0.0809	.0426	.0146	-.0130	-0.0039
0	0.0000	.0091	-.0033	0.0005	-.0003	0	-0.0058	.0412	.0038	-.0021	-0.0024
1	.0868	.0109	-.0024	.0101	-.0001	1	.0867	.0419	-.0083	0.0084	-0.0037
2	.1718	.0140	.0021	.0197	-.0001	2	.1734	.0470	-.0186	.0182	-0.0038
3	.2531	.0190	.0090	.0291	-.0002	3	.2572	.0548	-.0291	.0281	-0.0041
5	.4249	.0373	.0123	.0489	-.0015	5	.4103	.0746	-.0495	.0456	-0.0060
7	.5820	.0722	-.0024	.0695	-.0042	7	.5548	.1086	-.0754	.0635	-0.0096
10	.7484	.1339	-.0257	.0919	-.0116	10	.7542	.1719	-.1247	.0880	-0.0171
15	.7945	.2444	-.1075	.1005	-.0518	15	1.0836	.3304	-.2116	.1287	-0.0364
20	.8129	.3379	-.1414	.1009	-.0385	20	1.3206	.5343	-.2749	.1589	-0.0612
25	.8406	.4397	-.1504	.1031	-.0511	25	1.2425	.6295	-.2845	.1620	-0.0791
$M = 0.90$											
-10	-0.7844	.1461	.0824	-.0996	-.0225	-10	-0.7282	.1566	.1277	-.0897	-0.0245
-7	-0.6072	.0827	.0515	-.0757	-.0135	-7	-0.5399	.0940	.0818	-.0662	-0.0156
-5	-0.4595	.0504	.0261	-.0558	-.0072	-5	-0.4070	.0647	.0588	-.0507	-0.0114
-3	-0.2855	.0263	.0116	-.0351	0.0032	-3	-0.2492	.0476	.0368	-.0319	-0.0073
-2	-0.1969	.0190	.0043	-.0231	0.0022	-2	-0.1661	.0422	.0258	-.0225	-0.0060
-1	-0.0985	.0169	.0026	-.0120	0.0012	-1	-0.0831	.0408	.0144	-.0131	-0.0044
0	.0098	.0138	-.0011	0.0008	-.0006	0	0.0000	.0401	.0022	-.0034	-0.0041
1	.0985	.0158	-.0047	.0120	-.0008	1	.0914	.0408	-.0086	.0067	-0.0044
2	.1969	.0210	-.0083	.0231	-.0007	2	.1661	.0450	-.0193	.0161	-0.0043
3	.2954	.0290	-.0149	.0351	-.0010	3	.2975	.0525	-.0300	.0252	-0.0047
5	.4529	.0525	-.0283	.0546	-.0031	5	.4070	.0728	-.0530	.0427	-0.0064
7	.6138	.0916	-.0519	.0749	-.0073	7	.5538	.1055	-.0799	.0605	-0.0105
10	.7976	.1574	-.0845	.0992	-.0156	10	.7421	.1668	-.1203	.0897	-0.0174
15	.9682	.2801	-.1245	.1215	-.0312	15	1.0522	.3180	-.2058	.1216	-0.0355
20	.9288	.3786	-.1735	.1139	-.0431	20	1.2820	.5161	-.2714	.1512	-0.0589
25	.9846	.4956	-.1910	.1155	-.0569						

TABLE 4. - THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

 $\frac{k}{c} = 0.06 \quad \frac{C_L^2}{c} = 0.20$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$											
-10	-0.2633					-10	-0.6336	.1469	.0135	-0.0761	-0.0205
-7	-0.0967					-7	-0.4396	.0900	-.0235	-0.0516	-0.0126
-5	.0242					-5	-0.2941	.0562	-.0460	-0.0336	-0.0076
-3	.1686					-3	-0.1889	.0462	-.0737	-0.0122	-0.0045
-2	.2257					-2	-0.1110	.0438	-0.1014	.0023	-0.0036
-1	.2982					-1	.1549	.0454	-.1377	.0186	-0.0035
0	.3519					0	.2879	.0538	-.1654	.0338	-0.0045
1	.4137					1	.4177	.0630	-.1876	.0490	-0.0060
2	.4594					2	.5225	.0385	-.1751	.0613	-0.0077
3	.5215					3	.6242	.0393	-.2167	.0737	-0.0098
5	.6340					5	.7776	.1284	-.2339	.0923	-0.0145
7	.7415					7	.9356	.2570	-.2561	.1097	-0.0208
10	.8650					10	1.1271	.2671	-.2893	.1321	-0.0319
15	.9919					15	1.3486	.4309	-.3101	.1606	-0.0544
20	.8465					20	1.2015	.5110	-.2963	.1416	-0.0644
25	.8194					25	1.1390	.6109	-.2893	.1341	-0.0771
$M = 0.60$											
-10	-0.3235	.0780	-.0806	-.0389	-.0096	-10	-0.6129	.1441	.0213	-.0727	-0.0200
-7	-0.1110	.0236	-.1280	-.0115	-.0020	-7	-0.4146	.0539	-.0259	-0.0465	-0.0130
-5	.0234	.0167	-.1292	-.0003	-.0032	-5	-0.2569	.0709	-.0541	-0.0288	-0.0085
-3	.1590	.0181	-.1261	.0196	-.0007	-3	-0.0916	.0584	-.0834	-.0089	-0.0059
-2	.2207	.0203	-.1231	.0263	-.0006	-2	.0000	.0559	-.0930	.0015	-0.0052
-1	.2953	.0236	-.1219	.0346	-.0009	-1	.1067	.0547	-.1173	.0135	-0.0052
0	.3646	.0296	-.1213	.0432	-.0012	0	.2238	.0569	-.1389	.0261	-0.0052
1	.4674	.0370	-.1280	.0562	-.0018	1	.3545	.0643	-.1648	.0407	-0.0060
2	.5236	.0430	-.1189	.0625	-.0027	2	.4567	.0754	-.1868	.0534	-0.0076
3	.5880	.0518	-.1170	.0700	-.0035	3	.5618	.0916	-.2047	.0662	-0.0092
5	.7320	.0768	-.1098	.0818	-.0067	5	.7301	.1256	-.2313	.0853	-0.0135
7	.8526	.1179	-.1122	.1011	-.0115	7	.8893	.1752	-.2519	.1032	-0.0197
10	.9239	.1900	-.1503	.1131	-.0210	10	1.0636	.2564	-.2811	.1243	-0.0301
15	.9431	.3013	-.1898	.1201	-.0359	15	1.3040	.4241	-.3210	.1542	-0.0261
20	.9019	.4195	-.2007	.1115	-.0488	20	1.4241	.5940	-.3369	.1725	-0.0375
25	.8965	.2434	-.2086	.1091	-.0604						
$M = 0.80$											
-10	-0.4393	.0875	-.0906	-.0528	-.0116	-10	-0.5851	.1389	.0176	-.0691	-0.0193
-7	-0.2465	.0415	-.1209	-.0283	-.0054	-7	-0.3953	.0905	-.0250	-0.0455	-0.0129
-5	-0.0760	.0247	-.1316	-.0087	-.0022	-5	-0.2491	.0691	-.0506	-.0283	-0.0085
-3	.1094	.0204	-.1332	.0131	-.0010	-3	-0.0912	.0584	-.0779	-.0097	-0.0062
-2	.1983	.0228	-.1304	.0232	-.0010	-2	.0058	.0549	-.0938	.0012	-0.0055
-1	.2855	.0282	-.1341	.0335	-.0013	-1	.1028	.0556	-.1099	.0128	-0.0052
0	.3785	.0337	-.1353	.0439	-.0021	0	.2013	.0591	-.1275	.0259	-0.0054
1	.4671	.0419	-.1398	.0547	-.0027	1	.3027	.0669	-.1467	.0357	-0.0062
2	.5728	.0510	-.1407	.0677	-.0040	2	.4011	.0762	-.1653	.0471	-0.0075
3	.6655	.0625	-.1365	.0783	-.0053	3	.5127	.0898	-.1845	.0596	-0.0093
5	.8008	.0921	-.1365	.0952	-.0089	5	.6821	.1240	-.2210	.0801	-0.0136
7	.9157	.1368	-.1427	.1116	-.0146	7	.8428	.1869	-.2441	.0988	-0.0194
10	1.0529	.1172	-.1530	.1291	-.0239	10	1.0195	.2472	-.2678	.1195	-0.0300
15	.9750	.3146	-.2115	.1192	-.0387	15	1.2657	.4188	-.3216	.1497	-0.0512
20	.9605	.4139	-.2190	.1152	-.0457	20	1.4308	.6124	-.3601	.1722	-0.0766
25	.9602	.5196	-.2305	.1138	-.0645						
$M = 0.90$											
-10	-0.5699	.1244	-.0248	-.0722	-.0244	-10	-0.5632	.1351	.0206	-.0658	-0.0191
-7	-0.3936	.0697	-.0557	-.0464	-.0091	-7	-0.3870	.0866	-.0199	-0.0436	-0.0124
-5	-0.2273	.0445	-.0867	-.0284	-.0052	-5	-0.2469	.0683	-.0454	-.0274	-0.0090
-3	.0000	.0308	-.1286	.0006	-.0024	-3	-0.0916	.0559	-.0746	-.0088	-.0065
-2	.1301	.0311	-.1469	.0156	-.0023	-2	.0042	.0526	-.0899	.0013	-0.0058
-1	.2372	.0352	-.1523	.0278	-.0024	-1	.0943	.0540	-.1053	.0114	-0.0056
0	.3278	.0413	-.1534	.0390	-.0031	0	.1859	.0259	-.1218	.0219	-0.0056
1	.4548	.0511	-.1577	.0514	-.0043	1	.2816	.0427	-.1374	.0327	-0.0063
2	.5287	.0616	-.1650	.0622	-.0043	2	.3787	.0730	-.1553	.0439	-0.0076
3	.6325	.0777	-.1741	.0746	-.0074	3	.4689	.0853	-.1731	.0547	-0.0090
5	.7791	.1110	-.1948	.0932	-.0120	5	.6381	.1160	-.2056	.0744	-0.0131
7	.9289	.1608	-.2098	.1111	-.0187	7	.8018	.1618	-.2320	.0929	-0.0190
10	1.0805	.2365	-.2244	.1307	-.0291	10	.9683	.2354	-.2596	.1121	-0.0285
15	1.1602	.3500	-.2426	.1331	-.0440	15	1.2041	.3984	-.3081	.1421	-0.0490
20	1.0574	.4488	-.2500	.1251	-.0565						
25	1.0343	.5574	-.2594	.1231	-.0699						
$M = 1.00$											
-10	-0.5632	.1351	-.0206	-.0658	-.0191						
-7	-0.3870	.0866	-.0199	-0.0436	-0.0124						
-5	-0.2469	.0683	-.0454	-.0274	-0.0090						
-3	-0.0916	.0559	-.0746	-.0088	-.0065						
-2	.0042	.0526	-.0899	.0013	-0.0058						
-1	.0943	.0540	-.1053	.0114	-0.0056						
0	.1859	.0259	-.1218	.0219	-0.0056						
1	.2816	.0427	-.1374	.0327	-0.0063						
2	.3787	.0730	-.1553	.0439	-0.0076						
3	.4689	.0853	-.1731	.0547	-0.0090						
5	.6381	.1160	-.2056	.0744	-0.0131						
7	.8018	.1618	-.2320	.0929	-0.0190						
10	.9683	.2354	-.2596	.1121	-0.0285						
15	1.2041	.3984	-.3081	.1421	-0.0490						

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TABLE 4-- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Continued

$\frac{t}{c} = 0.06$        $\frac{C_L}{c} = 0.30$

$c_a$ deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$c_a$ deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$											
-10	-0.2280					-10	-0.5561	.1403	-0.0377	-0.0654	-0.0192
-7	-0.0510					-7	-0.3530	.0911	-0.0757	-0.0389	-0.0116
-5	.1019					-5	-0.1968	.0692	-0.0982	-0.0203	-0.0076
-3	.2360					-3	.0359	.0550	-0.1448	.0047	-0.0050
-2	.2816					-2	.1718	.0576	-0.1741	.0224	-0.0050
-1	.3406					-1	.3046	.0669	-0.2066	.0379	-0.0055
0	.4103					0	.4280	.0750	-0.2163	.0521	-0.0070
1	.4613					1	.5264	.0884	-0.2260	.0711	-0.0085
2	.5176					2	.6217	.1018	-0.2308	.0745	-0.0103
3	.5686					3	.7107	.1190	-0.2412	.0883	-0.0128
5	.6839					5	.8497	.1567	-0.2550	.1024	-0.0171
7	.8046					7	.9934	.2127	-0.2723	.1202	-0.0238
10	.8958					10	1.1558	.2957	-0.2826	.1395	-0.0348
15	.8931					15	1.3433	.4533	-0.2965	.1649	-0.0554
20	.8556					20	1.0215	.5223	-0.2792	.1422	-0.0641
25	.8234					25	1.0777	.5946	-0.2619	.1346	-0.0763
$M = 0.60$											
-10	-0.2669	.0775	-0.0973	-0.0332	-0.0074	-10	-0.5474	.1390	-0.0252	-0.0641	-0.0183
-7	-0.0520	.0268	-0.1266	-0.0050	-0.0014	-7	-0.3449	.0940	-0.0683	-0.0382	-0.0115
-5	.0986	.0222	-0.1296	.0191	-0.0006	-5	-0.1920	.0741	-0.0975	-0.0191	-0.0081
-3	.2332	.0249	-0.1338	.0309	-0.0006	-3	-0.0075	.0645	-0.1290	.0027	-0.0065
-2	.3409	.0277	-0.1278	.0399	-0.0006	-2	.0975	.0645	-0.1483	.0151	-0.0059
-1	.4011	.0337	-0.1326	.0478	-0.0009	-1	.2235	.0678	-0.1719	.0273	-0.0059
0	.4750	.0397	-0.1326	.0565	-0.0014	0	.3599	.0748	-0.2017	.0437	-0.0067
1	.5311	.0452	-0.1254	.0638	-0.0024	1	.4724	.0859	-0.2176	.0573	-0.0085
2	.5914	.0517	-0.1193	.0706	-0.0033	2	.5624	.0991	-0.2249	.0683	-0.0102
3	.6571	.0641	-0.1144	.0791	-0.0047	3	.6524	.1162	-0.2342	.0790	-0.0122
5	.8022	.0909	-0.1115	.0944	-0.0077	5	.8059	.1530	-0.2481	.0965	-0.0171
7	.9117	.1339	-0.1115	.1090	-0.0091	7	.9568	.2080	-0.2674	.1147	-0.0235
10	.9500	.2086	-0.1508	.1173	-0.0233	10	1.1218	.2942	-0.2926	.1351	-0.0344
15	.9309	.3118	-0.1872	.1180	-0.0366	15	1.3498	.4646	-0.3244	.1638	-0.0570
20	.9035	.4079	-0.1932	.1123	-0.0490						
25	.8980	.5021	-0.1992	.1100	-0.0606						
$M = 0.80$											
-10	-0.3339	.0820	-0.1223	-0.0387	-0.0097	-10	-0.5349	.1365	-0.0230	-0.0614	-0.0187
-7	-0.1471	.0409	-0.1493	-0.0146	-0.0043	-7	-0.3383	.0921	-0.0669	-0.0368	-0.0117
-5	.0296	.0259	-0.1530	.0056	-0.0017	-5	-0.1879	.0729	-0.0892	-0.0193	-0.0087
-3	.2007	.0305	-0.1490	.0261	-0.0017	-3	-0.0101	.0622	-0.1218	.0018	-0.0065
-2	.2821	.0340	-0.1465	.0344	-0.0020	-2	.0896	.0619	-0.1388	.0132	-0.0061
-1	.3691	.0409	-0.1498	.0449	-0.0026	-1	.1981	.0653	-0.1548	.0254	-0.0065
0	.4468	.0477	-0.1477	.0544	-0.0035	0	.3036	.0711	-0.1778	.0377	-0.0071
1	.5421	.0577	-0.1477	.0476	-0.0045	1	.4236	.0818	-0.1989	.0514	-0.0088
2	.6364	.0668	-0.1436	.0761	-0.0058	2	.5104	.0935	-0.2104	.0619	-0.0103
3	.7160	.0786	-0.1346	.0853	-0.0072	3	.6130	.1099	-0.2264	.0746	-0.0120
5	.8529	.1123	-0.1334	.1033	-0.0115	5	.7634	.1450	-0.2424	.0912	-0.0166
7	.9805	.1560	-0.1407	.1203	-0.0168	7	.9166	.1976	-0.2584	.1095	-0.0227
10	1.0915	.2298	-0.1518	.1370	-0.0267	10	1.0843	.2787	-0.2820	.1291	-0.0340
15	.9324	.3230	-0.1981	.1167	-0.0387	15	1.3012	.4508	-0.3288	.1579	-0.0279
20	.9620	.4285	-0.2177	.1172	-0.0424						
25	.9435	.5332	-0.2251	.1150	-0.0324						
$M = 0.90$											
-10	-0.4884	.1172	-0.0706	-0.0599	-0.0157	-10	-0.5055	.1294	-0.0248	-0.0585	-0.0175
-7	-0.3206	.0715	-0.0979	-0.0359	-0.0087	-7	-0.3158	.1032	-0.0640	-0.0348	-0.0114
-5	-1.233	.0477	-0.1306	-0.0120	-0.0045	-5	-0.1731	.0698	-0.0901	-0.0177	-0.0083
-3	.1019	.0416	-0.1626	.0140	-0.0029	-3	.0028	.0607	-0.1207	.0029	-0.0066
-2	.2138	.0424	-0.1688	.0269	-0.0032	-2	.0900	.0596	-0.1330	.0135	-0.0064
-1	.3174	.0485	-0.1688	.0385	-0.0043	-1	.1842	.0633	-0.1474	.0237	-0.0067
0	.4061	.0554	-0.1706	.0493	-0.0050	0	.2839	.0698	-0.1679	.0361	-0.0073
1	.4982	.0668	-0.1713	.0603	-0.0062	1	.3809	.0789	-0.1832	.0471	-0.0085
2	.6002	.0809	-0.1778	.0718	-0.0080	2	.4778	.0913	-0.1992	.0583	-0.0097
3	.6824	.0950	-0.1876	.0828	-0.0102	3	.5678	.1046	-0.2146	.0689	-0.0115
5	.8386	.1314	-0.1971	.1014	-0.0151	5	.7257	.1403	-0.2322	.0874	-0.0159
7	.9669	.1835	-0.2139	.1185	-0.0226	7	.8670	.1907	-0.2487	.1042	-0.0218
10	1.1017	.2619	-0.2248	.1377	-0.0325	10	1.0304	.2670	-0.2708	.1230	-0.0325
15	1.0162	.3453	-0.2284	.1253	-0.0435	15	1.2548	.4386	-0.3223	.1513	-0.0541
20	1.0491	.4690	-0.2473	.1277	-0.0587						
25	1.0326	.5726	-0.2539	.1253	-0.0723						
$M = 1.10$											
-10	-0.5055	.1294	-0.0248	-0.0585	-0.0175						
-7	-0.3158	.1032	-0.0640	-0.0348	-0.0114						
-5	-0.1731	.0698	-0.0901	-0.0177	-0.0083						
-3	.0028	.0607	-0.1207	.0029	-0.0066						
-2	.0900	.0596	-0.1330	.0135	-0.0064						
-1	.1842	.0633	-0.1474	.0237	-0.0067						
0	.2839	.0698	-0.1679	.0361	-0.0073						
1	.3809	.0789	-0.1832	.0471	-0.0085						
2	.4778	.0913	-0.1992	.0583	-0.0097						
3	.5678	.1046	-0.2146	.0689	-0.0115						
5	.7257	.1403	-0.2322	.0874	-0.0159						
7	.8670	.1907	-0.2487	.1042	-0.0218						
10	1.0304	.2670	-0.2708	.1230	-0.0325						
15	1.2548	.4386	-0.3223	.1513	-0.0541						

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TABLE 4.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 4 MODEL - Concluded

 $\frac{c}{c} = 0.06$      $\frac{c_f}{c} = 0.40$ 

$a,$ deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$	$a,$ deg	$c_L$	$c_D$	$c_M$	$c_l$	$c_n$
$M = 0.40$											
$M = 0.55$											
-10	.1719					-10	-.4769	.1280	-.0602	-.0543	-.0180
-7	.0081					-7	-.2533	.0857	-.1121	-.0254	-.0117
-5	.1478					-5	-.0547	.0696	-.1557	-.0025	-.0080
-3	.2767					-3	.1845	.0605	-.1971	.0245	-.0070
-2	.3439					-2	.3064	.0727	-.2109	.0381	-.0073
-1	.4057					-1	.4096	.0804	-.2213	.0495	-.0083
0	.4513					0	.5253	.0915	-.2241	.0607	-.0095
1	.5131					1	.5957	.1069	-.2331	.0713	-.0109
2	.5668					2	.6895	.1199	-.2331	.0818	-.0126
3	.6260					3	.7735	.1391	-.2414	.0922	-.0148
5	.7415					5	.9256	.1810	-.2608	.1097	-.0199
7	.8462					7	1.0725	.2384	-.2712	.1249	-.0265
10	.9053					10	1.1882	.3175	-.2850	.1408	-.0364
15	.8624					15	1.3665	.4783	-.2988	.1643	-.0573
20	.8465					20	1.1976	.5505	-.2884	.1408	-.0658
25	.8113					25	1.1194	.6351	-.2760	.1302	-.0767
$M = 0.60$											
$M = 0.95$											
-10	-.2045	.0513	-.0861	-.0248	-.0054	-10	-.4834	.1237	-.0525	-.0521	-.0172
-7	.0206	.0189	-.1182	.0035	-.0018	-7	-.2417	.0868	-.1036	-.0253	-.0112
-5	.1659	.0189	-.1206	.0203	-.0010	-5	-.0826	.0739	-.1388	-.0062	-.0087
-3	.3166	.0255	-.1225	.0376	-.0008	-3	.1066	.0706	-.1707	.0153	-.0076
-2	.3989	.0291	-.1213	.0459	-.0010	-2	.2417	.0739	-.1940	.0299	-.0076
-1	.4510	.0365	-.1201	.0539	-.0016	-1	.3678	.0805	-.2125	.0445	-.0081
0	.5089	.0430	-.1177	.0607	-.0019	0	.4624	.0898	-.2205	.0554	-.0093
1	.5730	.0526	-.1170	.0689	-.0022	1	.5600	.1027	-.2251	.0663	-.0107
2	.6456	.0628	-.1140	.0769	-.0028	2	.6425	.1174	-.2305	.0763	-.0123
3	.7210	.0762	-.1110	.0852	-.0042	3	.7296	.1351	-.2371	.0864	-.0143
5	.8307	.1031	-.1074	.0985	-.0073	5	.8797	.1776	-.2544	.1035	-.0196
7	.9294	.1516	-.1153	.1118	-.0128	7	1.0058	.2348	-.2756	.1188	-.0259
10	.9540	.2245	-.1529	.1184	-.0223	10	1.1820	.3226	-.3002	.1381	-.0365
15	.9074	.3208	-.1826	.1138	-.0362	15	1.3601	.4917	-.3175	.1625	-.0581
20	.8937	.4153	-.1886	.1095	-.0478	20	1.4202	.6541	-.3281	.1680	-.0777
25	.8855	.5151	-.2001	.1075	-.0598	25	1.3151	.7707	-.3387	.1552	-.0943
$M = 0.80$											
$M = 1.05$											
-10	-.2529	.0697	-.1176	-.0283	-.0084	-10	-.4456	.1199	-.0480	-.0507	-.0175
-7	.0491	.0374	-.1505	-.0030	-.0039	-7	-.2488	.0839	-.0954	-.0507	-.0113
-5	.1204	.0324	-.1537	.0159	-.0024	-5	-.0926	.0722	-.1293	-.0077	-.0087
-3	.2761	.0400	-.1525	.0341	-.0024	-3	.0810	.0680	-.1581	.0121	-.0080
-2	.3484	.0456	-.1525	.0430	-.0032	-2	.1924	.0705	-.1773	.0248	-.0081
-1	.4290	.0524	-.1508	.0524	-.0040	-1	.3110	.0765	-.1965	.0379	-.0086
0	.5152	.0619	-.1517	.0616	-.0051	0	.4297	.0858	-.2125	.0520	-.0097
1	.5874	.0702	-.1463	.0699	-.0058	1	.5281	.0982	-.2208	.0630	-.0109
2	.6504	.0775	-.1345	.0778	-.0066	2	.6149	.1128	-.2278	.0727	-.0125
3	.7468	.0930	-.1340	.0891	-.0082	3	.7017	.1309	-.2349	.0827	-.0146
5	.8058	.1268	-.1369	.1077	-.0115	5	.8478	.1707	-.2528	.0997	-.0194
7	1.0118	.1768	-.1465	.1241	-.0174	7	.9722	.2248	-.2707	.1148	-.0130
10	1.1007	.2533	-.1565	.1376	-.0269	10	1.1313	.3138	-.2957	.1338	-.0184
15	.9599	.3417	-.1975	.1174	-.0393	15	1.3541	.4923	-.3380	.1608	-.0589
20	.9673	.4466	-.2107	.1196	-.1039	20	1.4988	.6759	-.3546	.1784	-.0824
25	.9488	.5504	-.2205	.1138	-.0651	25	1.3946	.7912	-.3584	.1643	-.0993
$M = 0.90$											
$M = 1.10$											
-10	-.3905	.1065	-.0991	-.0454	-.0151	-10	-.4324	.1193	-.0460	-.0488	-.0168
-7	.2009	.0713	-.1319	-.0194	-.0078	-7	-.2453	.0818	-.0901	-.0259	-.0115
-5	.0165	.0547	-.1683	.0054	-.0053	-5	-.0970	.0703	-.1202	-.0086	-.0089
-3	.2371	.0583	-.1909	.0305	-.0055	-3	.0804	.0668	-.1533	.0113	-.0081
-2	.3342	.0644	-.1930	.0416	-.0060	-2	.1885	.0771	-.1711	.0230	-.0084
-1	.4093	.0685	-.1872	.0502	-.0071	-1	.2827	.0750	-.1870	.0346	-.0090
0	.5071	.0822	-.1945	.0611	-.0082	0	.3950	.0825	-.2023	.0476	-.0101
1	.5894	.0924	-.1909	.0711	-.0095	1	.4989	.0941	-.2128	.0585	-.0114
2	.6767	.1044	-.1887	.0818	-.0109	2	.5834	.1091	-.2201	.0691	-.0130
3	.7528	.1207	-.1981	.0973	-.0127	3	.6652	.1247	-.2262	.0784	-.0151
5	.8759	.1551	-.1981	.1063	-.0168	5	.8038	.1621	-.2404	.0949	-.0201
7	1.0044	.2114	-.2164	.1223	-.0232	7	.9313	.2133	-.2606	.1097	-.0263
10	1.1394	.2939	-.2382	.1403	-.0337	10	1.0810	.2985	-.2881	.1278	-.0367
15	1.0439	.2923	-.2309	.1251	-.0440	15	1.2916	.4676	-.3274	.1544	-.0588
20	1.0636	.4987	-.2491	.1278	-.0594	20	1.4579	.6652	-.3630	.1742	-.0864
25	1.0241	.5992	-.2513	.1227	-.0722	25	1.4025	.6925	-.4745	.1722	-.1054

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TABLE 5.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 5 MODEL

 $\frac{k}{c} = 0.06$      $\frac{C_f}{c} = \text{NONE}$ 

$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_H$	$C_I$	$C_n$
$M = 0.40$											
-10	-0.5664					-10	-0.9107	.1824	.1735	-0.1116	-0.0280
-7	-0.4377					-7	-0.6750	.1086	.1092	-0.0824	-0.0169
-5	-0.3175					-5	-0.5062	.0690	.0746	-0.0619	-0.0115
-3	-0.2060					-3	-0.3251	.0458	.0472	-0.0405	-0.0069
-2	-0.1459					-2	-0.2333	.0378	.0335	-0.0294	-0.0054
-1	-0.0858					-1	-0.1290	.0299	.0159	-0.0161	-0.0039
0	-0.0172					0	-0.0223	.0354	.0088	-0.0043	-0.0027
1	.0472					1	.0819	.0335	.0005	.0080	-0.0024
2	.1159					2	.1762	.0385	-.0082	.0186	-0.0025
3	.1802					3	.2730	.0458	-.0214	.0304	-0.0032
5	.3004					5	.4467	.0720	.0488	.0508	-0.0057
7	.4334					7	.6328	.1117	-.0884	.0730	-0.0098
10	.6179					10	.8958	.1941	-.1641	.1031	-0.0192
15	.6995					15	1.1812	.3576	-.2135	.1407	-0.0386
20	.7381										
25	.6995										
$M = 0.55$											
-10	-0.9107	.1824	.1735	-0.1116	-0.0280	-10	-0.8492	.1732	.1605	-0.1044	-0.0261
-7	-0.6750	.1086	.1092	-0.0824	-0.0169	-7	-0.6351	.1047	.1016	-0.0776	-0.0161
-5	-0.5062	.0690	.0746	-0.0619	-0.0115	-5	-0.4805	.0708	.0742	-0.0587	-0.0116
-3	-0.3251	.0458	.0472	-0.0405	-0.0069	-3	-0.3069	.0485	.0505	-0.0386	-0.0074
-2	-0.2333	.0378	.0335	-0.0294	-0.0054	-2	-0.2141	.0403	.0358	-0.0270	-0.0058
-1	-0.1290	.0299	.0159	-0.0161	-0.0039	-1	-0.1189	.0381	.0216	-0.0155	-0.0047
0	-0.0223	.0354	.0088	-0.0043	-0.0027	0	-0.0071	.0387	.0084	-0.0035	-0.0034
1	.0819	.0335	.0005	.0080	-0.0024	1	.0880	.0409	-.0058	.0079	-0.0034
2	.1762	.0385	-.0082	.0186	-0.0025	2	.1879	.0467	-.0195	.0187	-0.0038
3	.2730	.0458	-.0214	.0304	-0.0032	3	.2759	.0544	-.0321	.0298	-0.0043
5	.4467	.0720	.0488	.0508	-0.0057	5	.4577	.0778	.0600	.0494	-0.0066
7	.6328	.1117	-.0884	.0730	-0.0098	7	.6137	.1158	-.0981	.0702	-0.0107
10	.8958	.1941	-.1641	.1031	-0.0192	10	.8302	.1842	-.1426	.0958	-0.0193
15	1.1812	.3576	-.2135	.1407	-0.0386	15	1.1798	.3544	-.2205	.1372	-0.0394
$M = 1.00$											
-10	-0.8492	.1732	.1605	-0.1044	-0.0261	-10	-0.8492	.1732	.1605	-0.1044	-0.0261
-7	-0.6351	.1047	.1016	-0.0776	-0.0161	-7	-0.6351	.1047	.1016	-0.0776	-0.0161
-5	-0.4805	.0708	.0742	-0.0587	-0.0116	-5	-0.4805	.0708	.0742	-0.0587	-0.0116
-3	-0.3069	.0485	.0505	-0.0386	-0.0074	-3	-0.3069	.0485	.0505	-0.0386	-0.0074
-2	-0.2141	.0403	.0358	-0.0270	-0.0058	-2	-0.2141	.0403	.0358	-0.0270	-0.0058
-1	-0.1189	.0381	.0216	-0.0155	-0.0047	-1	-0.1189	.0381	.0216	-0.0155	-0.0047
0	-0.0071	.0387	.0084	-0.0035	-0.0034	0	-0.0071	.0387	.0084	-0.0035	-0.0034
1	.0880	.0409	-.0058	.0079	-0.0034	1	.0880	.0409	-.0058	.0079	-0.0034
2	.1879	.0467	-.0195	.0187	-0.0038	2	.1879	.0467	-.0195	.0187	-0.0038
3	.2759	.0544	-.0321	.0298	-0.0043	3	.2759	.0544	-.0321	.0298	-0.0043
5	.4577	.0778	.0600	.0494	-0.0066	5	.4577	.0778	.0600	.0494	-0.0066
7	.6137	.1158	-.0981	.0702	-0.0107	7	.5837	.1110	-.0867	.0666	-0.0106
10	.7912	.1783	-.1367	.0910	-0.0179	10	.7912	.1783	-.1367	.0910	-0.0179
15	1.1218	.3397	-.2224	.1306	-0.0370	15	1.1218	.3397	-.2224	.1306	-0.0370
$M = 1.05$											
-10	-0.8492	.1732	.1605	-0.1044	-0.0261	-10	-0.8492	.1732	.1605	-0.1044	-0.0261
-7	-0.6351	.1047	.1016	-0.0776	-0.0161	-7	-0.6351	.1047	.1016	-0.0776	-0.0161
-5	-0.4537	.0678	.0721	-0.0558	-0.0114	-5	-0.4537	.0678	.0721	-0.0558	-0.0114
-3	-0.2896	.0465	.0469	-0.0361	-0.0073	-3	-0.2896	.0465	.0469	-0.0361	-0.0073
-2	-0.1984	.0398	.0333	-0.0255	-0.0060	-2	-0.1984	.0398	.0333	-0.0255	-0.0060
-1	-0.1094	.0381	.0217	-0.0151	-0.0047	-1	-0.1094	.0381	.0217	-0.0151	-0.0047
0	-0.0091	.0381	.0081	-0.0038	-0.0036	0	-0.0091	.0381	.0081	-0.0038	-0.0036
1	.0889	.0409	-.0071	.0080	-0.0036	1	.0889	.0409	-.0071	.0080	-0.0036
2	.1756	.0454	-.0187	.0179	-0.0037	2	.1756	.0454	-.0187	.0179	-0.0037
3	.2645	.0532	-.0313	.0283	-0.0044	3	.2645	.0532	-.0313	.0283	-0.0044
5	.4195	.0746	-.0570	.0467	-0.0065	5	.4195	.0746	-.0570	.0467	-0.0065
7	.5837	.1110	-.0867	.0666	-0.0106	7	.5837	.1110	-.0867	.0666	-0.0106
10	.7912	.1783	-.1367	.0910	-0.0179	10	.7912	.1783	-.1367	.0910	-0.0179
15	1.1218	.3397	-.2224	.1306	-0.0370	15	1.1218	.3397	-.2224	.1306	-0.0370
$M = 1.10$											
-10	-0.7783	.1622	.1436	-0.0956	-0.0247	-10	-0.7783	.1622	.1436	-0.0956	-0.0247
-7	-0.5854	.0959	.0975	-0.0717	-0.0158	-7	-0.5854	.0959	.0975	-0.0717	-0.0158
-5	-0.4407	.0636	.0708	-0.0543	-0.0117	-5	-0.4407	.0636	.0708	-0.0543	-0.0117
-3	-0.2784	.0426	.0451	-0.0347	-0.0074	-3	-0.2784	.0426	.0451	-0.0347	-0.0074
-2	-0.1885	.0366	.0320	-0.0245	-0.0059	-2	-0.1885	.0366	.0320	-0.0245	-0.0059
-1	-0.1009	.0356	.0209	-0.0138	-0.0048	-1	-0.1009	.0356	.0209	-0.0138	-0.0048
0	-0.0088	.0362	.0078	-0.0040	-0.0038	0	-0.0088	.0362	.0078	-0.0040	-0.0038
1	.0833	.0366	-.0053	.0070	-0.0040	1	.0833	.0366	-.0053	.0070	-0.0040
2	.1732	.0410	-.0179	.0172	-0.0044	2	.1732	.0410	-.0179	.0172	-0.0044
3	.2543	.0486	-.0296	.0273	-0.0050	3	.2543	.0486	-.0296	.0273	-0.0050
5	.4056	.0700	-.0543	.0451	-0.0071	5	.4056	.0700	-.0543	.0451	-0.0071
7	.5656	.1046	-.0854	.0643	-0.0112	7	.5656	.1046	-.0854	.0643	-0.0112
10	.7673	.1714	-.1329	.0881	-0.0183	10	.7673	.1714	-.1329	.0881	-0.0183
15	1.0655	.3246	-.2071	.1256	-0.0366	15	1.0655	.3246	-.2071	.1256	-0.0366

TABLE 5.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 5 MODEL - Continued

$$\frac{c_t}{c} = 0.06 \quad \frac{c_f}{c} = 0.20$$

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$
$M = 0.40$						$M = 0.95$					
-10	-.3101					-10	-.7009	.1559	.0342	-.0825	-.0225
-7	-.1240					-7	-.4904	.0962	-.0115	-.0558	-.0134
-5	.0043					-5	-.3220	.0718	-.0430	-.0361	-.0096
-3	.1433					-3	-.1684	.0476	-.0567	-.0168	-.0052
-2	.2181					-2	-.0545	.0445	-.0813	-.0031	-.0040
-1	.2823					-1	.0842	.0445	-.1169	.0127	-.0040
0	.3425					0	.2452	.0530	-.1572	.0313	-.0047
1	.4045					1	.3886	.0670	-.1874	.0474	-.0058
2	.4534					2	.5077	.0780	-.2038	.0618	-.0071
3	.5132					3	.6241	.0962	-.2202	.0757	-.0090
5	.6331					5	.7891	.1327	-.2416	.0945	-.0132
7	.7635					7	.9461	.1851	-.2662	.1130	-.0190
10	.8447					10	1.1318	.2759	-.2969	.1371	-.0295
15	.8640										
20	.8383										
25	.7934										
$M = 0.60$						$M = 1.00$					
-10	-.3585	.0798	-.0665	-.0431	-.0104	-10	-.6785	.1476	.0362	-.0790	-.0215
-7	-.1535	.0248	-.1271	-.0160	-.0028	-7	-.4697	.0951	-.0129	-.0528	-.0138
-5	.0022	.0125	-.1285	.0024	-.0014	-5	-.3060	.0718	-.0451	-.0334	-.0098
-3	.1513	.0145	-.1271	.0195	-.0011	-3	-.1423	.0590	-.0714	-.0138	-.0067
-2	.2214	.0162	-.1223	.0271	-.0008	-2	-.0427	.0554	-.0890	-.0016	-.0059
-1	.2850	.0204	-.1173	.0341	-.0010	-1	.0641	.0250	-.1081	.0104	-.0052
0	.3639	.0259	-.1189	.0433	-.0012	0	.1756	.0595	-.1270	.0230	-.0052
1	.4758	.0340	-.1271	.0577	-.0013	1	.3013	.0648	-.1516	.0362	-.0057
2	.5393	.0421	-.1173	.0654	-.0015	2	.4318	.0764	-.1821	.0537	-.0070
3	.6227	.0528	-.1150	.0741	-.0023	3	.5623	.0922	-.2083	.0684	-.0088
5	.7586	.0787	-.1081	.0900	-.0048	5	.7354	.1278	-.2330	.0887	-.0129
7	.8814	.1201	-.1077	.1054	-.0101	7	.8944	.1797	-.2550	.1071	-.0187
10	.9365	.1903	-.1562	.1171	-.0191	10	1.0771	.2655	-.2835	.1313	-.0285
15	.9230	.2944	-.1867	.1213	-.0338						
20	.8901	.3806	-.1886	.1141	-.0454						
$M = 0.80$						$M = 1.05$					
-10	-.4898	.0944	-.0735	-.0593	-.0137	-10	-.6440	.1427	.0335	-.0751	-.0212
-7	-.3383	.0459	-.1080	-.0386	-.0066	-7	-.4460	.0912	-.0106	-.0508	-.0138
-5	.0151	.0238	-.1253	-.0164	-.0026	-5	-.2981	.0700	-.0400	-.0331	-.0097
-3	.0677	.0181	-.1314	.0090	-.0011	-3	-.1320	.0577	-.0684	-.0132	-.0069
-2	.1721	.0210	-.1340	.0207	-.0010	-2	-.0410	.0537	-.0848	-.0022	-.0059
-1	.2751	.0257	-.1340	.0329	-.0012	-1	.0614	.0520	-.1012	.0095	-.0058
0	.3765	.0310	-.1350	.0447	-.0019	0	.1661	.0560	-.1206	.0219	-.0058
1	.4810	.0402	-.1422	.0571	-.0025	1	.2731	.0631	-.1404	.0345	-.0065
2	.6089	.0499	-.1454	.0731	-.0038	2	.3800	.0717	-.1621	.0471	-.0075
3	.7060	.0632	-.1416	.0848	-.0048	3	.5097	.1432	-.1888	.0619	-.0091
5	.8649	.0980	-.1497	.1057	-.0085	5	.6895	.1220	-.2235	.0833	-.0130
7	.9737	.1440	-.1568	.1214	-.0142	7	.8510	.1707	-.2436	.1016	-.0185
10	1.0737	.2177	-.1584	.1354	-.0236	10	1.0285	.2535	-.2746	.1250	-.0281
15	.9237	.3009	-.2017	.1171	-.0356						
$M = 0.90$						$M = 1.10$					
-10	-.6772	.1397	-.0051	-.0835	-.0206	-10	-.6194	.1389	.0358	-.0727	-.0213
-7	-.5051	.0812	-.0233	-.0596	-.0119	-7	-.4334	.0899	-.0102	-.0510	-.0135
-5	-.3495	.0524	-.0519	-.0384	-.0075	-5	-.2911	.0673	-.0361	-.0321	-.0100
-3	-.0983	.0343	-.1093	-.0072	-.0032	-3	-.1226	.0595	-.0658	-.0121	-.0071
-2	.0464	.0336	-.1335	-.0353	-.0025	-2	-.0350	.0517	-.0811	-.0021	-.0063
-1	.1966	.0375	-.1613	.0260	-.0025	-1	.0613	.0511	-.0981	.0091	-.0059
0	.3195	.0456	-.1673	.0398	-.0029	0	.1576	.0546	-.1155	.0202	-.0060
1	.4367	.0557	-.1764	.0533	-.0041	1	.2670	.0608	-.1345	.0329	-.0065
2	.5516	.0692	-.1824	.0668	-.0057	2	.3655	.0705	-.1544	.0446	-.0074
3	.6608	.0860	-.1915	.0795	-.0074	3	.4772	.0850	-.1767	.0574	-.0088
5	.8328	.1256	-.2096	.1007	-.0121	5	.6610	.1173	-.2140	.0786	-.0127
7	.9966	.1834	-.2398	.1206	-.0187	7	.8164	.1442	-.2363	.0971	-.0178
10	1.1796	.2759	-.2670	.1447	-.0295	10	.9937	.2449	-.2634	.1199	-.0272
15	1.3762	.2135	-.2730	.1670	-.0493						
20	1.1195	.2384	-.2609	.1352	-.0554						

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TABLE 5 -- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 5 MODEL - Continued

$$\frac{L}{c} = 0.06 \quad \frac{C_L}{c} = 0.30$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_z$	$C_n$	$C_L$	$C_D$	$C_M$	$C_z$	$C_n$
$K = 0.40$										
$M = 0.40$										
-10	-0.2299					-0.6020	0.1420	-0.0151	-0.0722	-0.0206
-7	-0.0434					-0.3815	0.0927	-0.0636	-0.0440	-0.0128
-5	0.0954					-0.2180	0.0701	-0.0951	-0.0233	-0.0087
-3	0.2342					-0.0198	0.0555	-0.1293	0.0000	-0.0061
-2	0.3036					-0.1338	0.0578	-0.1655	0.0176	-0.0062
-1	0.3774					-0.2899	0.0639	-0.1929	0.0349	-0.0062
0	0.4294					0.4385	0.0725	-0.2093	0.0493	-0.0074
1	0.4858					1.5351	0.0853	-0.2176	0.0637	-0.0088
2	0.5422					2.6243	0.1005	-0.2247	0.0741	-0.0103
3	0.6029					3.7259	0.1207	-0.2340	0.0854	-0.0125
5	0.7243					5.8721	0.1614	-0.2493	0.1027	-0.0168
7	0.8458					7.1.0132	0.2163	-0.2669	0.1188	-0.0232
10	0.9152					10.1.1817	0.3065	-0.2871	0.1419	-0.0338
15	0.9055									
20	0.8111									
25	0.8284									
$M = 0.60$										
$M = 1.00$										
-10	-0.2850	0.0702	-0.0776	-0.0349	-0.0085	-0.5723	0.1373	-0.0123	-0.0692	-0.0198
-7	-0.0514	0.0259	-0.1256	-0.0060	-0.0014	-0.3633	0.0893	-0.0583	-0.0420	-0.0128
-5	0.1030	0.0162	-0.1256	0.0134	-0.0010	-0.2019	0.0706	-0.0903	-0.0224	-0.0089
-3	0.2784	0.0195	-0.1280	0.0332	-0.0010	-0.0309	0.0625	-0.1195	-0.0012	-0.0071
-2	0.3596	0.0237	-0.1295	0.0426	-0.0012	-0.0760	0.0613	-0.1376	0.0106	-0.0066
-1	0.4429	0.0303	-0.1295	0.0515	-0.0017	-0.1995	0.0642	-0.1586	0.0242	-0.0065
0	0.5152	0.0344	-0.1271	0.0607	-0.0025	0.3420	0.0718	-0.1875	0.0415	-0.0072
1	0.5635	0.0410	-0.1183	0.0668	-0.0029	1.4916	0.0835	-0.2085	0.0572	-0.0087
2	0.6405	0.0506	-0.1150	0.0754	-0.0038	2.5747	0.0963	-0.2190	0.0685	-0.0102
3	0.7104	0.0625	-0.1101	0.0834	-0.0049	3.6744	0.1168	-0.2269	0.0795	-0.0123
5	0.8551	0.0895	-0.1028	0.1007	-0.0082	5.8240	0.1547	-0.2400	0.0968	-0.0165
7	0.9406	0.1348	-0.1077	0.1145	-0.0136	7.9546	0.2073	-0.2553	0.1130	-0.0224
10	0.9757	0.2081	-0.1450	0.1222	-0.0234	10.1.280	0.2960	-0.2826	0.1360	-0.0328
15	0.9825	0.3223	-0.1794	0.1286	-0.0394					
20	0.9077	0.3925	-0.1804	0.1179	-0.0489					
25	0.8989	0.4907	-0.1984	0.1137	-0.0610					
$M = 0.80$										
$M = 1.05$										
-10	-0.3824	0.0833	-0.1103	-0.0446	-0.0116	-0.5465	0.1343	-0.0126	-0.0652	-0.0192
-7	-0.1956	0.0427	-0.1406	-0.0210	-0.0056	-0.3229	0.0879	-0.0539	-0.0409	-0.0126
-5	-0.0059	0.0275	-0.1506	0.0009	0.0025	-0.2004	0.0689	-0.0816	-0.0223	-0.0095
-3	0.1839	0.0290	-0.1461	0.0223	-0.0023	-0.0296	0.0614	-0.1118	-0.0022	-0.0071
-2	0.2780	0.0318	-0.1451	0.0331	-0.0025	-0.0766	0.0587	-0.1254	0.0097	-0.0070
-1	0.3663	0.0377	-0.1419	0.0436	-0.0032	-0.1708	0.0633	-0.1435	0.0210	-0.0067
0	0.4619	0.0449	-0.1413	0.0546	-0.0040	0.2892	0.0689	-0.1647	0.0347	-0.0073
1	0.5545	0.0543	-0.1431	0.0657	-0.0049	1.4190	0.0789	-0.1899	0.0497	-0.0086
2	0.6725	0.0672	-0.1454	0.0805	-0.0062	2.5283	0.0923	-0.2050	0.0630	-0.0098
3	0.7707	0.0796	-0.1397	0.0923	-0.0075	3.6239	0.1092	-0.2176	0.0741	-0.0115
5	0.9149	0.1172	-0.1448	0.1105	-0.0112	5.7788	0.1462	-0.2317	0.0913	-0.0155
7	1.0179	0.1664	-0.1503	0.1254	-0.0169	7.9545	0.1959	-0.2463	0.1072	-0.0213
10	1.1179	0.2481	-0.1568	0.1385	-0.0265	10.1.0770	0.2822	-0.2730	0.1296	-0.0311
15	0.9414	0.3183	-0.1900	0.1171	-0.0374					
20	0.9649	0.4296	-0.2063	0.1185	-0.0504					
$M = 0.90$										
$M = 1.10$										
-10	-0.5639	0.1330	-0.0595	-0.0703	-0.0189	-0.5279	0.1281	-0.0106	-0.0629	-0.0185
-7	-0.4345	0.0826	-0.0640	-0.0504	-0.0117	-0.3417	0.0845	-0.0494	-0.0396	-0.0124
-5	-0.2225	0.0537	-0.1048	-0.0239	-0.0069	-0.2037	0.0657	-0.0760	-0.0223	-0.0093
-3	0.0642	0.0437	-0.1613	0.0093	-0.0041	-0.0350	0.0592	-0.1051	-0.0021	-0.0074
-2	0.2075	0.0470	-0.1782	0.0260	-0.0043	-0.0591	0.0582	-0.1202	0.0085	-0.0068
-1	0.3372	0.0571	-0.1945	0.0403	-0.0053	-0.1533	0.0609	-0.1366	0.0202	-0.0067
0	0.4533	0.0605	-0.1885	0.0517	-0.0057	0.2628	0.0663	-0.1526	0.0315	-0.0076
1	0.5488	0.0759	-0.2005	0.0650	-0.0072	1.3811	0.1034	-0.1754	0.0453	-0.0084
2	0.6444	0.0886	-0.1975	0.0769	-0.0088	2.4928	0.0888	-0.1948	0.0585	-0.0098
3	0.7485	0.1095	-0.2066	0.0901	-0.0116	3.5957	0.1050	-0.2093	0.0708	-0.0119
5	0.9205	0.1859	-0.2277	0.1100	-0.0170	5.7425	0.1416	-0.2234	0.0872	-0.0153
7	1.0567	0.2082	-0.2398	0.1267	-0.0232	7.8805	0.1901	-0.2384	0.1029	-0.0207
10	1.2260	0.2989	-0.2609	0.1511	-0.0170	10.1.0338	0.2730	-0.2636	0.1238	-0.0300

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TABLE 5 - THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 5 MODEL - Concluded

$$\frac{t}{c} = 0.06 \quad \frac{c_f}{c} = 0.40$$

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$
$M = 0.40$											
-10	.2035					-10	.5252	.1371	-.0405	-.0592	-.0201
-7	-.0086					-7	.3208	.0935	-.0872	-.0334	-.0135
-5	.1264					-5	.1387	.0768	-.1268	-.0106	-.0093
-3	.2699					-3	.1189	.0706	-.1825	.0195	-.0078
-2	.3556					-2	.2676	.0746	-.2072	.0363	-.0084
-1	.4006					-1	.3828	.0819	-.2208	.0500	-.0092
0	.4627					0	.4905	.0935	-.2263	.0623	-.0104
1	.5377					1	.5834	.1073	-.2307	.0496	-.0118
2	.5869					2	.6788	.1242	-.2389	.0852	-.0135
3	.6341					3	.7754	.1453	-.2455	.0953	-.0154
5	.7476					5	.9315	.1919	-.2631	.1121	-.0202
7	.8483					7	1.0628	.2541	-.2828	.1397	-.0271
10	.8804					10	1.2338	.3497	-.3047	.1520	-.0383
15	.8847					15	1.4022	.4837	-.2992	.1664	-.0565
20	.8854										
25	.7840										
$M = 0.60$											
-10	-.2458	.0632	-.0718	-.0290	-.0082	-10	.4975	.1319	-.0383	-.0563	-.0191
-7	.0025	.0226	-.1165	.0019	-.0017	-7	.3016	.0911	-.0830	-.0114	-.0130
-5	.1679	.0189	-.1179	.0213	-.0017	-5	.1354	.0765	-.1190	-.0108	-.0095
-3	.3958	.0248	-.1204	.0403	-.0015	-3	.0724	.0712	-.1618	.0141	-.0082
-2	.4159	.0292	-.1146	.0507	-.0018	-2	.1971	.0730	-.1828	.0279	-.0080
-1	.4872	.0362	-.1204	.0595	-.0021	-1	.3218	.0791	-.2058	.0424	-.0085
0	.5509	.0432	-.1155	.0663	-.0025	0	.4465	.0885	-.2170	.0556	-.0094
1	.6211	.0525	-.1146	.0748	-.0033	1	.5474	.1016	-.2238	.0685	-.0108
2	.6935	.0621	-.1068	.0831	-.0047	2	.6317	.1174	-.2301	.0786	-.0127
3	.7571	.0751	-.1082	.0932	-.0059	3	.7148	.1357	-.2353	.0885	-.0146
5	.8779	.1058	-.1038	.1059	-.0095	5	.8573	.1793	-.2501	.1040	-.0191
7	.9503	.1473	-.1082	.1189	-.0168	7	1.0045	.2365	-.2711	.1250	-.0257
10	.9656	.2278	-.1520	.1240	-.0129	10	1.1613	.3352	-.3005	.1448	-.0368
15	.9020	.3226	-.1762	.1198	-.0196						
20	.9020	.4198	-.1913	.1176	-.0130						
25	.8800	.5094	-.1937	.1134	-.0158						
$M = 0.80$											
-10	-.2335	.0774	-.0970	-.0331	-.0106	-10	.4827	.1271	-.0342	-.0550	-.0183
-7	-.0564	.0431	-.1451	.0085	-.0043	-7	.2892	.0882	-.0776	-.0303	-.0125
-5	.0375	.0333	-.1523	.0126	-.0033	-5	.1298	.0733	-.1116	-.0108	-.0094
-3	.2656	.0387	-.1504	.0329	-.0036	-3	.0524	.0695	-.1501	.0113	-.0082
-2	.3465	.0449	-.1513	.0430	-.0039	-2	.1548	.0706	-.1682	.0234	-.0085
-1	.4355	.0531	-.1530	.0531	-.0048	-1	.2687	.0767	-.1854	.0367	-.0090
0	.5164	.0615	-.1507	.0629	-.0055	0	.3939	.0846	-.2030	.0504	-.0093
1	.6047	.0702	-.1491	.0751	-.0068	1	.4964	.0965	-.2130	.0632	-.0107
2	.6754	.0778	-.1328	.0826	-.0077	2	.5829	.1111	-.2181	.0732	-.0125
3	.8122	.0959	-.1393	.0989	-.0095	3	.6740	.1299	-.2267	.0836	-.0143
5	.9211	.1299	-.1425	.1151	-.0137	5	.8106	.1719	-.2433	.0995	-.0186
7	1.0182	.1791	-.1497	.1289	-.0198	7	.9541	.2279	-.2599	.1198	-.0252
10	1.0917	.1274	-.1572	.1423	-.0292	10	1.1066	.3214	-.2901	.1388	-.0354
15	.9593	.1689	-.1940	.1234	-.0411						
20	.9446	.4444	-.2038	.1211	-.0534						
25	.9299	.5456	-.2168	.1194	-.0633						
$M = 0.90$											
-10	-.4934	.1200	-.0642	-.0589	-.0176	-10	.4700	.1209	-.0310	-.0534	-.0178
-7	-.3216	.0778	-.0901	-.0346	-.0108	-7	.2859	.0854	-.0727	-.0296	-.0120
-5	-.0924	.0592	-.1376	-.0071	-.0066	-5	.1291	.0722	-.1049	-.0119	-.0092
-3	.1810	.0599	-.1860	.0241	-.0058	-3	.0449	.0670	-.1405	.0109	-.0082
-2	.3093	.0647	-.1918	.0380	-.0067	-2	.1490	.0692	-.1599	.0221	-.0085
-1	.4075	.0742	-.2004	.0498	-.0079	-1	.2585	.0746	-.1759	.0343	-.0084
0	.4960	.0838	-.2033	.0622	-.0087	0	.3571	.0824	-.1919	.0460	-.0094
1	.5637	.0871	-.1889	.0698	-.0093	1	.4776	.0940	-.2065	.0604	-.0110
2	.6666	.1047	-.1918	.0824	-.0112	2	.5587	.1075	-.2137	.0700	-.0124
3	.7577	.1239	-.2033	.0933	-.0130	3	.6485	.1258	-.2210	.0800	-.0143
5	.8931	.1623	-.2033	.1097	-.0176	5	.7866	.1665	-.2341	.0955	-.0183
7	1.0181	.2222	-.2292	.1294	-.0239	7	.9246	.2219	-.3451	.1153	-.0249
10	1.1639	.3067	-.2379	.1471	-.0348	10	1.0648	.3124	-.2743	.1323	-.0351
15	1.0511	.3803	-.2223	.1299	-.0442						
$M = 1.10$											
-10	-.4700	.1209	-.0642	-.0589	-.0176						
-7	.2859	.0854	-.0727	-.0296	-.0120						
-5	.1291	.0722	-.1049	-.0119	-.0092						
-3	.0449	.0670	-.1405	-.0109	-.0082						
-2	.1490	.0692	-.1599	-.0221	-.0085						
-1	.2585	.0746	-.1759	-.0343	-.0084						
0	.3571	.0824	-.1919	-.0460	-.0094						
1	.4776	.0940	-.2065	-.0604	-.0110						
2	.5587	.1075	-.2137	-.0700	-.0124						
3	.6485	.1258	-.2210	-.0800	-.0143						
5	.7866	.1665	-.2341	-.0955	-.0183						
7	.9246	.2219	-.3451	.1153	-.0249						
10	1.0648	.3124	-.2743	.1323	-.0351						

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TABLE 6.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 6 MODEL

 $\frac{t}{c} = 0.06$        $\frac{C_f}{t} = \text{NONE}$ 

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_i$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_i$	$c_n$
$M = 0.40$											
-10	-.6162					-10	.8915	.1751	.1455	-.1119	-.0264
-7	-.4843					-7	-.6596	.1005	.1016	-.0829	-.0149
-5	-.3542					-5	-.4932	.0643	.0676	-.0617	-.0096
-3	-.2169					-3	-.3056	.0419	.0373	-.0385	-.0056
-2	-.1500					-2	-.1897	.0332	.0233	-.0239	-.0035
-1	-.0813					-1	-.0822	.0280	.0093	-.0119	-.0023
0	-.0108					0	.0211	.0290	.0000	.0014	-.0020
1	.0813					1	.1286	.0332	-.0131	.0136	-.0015
2	.1536					2	.2297	.0399	-.0233	.0252	-.0016
3	.2440					3	.3309	.0513	-.0396	.0375	-.0024
5	.3650					5	.4995	.0798	-.0699	.0580	-.0050
7	.5096					7	.7060	.1280	-.1165	.0818	-.0100
10	.6813					10	.9463	.2145	-.1818	.1108	-.0199
15	.8187										
20	1.0012										
25	.7301										
$M = 0.50$											
-10	-.7031	.1239	.0610	-.0901	-.0177	-10	-.8383	.1659	.1564	-.1046	-.0251
-7	-.5906	.0585	.0108	-.0722	-.0086	-7	-.6181	.0989	.0929	-.0771	-.0146
-5	-.4312	.0272	-.0008	-.0522	-.0039	-5	-.4585	.0640	.0648	-.0579	-.0096
-3	-.2681	.0111	-.0208	-.0322	-.0015	-3	-.2787	.0427	.0402	-.0360	-.0056
-2	-.1744	.0073	-.0112	-.0218	-.0004	-2	-.1838	.0368	.0268	-.0239	-.0045
-1	-.0937	.0051	-.0033	-.0115	-.0001	-1	-.0869	.0342	.0121	-.0124	-.0031
0	-.0056	.0032	-.0000	-.0006	.0003	0	.0182	.0333	.0000	.0007	-.0027
1	.0787	.0064	.0008	.0091	.0005	1	.1192	.0378	-.0134	.0118	-.0027
2	.1669	.0101	.0021	.0188	.0006	2	.2161	.0442	-.0290	.0235	-.0027
3	.2587	.0171	.0021	.0291	.0001	3	.3131	.0551	-.0425	.0346	-.0035
5	.4219	.0088	.0021	.0479	-.0002	5	.4787	.0815	-.0715	.0546	-.0059
7	.5981	.0673	.0041	.0677	-.0028	7	.6403	.1222	-.1028	.0742	-.0098
10	.7537	.1378	-.0436	.0880	-.0102	10	.8726	.1992	-.1622	.1013	-.0185
15	.7856	.2417	-.1000	.0995	-.0238						
20	.7725	.3217	-.1120	.0983	-.0349						
25	.8044	.4177	-.1203	.0998	-.0470						
$M = 0.80$											
-10	-.8201	.1350	.0567	-.1060	-.0201	-10	-.7909	.1583	.1449	-.0988	-.0239
-7	-.7026	.0765	.0346	-.0882	-.0114	-7	-.5874	.0934	.0901	-.0737	-.0144
-5	-.5576	.0381	.0139	-.0884	-.0059	-5	-.4459	.0619	.0652	-.0558	-.0098
-3	-.3326	.0120	-.0083	-.0405	-.0019	-3	-.2695	.0405	.0395	-.0345	-.0056
-2	-.2075	.0068	-.0061	-.0255	-.0010	-2	-.1745	.0353	.0249	-.0229	-.0044
-1	-.1100	.0049	-.0025	-.0142	-.0004	-1	-.0756	.0333	.0107	-.0113	-.0029
0	-.0075	.0059	-.0000	-.0020	-.0001	0	.0174	.0338	-.0004	.0000	-.0028
1	.1100	.0089	.0011	.0121	.0001	1	.1144	.0386	-.0163	.0110	-.0026
2	.2150	.0129	.0061	.0235	.0001	2	.2074	.0448	-.0292	.0220	-.0028
3	.3101	.0196	.0144	.0344	-.0002	3	.2947	.0539	-.0407	.0323	-.0034
5	.5101	.0451	.0028	.0587	.0018	5	.4536	.0791	-.0665	.0511	-.0056
7	.6926	.0889	-.0199	.0817	-.0062	7	.6145	.1196	-.0986	.0706	-.0095
10	.8251	.1534	-.0429	.1003	-.0140	10	.8200	.1916	-.1509	.0941	-.0171
$M = 0.90$											
-10	-.8895	.1661	.1416	-.1107	-.0246	-10	-.7660	.1544	.1402	-.0959	-.0233
-7	-.6908	.0971	.0976	-.0850	-.0139	-7	-.5741	.0912	.0928	-.0709	-.0142
-5	-.5143	.0564	.0512	-.0625	-.0078	-5	-.4249	.0614	.0639	-.0534	-.0127
-3	-.3046	.0283	.0147	-.0368	-.0031	-3	-.2572	.0399	.0371	-.0326	-.0058
-2	-.2008	.0212	.0073	-.0239	-.0020	-2	-.1640	.0349	.0235	-.0217	-.0042
-1	-.1170	.0158	.0098	-.0136	-.0010	-1	-.0708	.0339	.0124	-.0112	-.0030
0	-.0066	.0141	.0039	-.0000	-.0002	0	.0168	.0339	-.0008	.0000	-.0026
1	.0993	.0184	-.0024	.0125	-.0001	1	.1118	.0376	-.0157	.0109	-.0024
2	.1920	.0238	-.0024	.0243	-.0003	2	.2032	.0440	-.0280	.0217	-.0026
3	.2980	.0348	-.0147	.0393	-.0009	3	.2870	.0531	-.0412	.0317	-.0033
5	.4679	.0608	-.0391	.0596	-.0032	5	.4455	.0788	-.0680	.0498	-.0054
7	.6555	.1064	-.0781	.0821	-.0080	7	.6001	.1178	-.0989	.0688	-.0091
10	.8586	.1786	-.1099	.1057	-.0172	10	.8052	.1866	-.1402	.0920	-.0173
15	1.1545	.3289	-.1587	.1336	-.0349						
20	.9800	.3897	-.1782	.1196	-.0212						
$M = 1.00$											
-10	-.8383	.1659	.1564	-.1046	-.0251	-10	-.7909	.1583	.1449	-.0988	-.0239
-7	-.6181	.0989	.0929	-.0771	-.0146	-7	-.5874	.0934	.0901	-.0737	-.0144
-5	-.4585	.0640	.0648	-.0579	-.0096	-5	-.4459	.0619	.0652	-.0558	-.0098
-3	-.2787	.0427	.0402	-.0360	-.0056	-3	-.2695	.0405	.0395	-.0345	-.0056
-2	-.1838	.0368	.0268	-.0239	-.0045	-2	-.1745	.0353	.0249	-.0229	-.0044
-1	-.0869	.0342	.0121	-.0124	-.0031	-1	-.0756	.0333	.0107	-.0113	-.0029
0	.0182	.0333	.0000	-.0007	-.0027	0	.0174	.0338	-.0004	.0000	-.0028
1	.1192	.0378	-.0134	.0118	-.0027	1	.1144	.0386	-.0163	.0110	-.0026
2	.2161	.0442	-.0290	.0235	-.0027	2	.2074	.0448	-.0292	.0220	-.0028
3	.3131	.0551	-.0425	.0346	-.0035	3	.2947	.0539	-.0407	.0323	-.0034
5	.4787	.0815	-.0715	.0546	-.0059	5	.4536	.0791	-.0665	.0511	-.0056
7	.6403	.1222	-.1028	.0742	-.0098	7	.6145	.1196	-.0986	.0706	-.0095
10	.8726	.1992	-.1622	.1013	-.0185	10	.8200	.1916	-.1509	.0941	-.0171
$M = 1.10$											
-10	-.7660	.1544	.1402	-.0959	-.0233	-10	-.7660	.1544	.1402	-.0959	-.0233
-7	-.5741	.0912	.0928	-.0709	-.0142	-7	-.5741	.0912	.0928	-.0709	-.0142
-5	-.4249	.0614	.0639	-.0534	-.0127	-5	-.4249	.0614	.0639	-.0534	-.0127
-3	-.2572	.0399	.0371	-.0326	-.0058	-3	-.2572	.0399	.0371	-.0326	-.0058
-2	-.1640	.0349	.0235	-.0217	-.0042	-2	-.1640	.0349	.0235	-.0217	-.0042
-1	-.0708	.0339	.0124	-.0112	-.0030	-1	-.0708	.0339	.0124	-.0112	-.0030
0	.0168	.0339	-.0008	.0000	-.0026	0	.0168	.0339	-.0008	.0000	-.0026
1	.1118	.0376	-.0157	.0109	-.0024	1	.1118	.0376	-.0157	.0109	-.0024
2	.2032	.0440	-.0280	.0217	-.0026	2	.2032	.0440	-.0280	.0217	-.0026
3	.2870	.0531	-.0412	.0317	-.0033	3	.2870	.0531	-.0412	.0317	-.0033
5	.4455	.0788	-.0680	.0498	-.0054	5	.4455	.0788	-.0680	.0498	-.0054
7	.6001	.1178	-.0989	.0688	-.0091	7	.6001	.1178	-.0989	.0688	-.0091
10	.8052	.1866	-.1402	.0920	-.0173	10	.8052	.1866	-.1402	.0920	-.0173

TABLE 6.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 6 MODEL - Continued

$$\frac{k}{c} = 0.06 \quad \frac{C_f}{c} = 0.20$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_n$
$M = 0.40$						$M = 0.95$					
-10	.3355					-10	.7749	.1613	.0551	-.0924	-.0246
-7	-.1439					-7	.3520	.0973	.0125	-.0651	-.0149
-5	.0072					-5	-.3791	.0492	-.0221	-.0425	-.0105
-3	.1507					-3	-.1979	.0461	-.0544	-.0199	-.0057
-2	.2260					-2	-.0771	.0435	-.0739	-.0064	-.0043
-1	.2960					-1	.0583	.0205	-.1074	.0098	-.0038
0	.3570					0	.2354	.0553	-.1525	.0290	-.0043
1	.4180					1	.3979	.0676	-.1908	.0482	-.0055
2	.4790					2	.5270	.0820	-.2087	.0634	-.0065
3	.5436					3	.6436	.0999	-.2216	.0772	-.0085
5	.6605					5	.8353	.1434	-.2433	.0981	-.0134
7	.7858					7	.9998	.2038	-.2728	.1190	-.0197
10	.8575										
15	1.0513										
20	1.2127										
25	1.2917										
$M = 0.60$						$M = 1.00$					
-10	-.3858	.0800	-.0592	-.0462	-.0114	-10	-.7322	.1560	.0561	-.0885	-.0234
-7	-.1692	.0257	-.1258	-.0179	-.0028	-7	-.3127	.0967	.0064	-.0600	-.0147
-5	-.0130	.0132	-.1312	-.0012		-5	-.3431	.0711	-.0304	-.0391	-.0104
-3	.1376	.0141	-.1258	-.0182	-.0008	-3	-.1536	.0565	-.0664	-.0158	-.0074
-2	.2203	.0169	-.1183	-.0269	-.0007	-2	-.0479	.0555	-.0856	-.0036	-.0059
-1	.2956	.0206	-.1180	-.0360	-.0009	-1	.0559	.0540	-.1061	.0087	-.0053
0	.3849	.0260	-.1177	-.0463	-.0011	0	.1855	.0589	-.1293	.0236	-.0051
1	.5132	.0335	-.1168	-.0608	-.0015	1	.3152	.0662	-.1571	.0391	-.0058
2	.5783	.0420	-.1168	-.0686	-.0018	2	.4648	.0805	-.1924	.0568	-.0070
3	.6582	.0536	-.1133	-.0779	-.0018	3	.6005	.0982	-.2127	.0713	-.0084
5	.7977	.0824	-.1063	-.0945	-.0039	5	.7601	.1374	-.2369	.0907	-.0122
7	.9130	.1303	-.1135	-.1092	-.0089	7	.9437	.1938	-.2613	.1114	-.0186
10	.9818	.2103	-.1587	-.1207	-.0192						
15	.9037	.3046	-.1855	-.1176	-.0324						
$M = 0.80$						$M = 1.05$					
-10	-.4676	.0955	-.0620	-.0544	-.0147	-10	-.7116	.1496	.0548	-.0839	-.0224
-7	-.4056	.0482	-.0922	-.0460	-.0082	-7	-.4858	.0941	.0072	-.0570	-.0143
-5	-.1947	.0234	-.1237	-.0209	-.0034	-5	-.3233	.0682	-.0277	-.0368	-.0105
-3	.0459	.0171	-.1278	-.0072	-.0015	-3	-.1473	.0550	-.0626	-.0132	-.0076
-2	.1625	.0198	-.1284	-.0203	-.0013	-2	-.0516	.0541	-.0806	-.0049	-.0054
-1	.2704	.0257	-.1306	-.0323	-.0012	-1	.0497	.0532	-.0977	.0080	-.0054
0	.3907	.0341	-.1377	-.0460	-.0012	0	.1626	.0564	-.1189	.0204	-.0053
1	.5122	.0448	-.1432	-.0604	-.0016	1	.2831	.0645	-.1420	.0380	-.0059
2	.6697	.0579	-.1526	-.0775	-.0024	2	.3940	.0753	-.1654	.0483	-.0065
3	.8012	.0747	-.1542	-.0925	-.0036	3	.5318	.0696	-.1993	.0638	-.0079
5	.9376	.1108	-.1542	-.1128	-.0075	5	.7230	.1307	-.2251	.0860	-.0119
7	1.0294	.1616	-.1596	-.1268	-.0127	7	.8837	.1834	-.2505	.1049	-.0176
10	1.1286	.2579	-.1652	-.1393	-.0224						
15	.9376	.3171	-.1997	-.1168	-.0346						
$M = 0.90$						$M = 1.10$					
-10	-.6708	.1362	-.0003	-.0822	-.0211	-10	-.6749	.1452	.0547	-.0815	-.0218
-7	-.5364	.0819	-.0075	-.0636	-.0130	-7	-.4726	.0905	.0049	-.0548	-.0139
-5	-.3810	.0485	-.0305	-.0423	-.0077	-5	-.3163	.0217	-.0242	-.0363	-.0102
-3	-.1686	.0328	-.0668	-.0167	-.0042	-3	-.1361	.0543	-.0582	-.0146	-.0074
-2	.0175	.0312	-.1216	-.0046	-.0023	-2	-.0405	.0533	-.0795	-.0033	-.0054
-1	.1883	.0350	-.1506	-.0241	-.0022	-1	.0570	.0520	-.0958	.0077	-.0057
0	.3131	.0430	-.1589	-.0383	-.0026	0	.1582	.0543	-.1143	.0196	-.0056
1	.4270	.0539	-.1651	-.0516	-.0038	1	.2722	.0615	-.1347	.0330	-.0062
2	.5386	.0668	-.1676	-.0636	-.0050	2	.3788	.0941	-.1562	.0455	-.0073
3	.6525	.0862	-.1869	-.0764	-.0066	3	.4984	.0868	-.1814	.0592	-.0082
5	.8167	.1238	-.2039	-.0985	-.0106	5	.6860	.1221	-.2148	.0812	-.0120
7	.9765	.1777	-.2184	-.1180	-.0166	7	.8515	.1741	-.2408	.1003	-.0176
10	1.1911	.2746	-.2625	-.1396	-.0277						

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TABLE 6.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 6 MODEL - Continued.

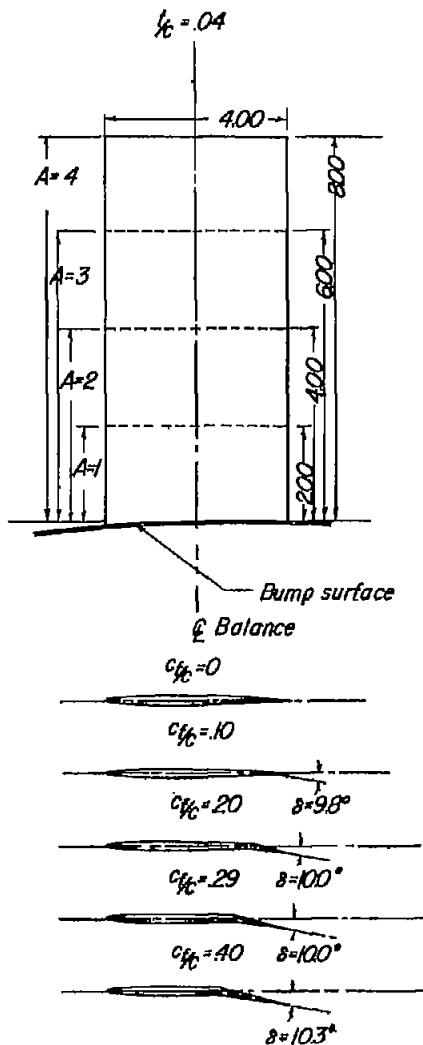
 $\frac{c}{c} = 0.06$        $\frac{c_L}{c} = 0.30$ 

$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$	$\alpha$ , deg	$c_L$	$c_D$	$c_M$	$c_I$	$c_n$
$M = 0.40$											
-10	-2418					-10	-6551	.1554	-.0069	-.0712	-.0218
-7	-.0412					-7	-4409	.1028	-.0215	-.0443	-.0135
-5	.1021					-5	-2599	.0782	-.0837	-.0215	-.0096
-3	.2507					-3	-0374	.0645	-.1251	.0062	-.0056
-2	.3349					-2	.1248	.0685	-.1573	.0239	-.0054
-1	.4083					-1	.2911	.0773	-.1950	.0432	-.0056
0	.4567					0	.4471	.0890	.2171	.0592	-.0062
1	.5176					1	.5677	.1043	-.2203	.0727	-.0072
2	.5821					2	.6696	.1248	-.2291	.0427	-.0088
3	.6358					3	.7694	.1463	-.2410	.0488	-.0109
5	.7522					5	.9358	.1923	-.2548	.0584	-.0159
7	.8686					7	1.1146	.2562	-.2778	.0676	-.0227
10	.9134										
15	1.0925										
20	1.2358										
25	1.2967										
$M = 0.60$											
-10	-.3063	.0835	-.0672	-.0401	-.0193	-10	-.6374	.1519	-.0066	-.0682	-.0215
-7	-.0898	.0319	-.0685	-.0249	-.0032	-7	-.4123	.1009	-.0494	-.0408	-.0140
-5	.1095	.0206	-.1302	.0364	-.0027	-5	-.2390	.0813	-.0837	-.0190	-.0100
-3	.2992	.0251	-.1302	.0362	-.0022	-3	-.0319	.0740	-.1229	.0060	-.0068
-2	.3898	.0301	-.1322	.0469	-.0022	-2	.0837	.0725	-.1419	.0197	-.0065
-1	.4734	.0369	-.1219	.0559	-.0025	-1	.2151	.0764	-.1661	.0350	-.0067
0	.5532	.0438	-.1199	.0656	-.0028	0	.3884	.0862	-.2057	.0548	-.0069
1	.6126	.0507	-.1138	.0721	-.0029	1	.5218	.1029	-.2177	.0672	-.0076
2	.6869	.1077	-.1035	.0817	-.0030	2	.6174	.1195	-.2238	.0785	-.0086
3	.7704	.0780	-.1035	.0901	-.0033	3	.7230	.1401	-.2291	.0919	-.0108
5	.9096	.1123	-.0945	.1069	-.0064	5	.8803	.1841	-.2485	.1096	-.0150
7	.9653	.1639	-.1135	.1163	-.0122	7	1.0476	.2439	-.2688	.1270	-.0216
10	1.0581	.2519	-.1454	.1292	-.0230						
15	1.0136	.3616	-.1753	.1295	-.0383						
20	.9245	.4327	-.1733	.1163	-.0484						
$M = 0.80$											
-10	-.3654	.0931	-.0992	-.0449	-.0115	-10	-.6111	.1474	-.0021	-.0557	-.0208
-7	-.2291	.0476	-.1408	-.0259	-.0049	-7	-.4010	.0967	-.0452	-.0391	-.0134
-5	-.0248	.0292	-.1518	-.0020	-.0024	-5	-.2349	.0766	-.0790	-.0162	-.0101
-3	.1858	.0338	-.1436	.0222	-.0022	-3	-.0382	.0714	-.1149	.0046	-.0067
-2	.2948	.0393	-.1397	.0338	-.0022	-2	.0726	.0700	-.1347	.0181	-.0068
-1	.3864	.0469	-.1381	.0441	-.0023	-1	.1814	.0198	-.1517	.0181	-.0068
0	.4954	.0566	-.1343	.0565	-.0025	0	.3132	.0835	-.1761	.0451	-.0071
1	.6131	.0704	-.1397	.0696	-.0028	1	.4583	.0982	-.2015	.0406	-.0078
2	.7679	.0871	-.1408	.0866	-.0038	2	.5767	.1146	-.2154	.0732	-.0087
3	.8769	.1063	-.1436	.1022	-.0052	3	.6722	.1352	-.2230	.0856	-.0104
5	1.0098	.1480	-.1452	.1191	-.0093	5	.8250	.1756	-.2357	.1026	-.0144
7	1.0949	.1985	-.1381	.1311	-.0156	7	.9930	.2353	-.2577	.1211	-.0208
10	1.1147	.2687	-.1518	.1431	-.0255						
15	.9686	.3545	-.1874	.1291	-.0369						
$M = 0.90$											
-10	-.5904	.1349	-.0517	-.0651	-.0195	-10	-.5838	.1427	-.0032	-.0324	-.0203
-7	-.4548	.0812	-.0517	-.0448	-.0117	-7	-.3874	.0944	-.0422	-.0193	-.0135
-5	-.2690	.0559	-.0856	-.0209	-.0070	-5	-.2240	.0749	-.0739	-.0089	-.0096
-3	.0437	.0462	-.1509	.0154	-.0040	-3	-.0367	.0690	-.1105	.0025	-.0073
-2	.2034	.0516	-.1678	.0329	-.0032	-2	.0734	.0682	-.1287	.0085	-.0065
-1	.3324	.0623	-.1799	.0472	-.0039	-1	.1652	.0722	-.1430	.0144	-.0068
0	.4461	.0715	-.1727	.0596	-.0044	0	.2846	.0803	-.1633	.0208	-.0070
1	.5467	.0892	-.1872	.0720	-.0053	1	.4223	.0948	-.1876	.0288	-.0076
2	.6451	.1027	-.1799	.0823	-.0064	2	.5471	.1111	-.2071	.0357	-.0087
3	.7544	.1253	-.1920	.0962	-.0083	3	.6463	.1300	-.2153	.0412	-.0105
5	.9118	.1726	-.2114	.1157	-.0131	5	.7987	.1697	-.2291	.0496	-.0143
7	1.0715	.2291	-.2235	.1334	-.0154						
$M = 1.00$											
-10	-.6374	.1519	-.0066	-.0682	-.0215						
-7	-.4123	.1009	-.0494	-.0408	-.0140						
-5	-.2390	.0813	-.0837	-.0190	-.0100						
-3	-.0319	.0740	-.1229	.0060	-.0068						
-2	.0837	.0725	-.1419	.0197	-.0065						
-1	.2151	.0764	-.1661	.0350	-.0067						
0	.3884	.0862	-.2057	.0548	-.0069						
1	.5218	.1029	-.2177	.0672	-.0076						
2	.6174	.1195	-.2238	.0785	-.0086						
3	.7230	.1401	-.2291	.0919	-.0108						
5	.8803	.1841	-.2485	.1096	-.0150						
7	1.0476	.2439	-.2688	.1270	-.0216						
$M = 1.05$											
-10	-.6111	.1474	-.0021	-.0557	-.0208						
-7	-.4010	.0967	-.0452	-.0391	-.0134						
-5	-.2349	.0766	-.0790	-.0162	-.0101						
-3	-.0382	.0714	-.1149	.0046	-.0067						
-2	.0726	.0700	-.1347	.0181	-.0068						
-1	.1814	.0198	-.1517	.0181	-.0068						
0	.3132	.0835	-.1761	.0451	-.0071						
1	.4583	.0982	-.2015	.0406	-.0078						
2	.5767	.1146	-.2154	.0732	-.0087						
3	.6722	.1352	-.2230	.0856	-.0104						
5	.8250	.1756	-.2357	.1026	-.0144						
7	.9930	.2353	-.2577	.1211	-.0208						
$M = 1.10$											
-10	-.5838	.1427	-.0032	-.0324	-.0203						
-7	-.3874	.0944	-.0422	-.0193	-.0135						
-5	-.2240	.0749	-.0739	-.0089	-.0096						
-3	-.0367	.0690	-.1105	.0025	-.0073						
-2	.0734	.0682	-.1287	.0085	-.0065						
-1	.1652	.0722	-.1430	.0144	-.0068						
0	.2846	.0803	-.1633	.0208	-.0070						
1	.4223	.0948	-.1876	.0288	-.0076						
2	.5471	.1111	-.2071	.0357	-.0087						
3	.6463	.1300	-.2153	.0412	-.0105						
5	.7987	.1697	-.2291	.0496	-.0143						

TABLE 6.- THE AERODYNAMIC CHARACTERISTICS OF THE ASPECT RATIO 6 MODEL - Concluded

$$\frac{L}{S} = 0.06 \quad \frac{C_f}{S} = 0.40$$

$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_B$	$\alpha$ , deg	$C_L$	$C_D$	$C_M$	$C_I$	$C_B$
$M = 0.40$						$M = 0.95$					
-10	-0.2076					-10	-0.5620	0.1454	-0.0497	-0.0694	-0.0408
-7	.0055					-7	-0.3487	0.3880	-0.0939	-0.0413	-0.0131
-5	.1694					-5	-0.1603	0.3245	-0.1345	-0.0177	-0.0093
-3	.3151					-3	.1072	0.2959	-0.1929	0.0147	-0.0074
-2	.3879					-2	.2758	0.3198	-0.2192	0.0337	-0.0073
-1	.4571					-1	.4038	0.3583	-0.2431	0.0480	-0.0077
0	.5154					0	.5131	0.4115	-0.2431	0.0606	-0.0086
1	.5973					1	.6266	0.4761	-0.2477	0.0733	-0.0098
2	.6520					2	.7306	0.4393	-0.2569	0.0859	-0.0115
3	.7066					3	.8245	0.4597	-0.2415	0.0963	-0.0137
5	.8341					5	.9784	0.4084	-0.2836	0.1159	-0.0190
7	.9324					7	1.1574	0.2784	-0.3030	0.1331	-0.0264
10	.9215										
15	1.1145										
20	1.2311										
25	1.2639										
$M = 0.60$						$M = 1.00$					
-10	-0.2495	0.0653	-0.0748	-0.0318	-0.0071	-10	-0.5263	0.1283	-0.0459	-0.0645	-0.0197
-7	.0170	0.185	-0.1207	-0.0228	-0.0022	-7	-0.3170	0.0990	-0.0944	-0.0219	-0.0132
-5	.2002	0.0194	-0.1440	-0.0238	-0.0014	-5	-0.1935	0.0843	-0.1297	-0.0177	-0.0098
-3	.3824	0.0255	-0.1270	-0.0443	-0.0010	-3	.0558	0.0785	-0.1729	0.0086	-0.0087
-2	.4683	0.0321	-0.1258	-0.0550	-0.0012	-2	.2014	0.0818	-0.2002	0.0255	-0.0080
-1	.5534	0.0391	-0.1136	-0.0636	-0.0017	-1	.3609	0.0892	-0.2329	0.0431	-0.0082
0	.6193	0.0474	-0.1165	-0.0718	-0.0019	0	.4885	0.1025	-0.2355	0.0573	-0.0088
1	.6987	0.0604	-0.1165	-0.0810	-0.0020	1	.5941	0.1186	-0.2417	0.0690	-0.0098
2	.7799	0.0729	-0.1082	-0.0895	-0.0024	2	.6978	0.1358	-0.2461	0.0816	-0.0117
3	.8554	0.0904	-0.1090	-0.0978	-0.0032	3	.7835	0.1569	-0.2567	0.0919	-0.0135
5	.9744	0.1254	-0.1061	.1121	-0.0064	5	.9291	0.2010	-0.2708	0.1090	-0.0185
7	1.0046	0.1792	-0.1278	.1173	-0.0150	7	1.1005	0.2667	-0.2946	0.1252	-0.0253
10	1.1216	0.2880	-0.1666	.1398	-0.0264						
15	.9366	0.3437	-0.1750	.1176	-0.0370						
20	.9177	0.4383	-0.1834	.1137	-0.0480						
$M = 0.80$						$M = 1.05$					
-10	-0.3272	0.0804	-0.1069	-0.0387	-0.0094	-10	-0.5107	0.1373	-0.0410	-0.0638	-0.0191
-7	.1334	0.0443	-0.1436	-0.0143	-0.0042	-7	-0.3156	0.0974	-0.0855	-0.0387	-0.0130
-5	.0692	0.0295	-0.1525	-0.0092	-0.0024	-5	-0.1530	0.0819	-0.1202	-0.0189	-0.0096
-3	.2618	0.0399	-0.1487	-0.0312	-0.0022	-3	.0593	0.0762	-0.1633	0.0059	-0.0085
-2	.3524	0.0477	-0.1470	-0.0413	-0.0023	-2	.1683	0.0800	-0.1815	0.0193	-0.0082
-1	.4430	0.0558	-0.1442	-0.0517	-0.0026	-1	.3060	0.0880	-0.2149	0.0341	-0.0082
0	.5702	0.0717	-0.1576	-0.0662	-0.0033	0	.4457	0.0988	-0.2259	0.0500	-0.0087
1	.6825	0.0842	-0.1525	-0.0784	-0.0041	1	.5585	0.1148	-0.2344	0.0619	-0.0097
2	.8131	0.0982	-0.1459	-0.0917	-0.0049	2	.6523	0.1327	0.2403	0.0724	-0.0112
5	.9138	.1188	-0.1525	.1055	-0.0066	3	.7422	0.1524	-0.2479	0.0842	-0.0132
5	1.0271	.1653	-0.1626	.1210	-0.0111	5	.8896	0.1962	-0.2657	0.1006	-0.0179
7	1.1152	.2196	-0.1682	.1344	-0.0177	7	1.0462	0.2606	-0.2869	0.1161	-0.0245
10	1.0145	.2643	-0.1693	.1251	-0.0262						
15	.8594	.3144	-0.1529	.1222	-0.0388						
20	.7124	.3367	-0.1108	.1214	-0.0529						
$M = 0.90$						$M = 1.10$					
-10	-0.4876	.1167	-0.0784	-0.0623	-0.0173	-10	-0.4905	0.1319	-0.0211	-0.0632	-0.0184
-7	.3258	0.0737	-0.0972	-0.0389	-0.0101	-7	-0.2976	0.0912	-0.0410	-0.0294	-0.0129
-5	.1028	.0581	-0.1432	-0.0117	-0.0059	-5	-0.1470	0.0777	-0.0585	-0.0208	-0.0095
-3	.1662	0.0559	-0.1756	-0.0198	-0.0043	-3	.0514	0.0741	-0.0789	0.0039	-0.0085
-2	.2886	0.0623	-0.1794	-0.0340	-0.0044	-2	.1617	0.0777	-0.0878	0.0156	-0.0083
-1	.3958	0.0721	-0.1809	-0.0460	-0.0051	-1	.2811	0.0845	-0.0984	0.0294	-0.0083
0	.4986	0.0844	-0.1857	-0.0577	-0.0059	0	.4097	0.0949	-0.1073	0.0439	-0.0087
1	.6254	.1054	-0.2051	.0718	-0.0075	1	.5236	0.1093	-0.1134	0.0572	-0.0097
2	.6800	.1086	-0.1881	.0796	-0.0076	2	.6118	0.1260	-0.1162	0.0669	-0.0111
3	.7828	.1212	-0.2003	.0924	-0.0099	3	.7056	0.1455	-0.1198	0.0782	-0.0132
5	.9206	.1742	-0.2099	.1097	-0.0143	5	.8414	0.1870	-0.1268	0.0936	-0.0176
7	1.0496	.2301	-0.2268	.1253	-0.0214	7	.9921	0.2457	-0.1561	0.1100	-0.0240
10	1.2376	.3302	-0.2651	.1444	-0.0321						



### General Dimensions

Airfoil section	65A004	65A006
Aspect ratios	4, 3, 2, 1	6, 5, 4, 3, 2
Sweep back $\%_4$	0°	0°
Taper ratio	1.0	1.0

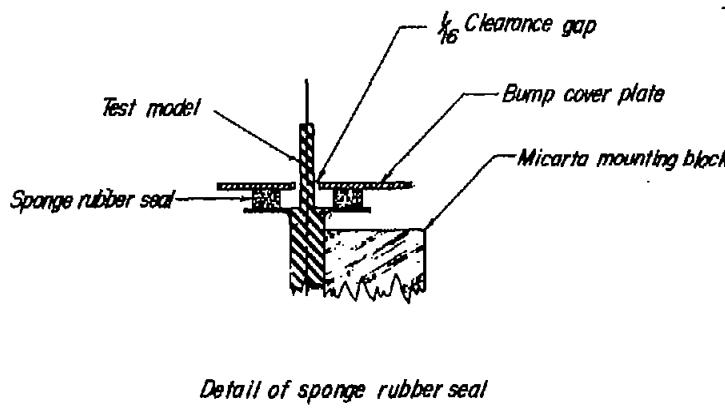
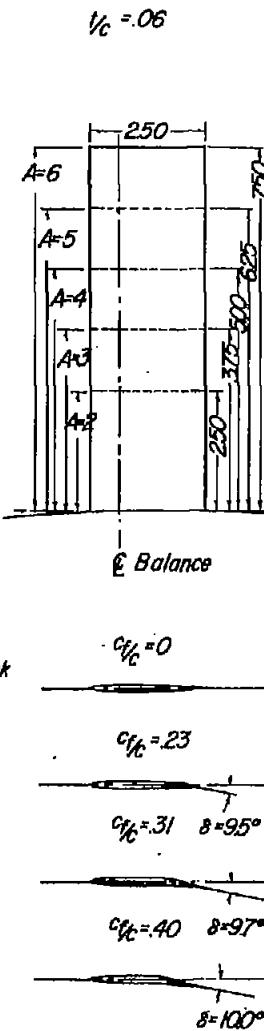


Figure 1.- General dimensions and model geometry. (All dimensions in inches unless otherwise noted.)

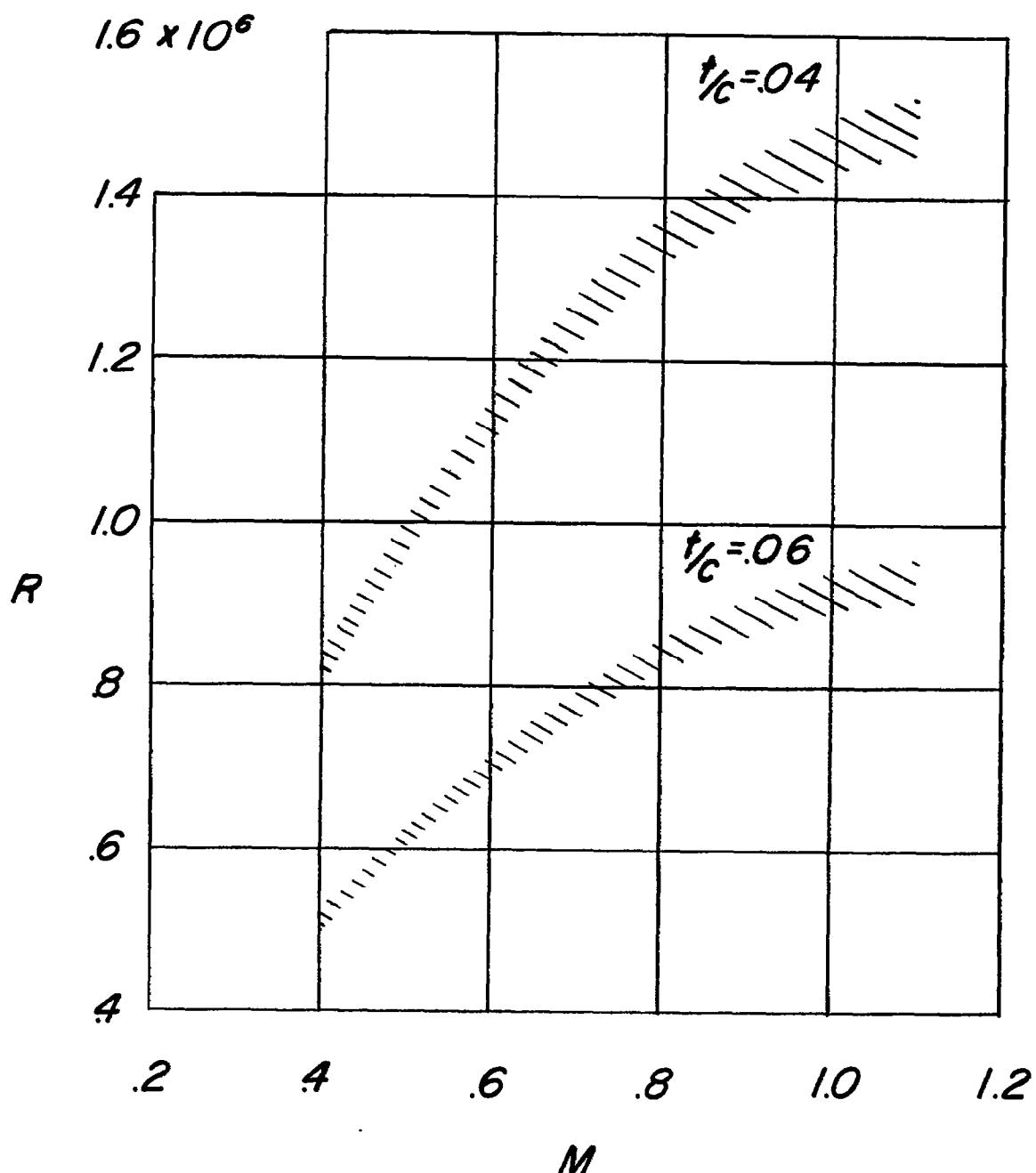


Figure 2.- Variation of test Reynolds number with Mach number.

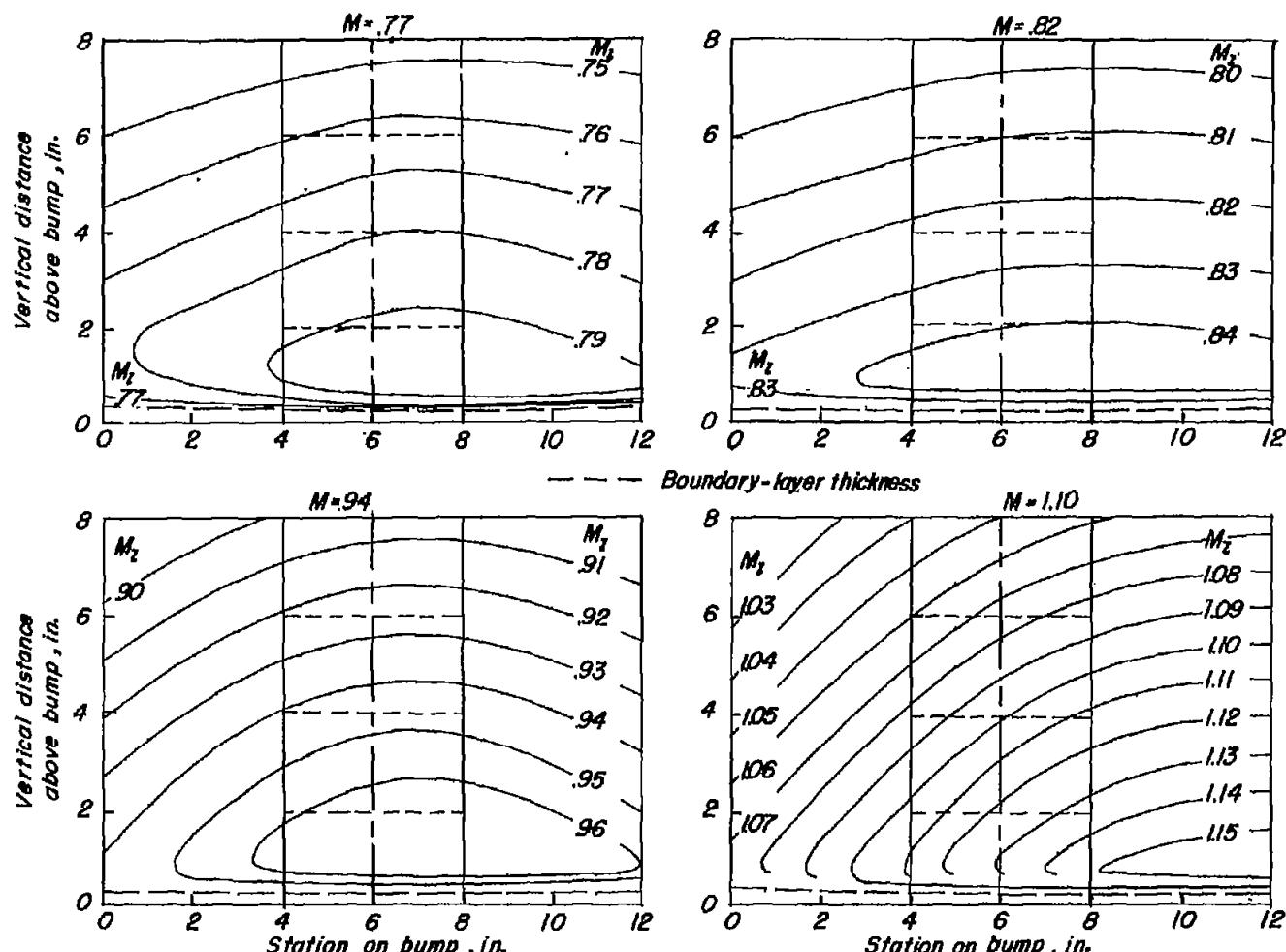
(a)  $t/c = 0.04$ .

Figure 3.- Typical Mach number contours over transonic bump in region of model location.

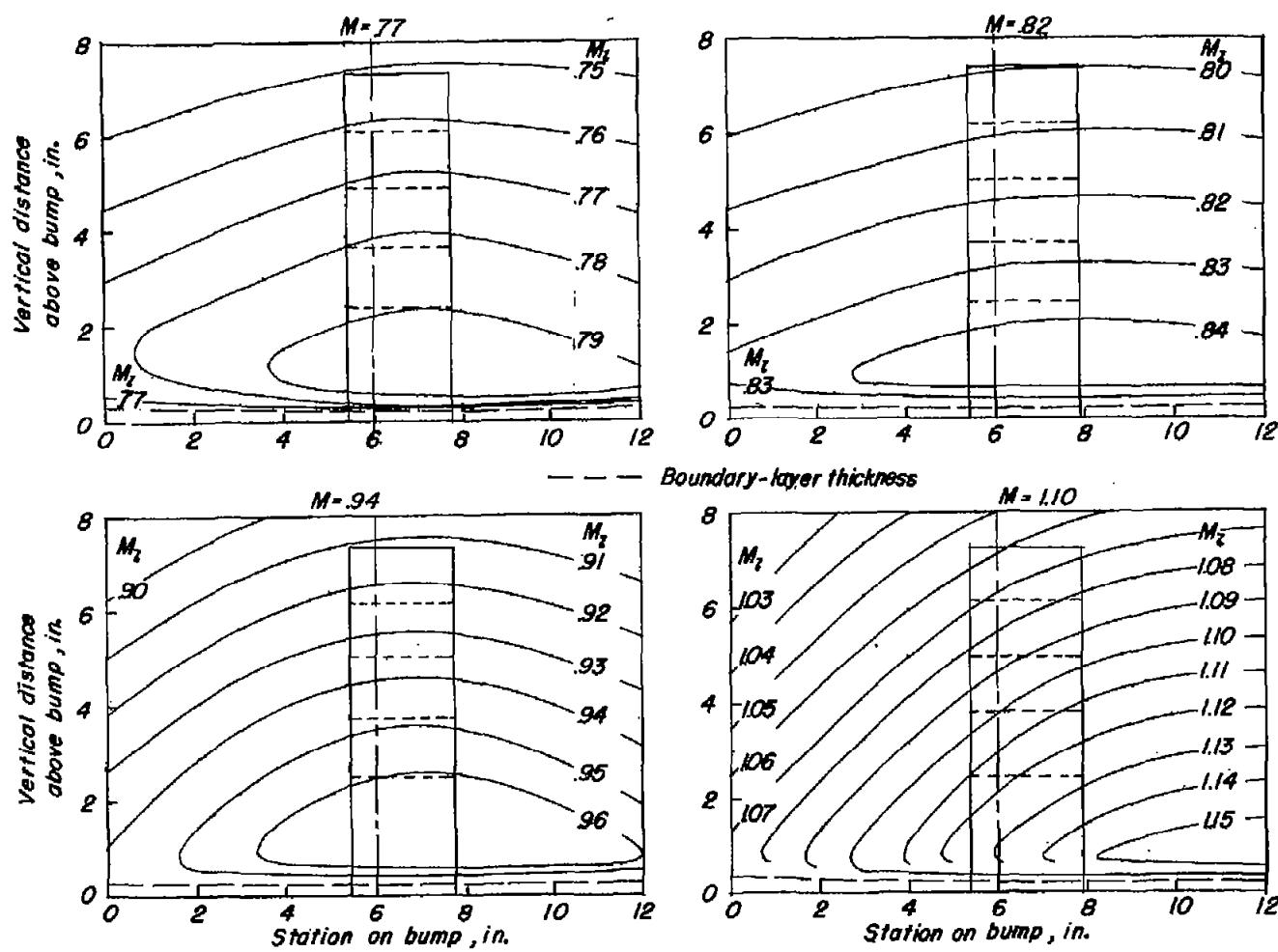
(b)  $t/c = 0.06$ .

Figure 3.- Concluded.

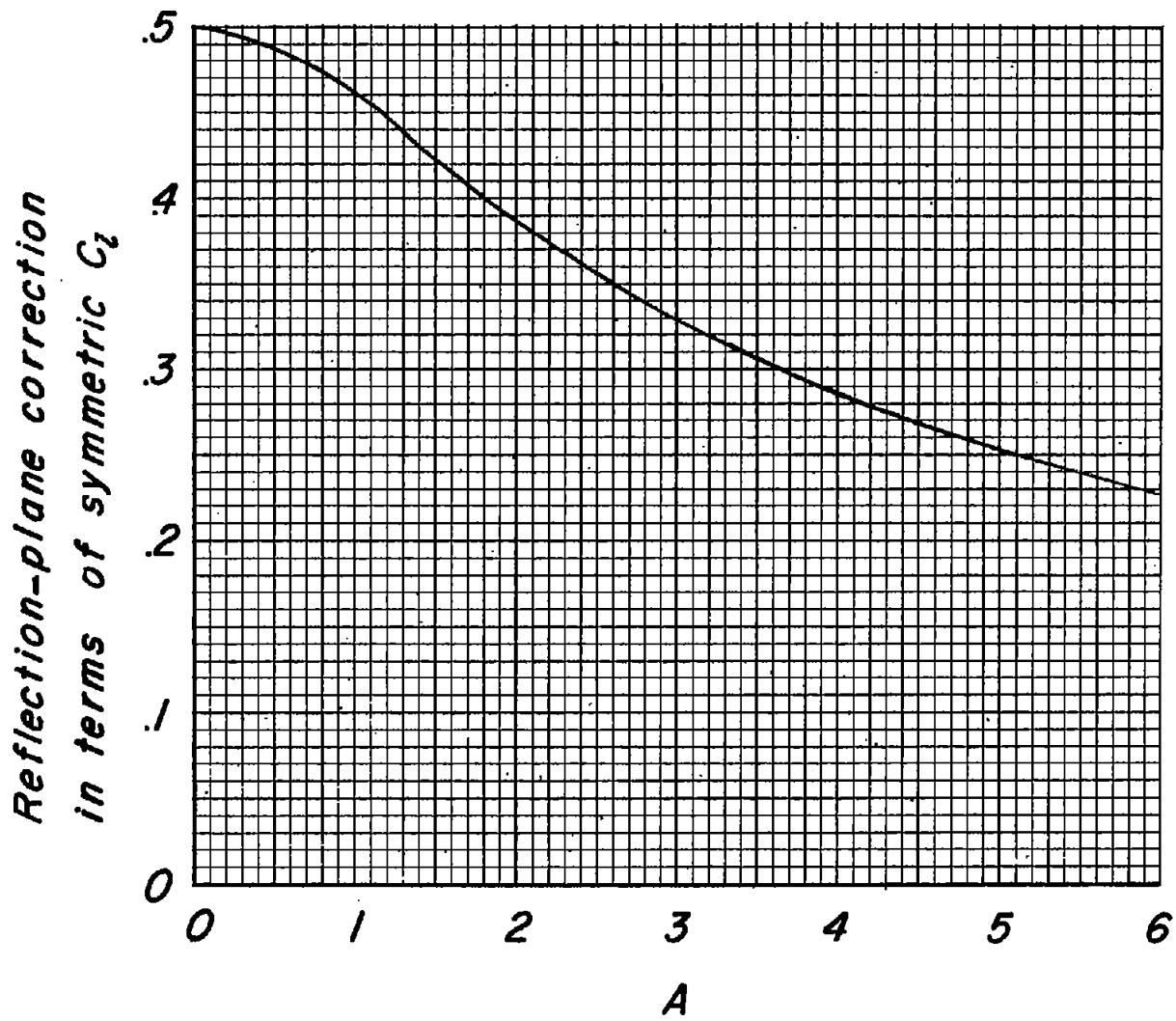


Figure 4.- Variation of reflection-plane correction with aspect ratio  
for full-span controls on untapered, unswept wings.

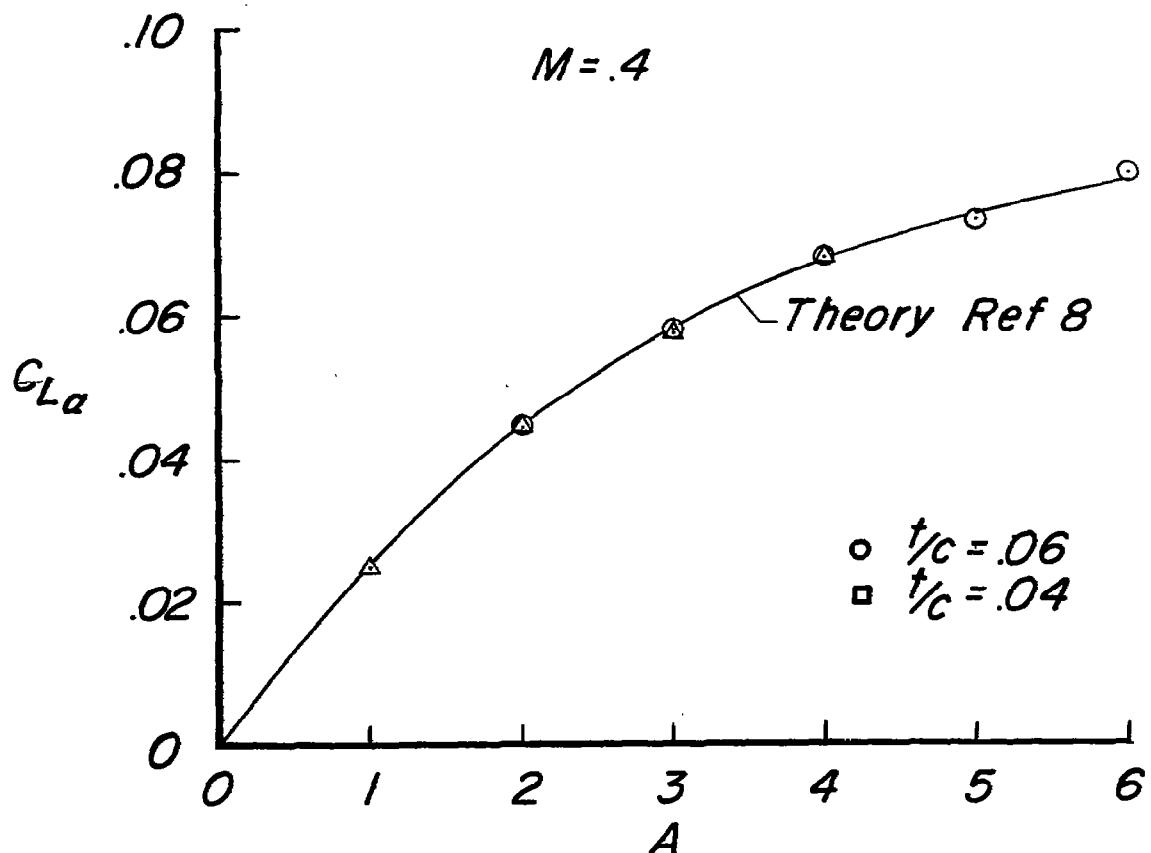


Figure 5.- Variation of the lift-curve slope with aspect ratio at a Mach number of 0.4.

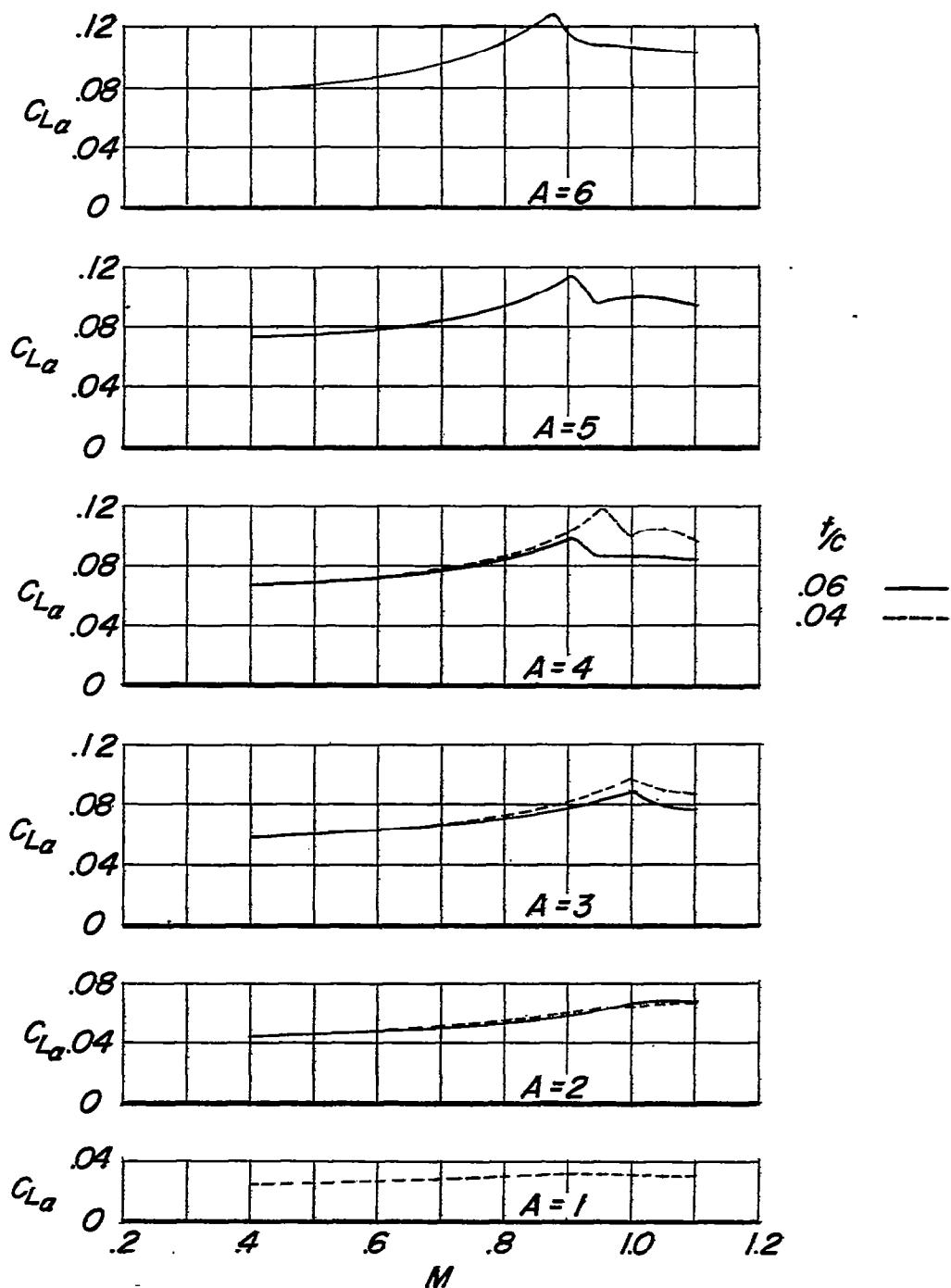


Figure 6.- The variation of lift-curve slope with Mach number.

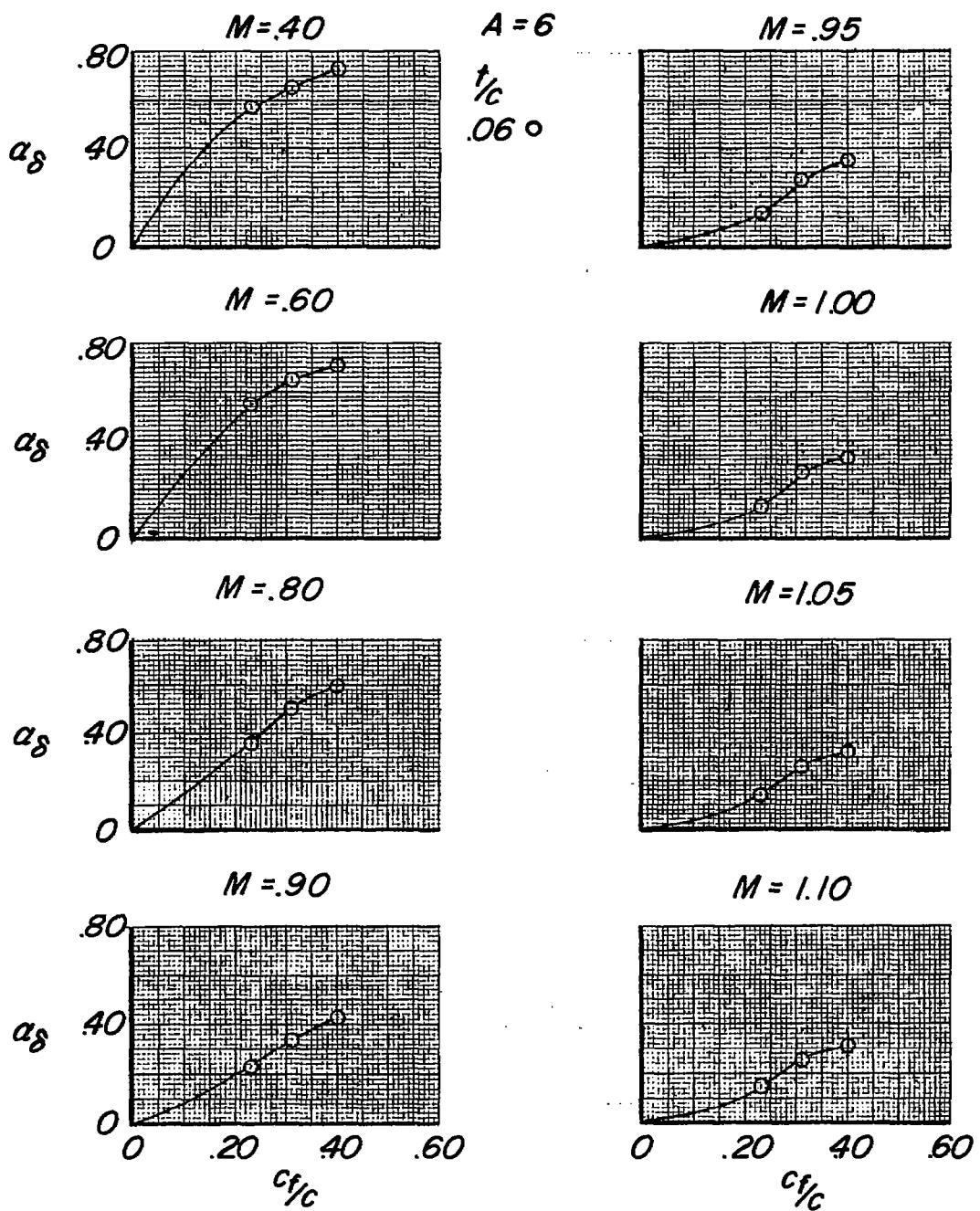
(a)  $A = 6.$ 

Figure 7.- Flap-effectiveness parameter  $\alpha_d$  as a function of  $c_f/c$ .

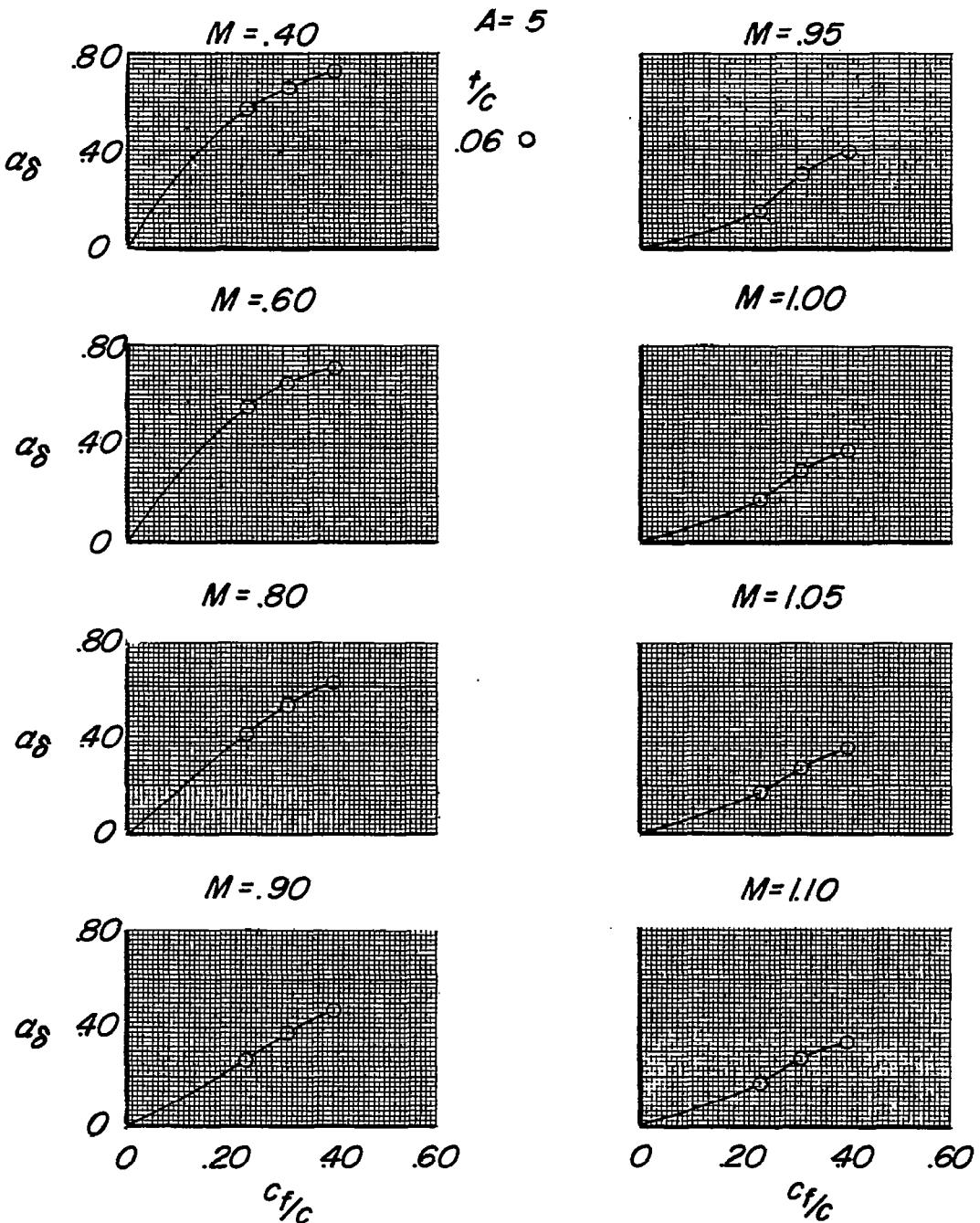
(b)  $A = 5.$ 

Figure 7.- Continued.

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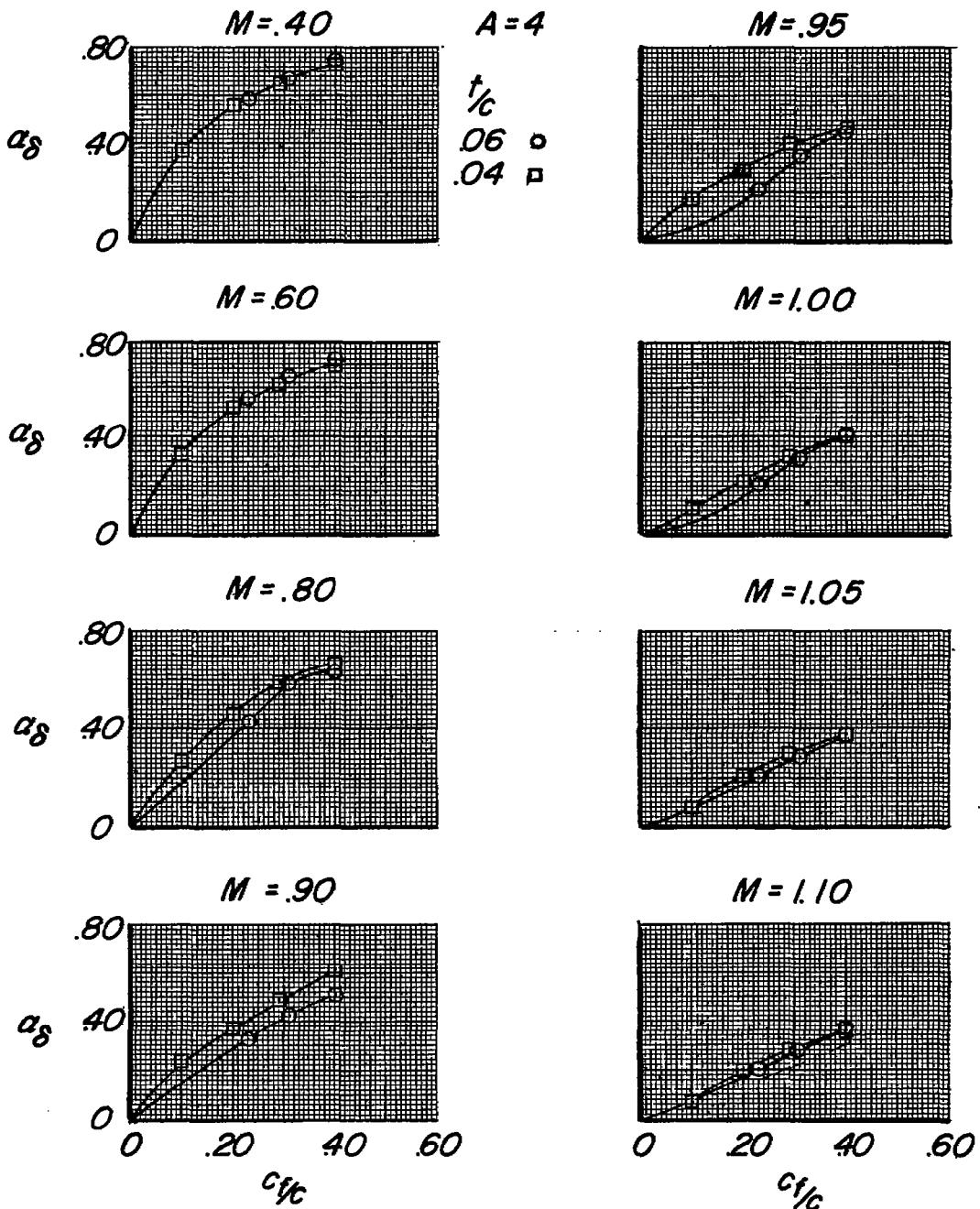
~~CONFIDENTIAL~~(c)  $A = 4.$ 

Figure 7.- Continued.

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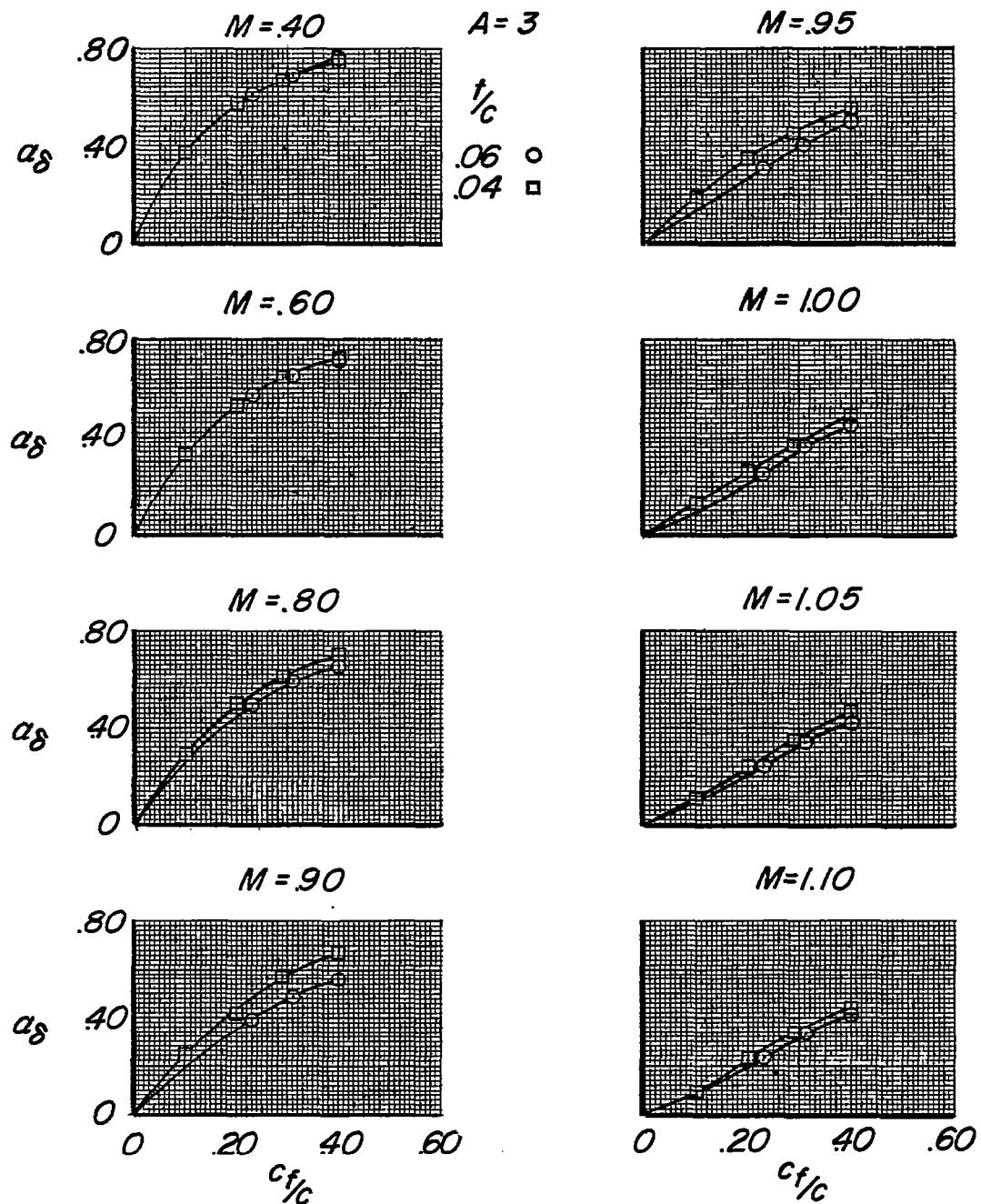
(d)  $A = 3.$ 

Figure 7.- Continued.

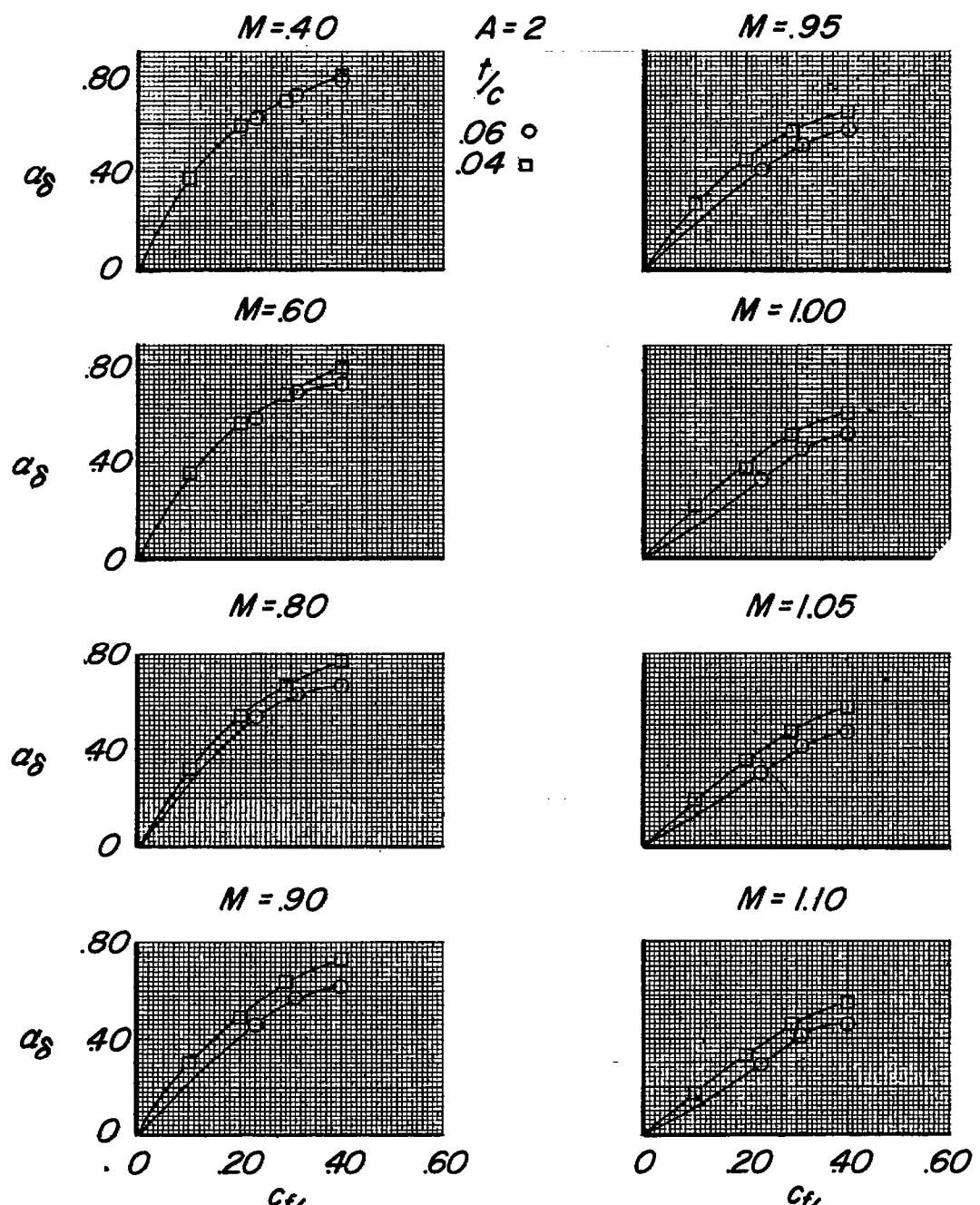
(e)  $A = 2$ .

Figure 7.- Continued.

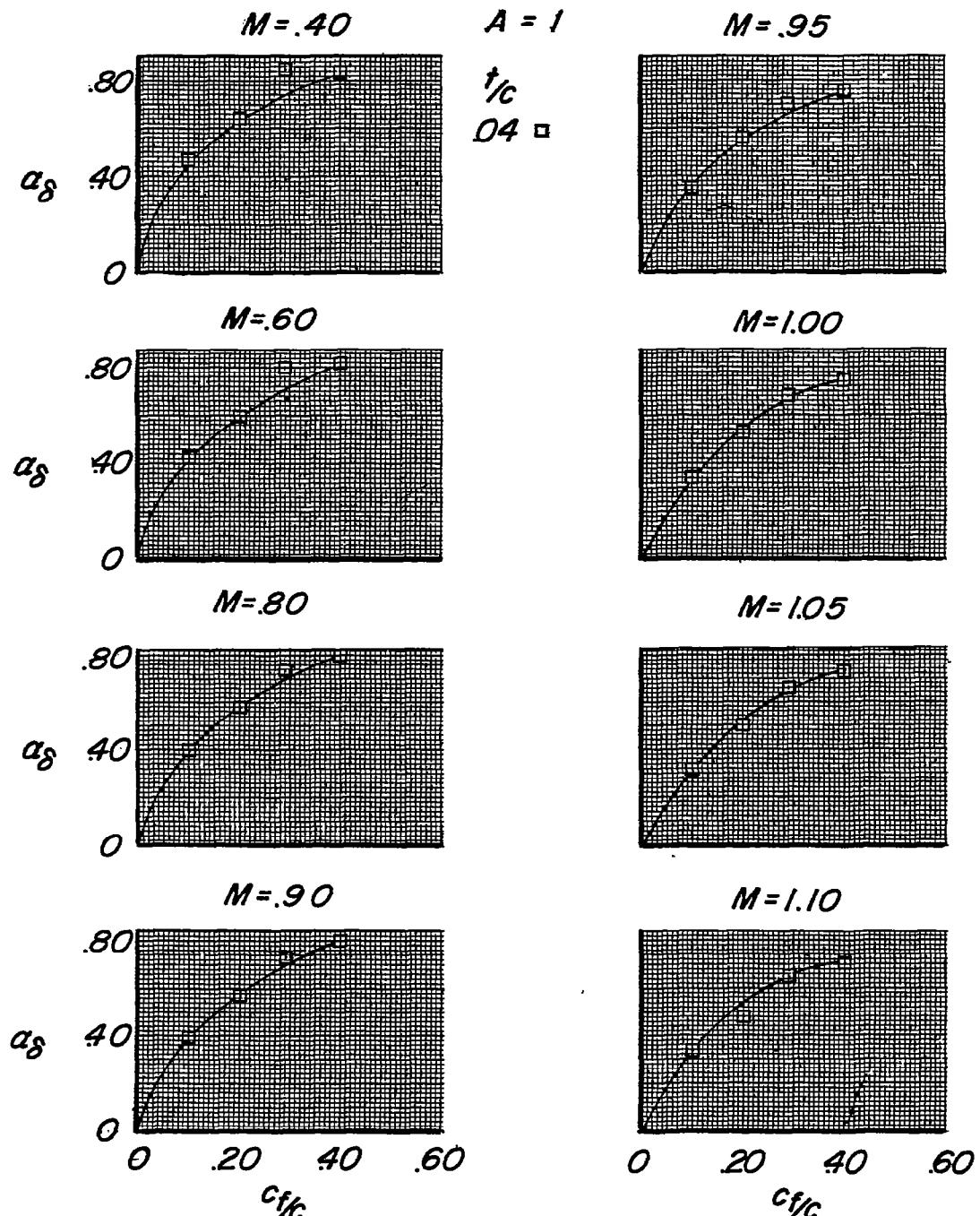
(f)  $A = 1.$ 

Figure 7.- Concluded.

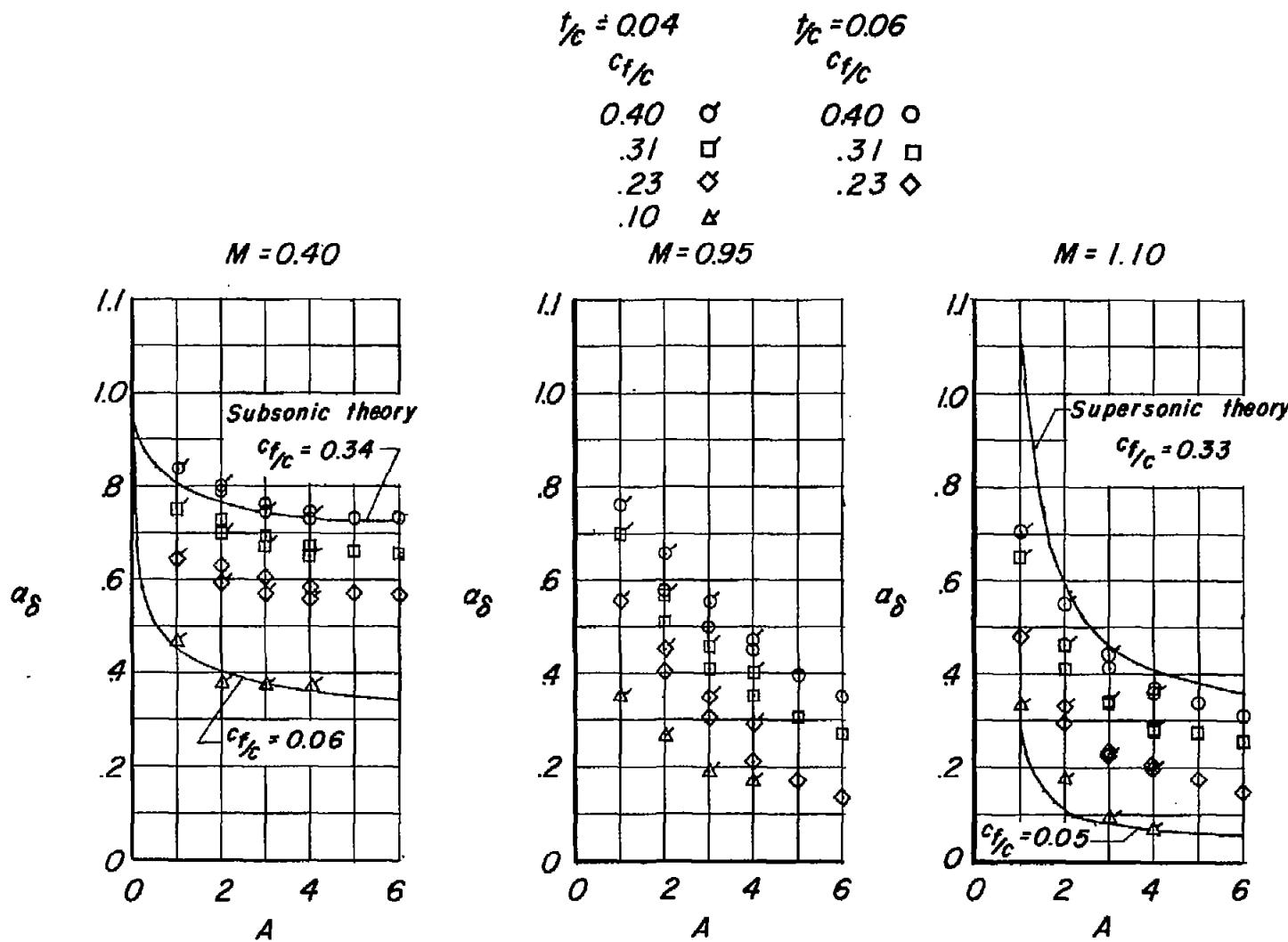


Figure 8.- Variation of  $a_\delta$  with aspect ratio at various Mach numbers.